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Westinghouse Astronuclear Laboratory



DEVELOPMENTAL TEST PROGRAM PRESENTATION

(TITLE UNCLASSIFIED)

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Westinghouse Astronuclear Laboratory



DEVELOPMENTAL TEST PROGRAM PRESENTATION

(TITLE UNCLASSIFIED)

SPECIAL REREVIEW FINAL DETERMINATION	Reviewer	Class.	Date
	KAW	U	4-12-82
Class: <u>U</u>			

Classification cancelled (or changed to _____)
by authority of _____
by A.F.C. TIC, date SEP 11 1973

~~GROUP 1~~
~~EXCLUDED FROM AUTOMATIC DOWNGRADING
AND DECLASSIFICATION~~

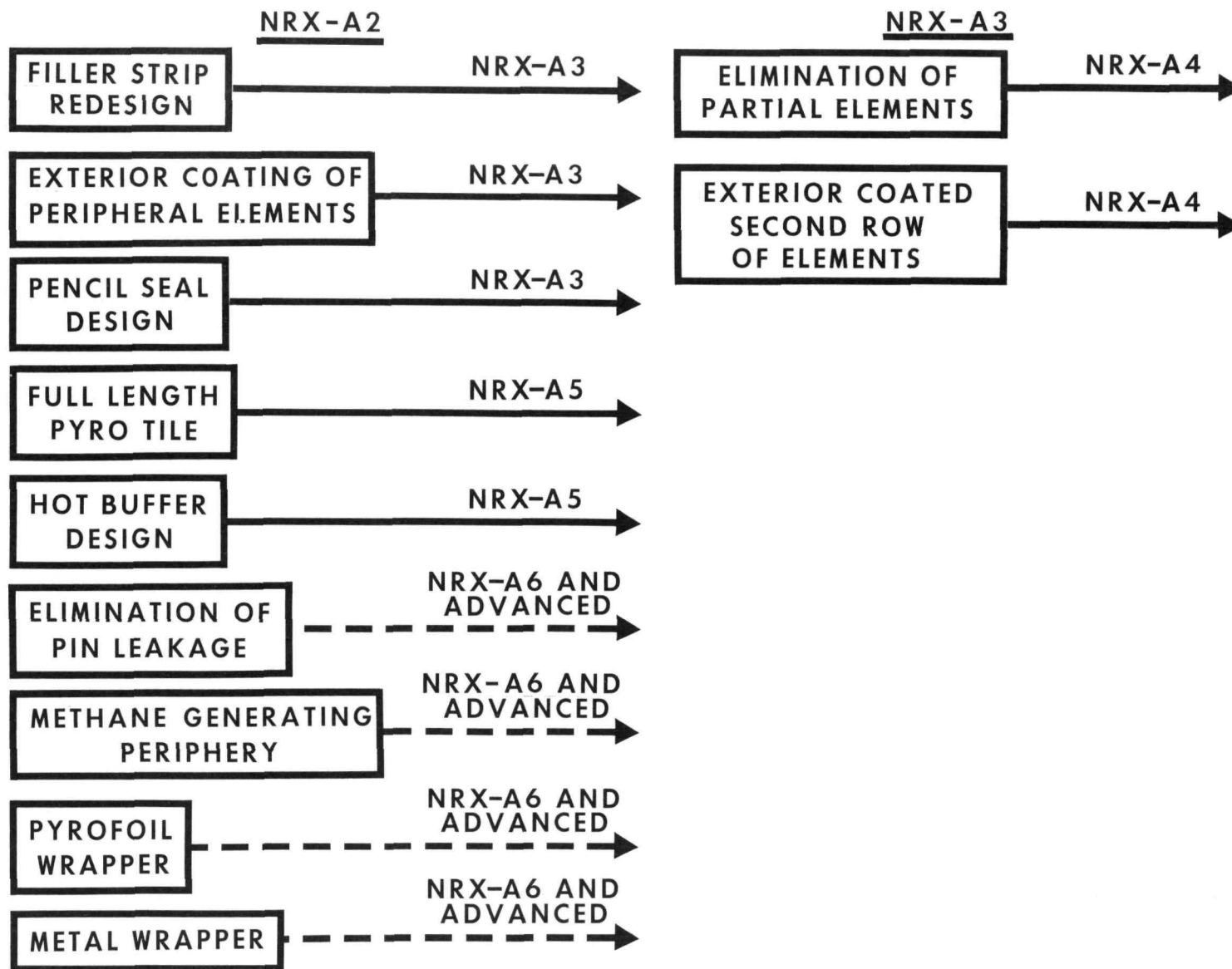
INFORMATION CATEGORY

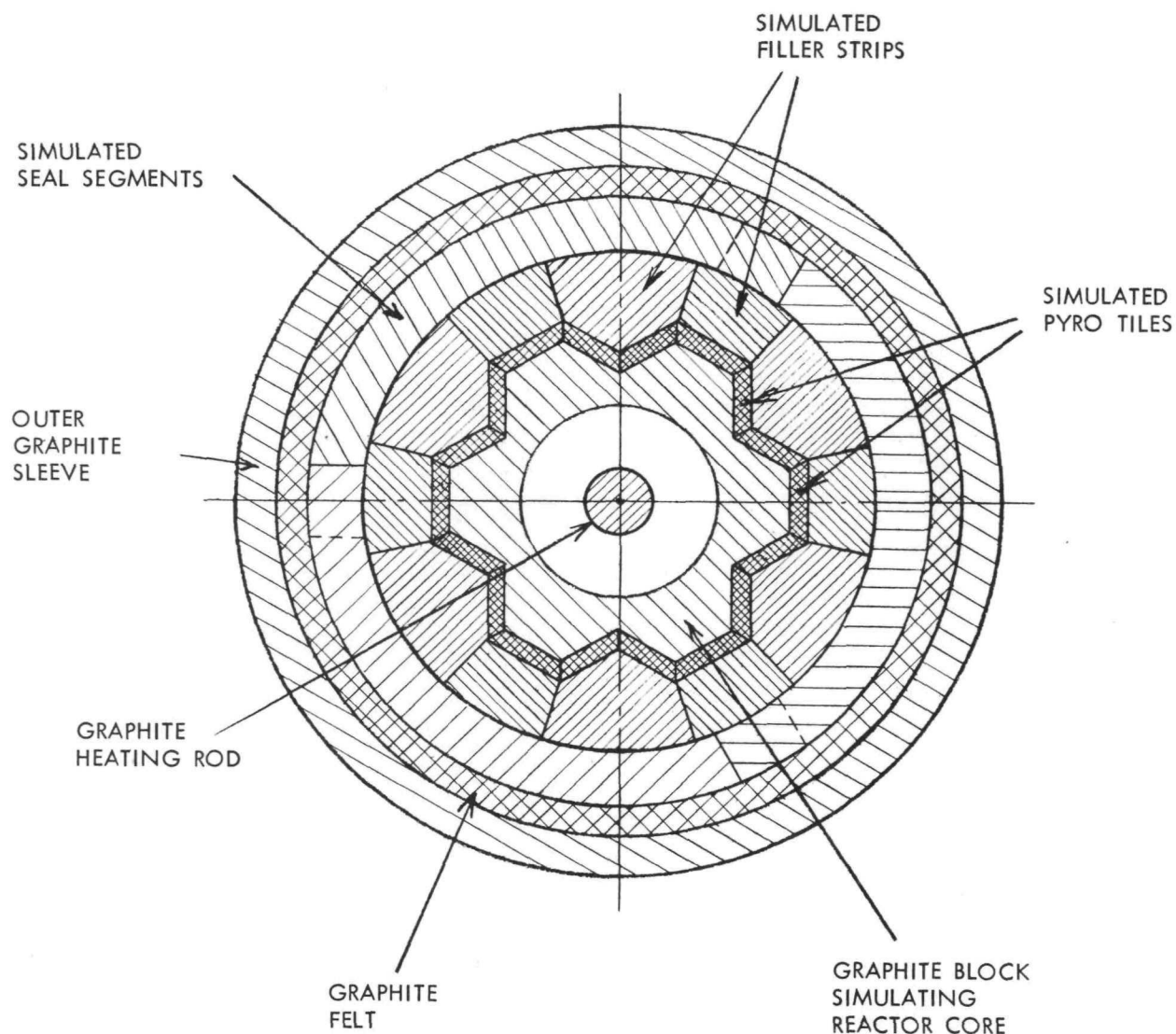
~~Confidential - R.D.~~
~~W.H.C. Casey 9/14/65~~
Authorized Classifier _____ Date _____

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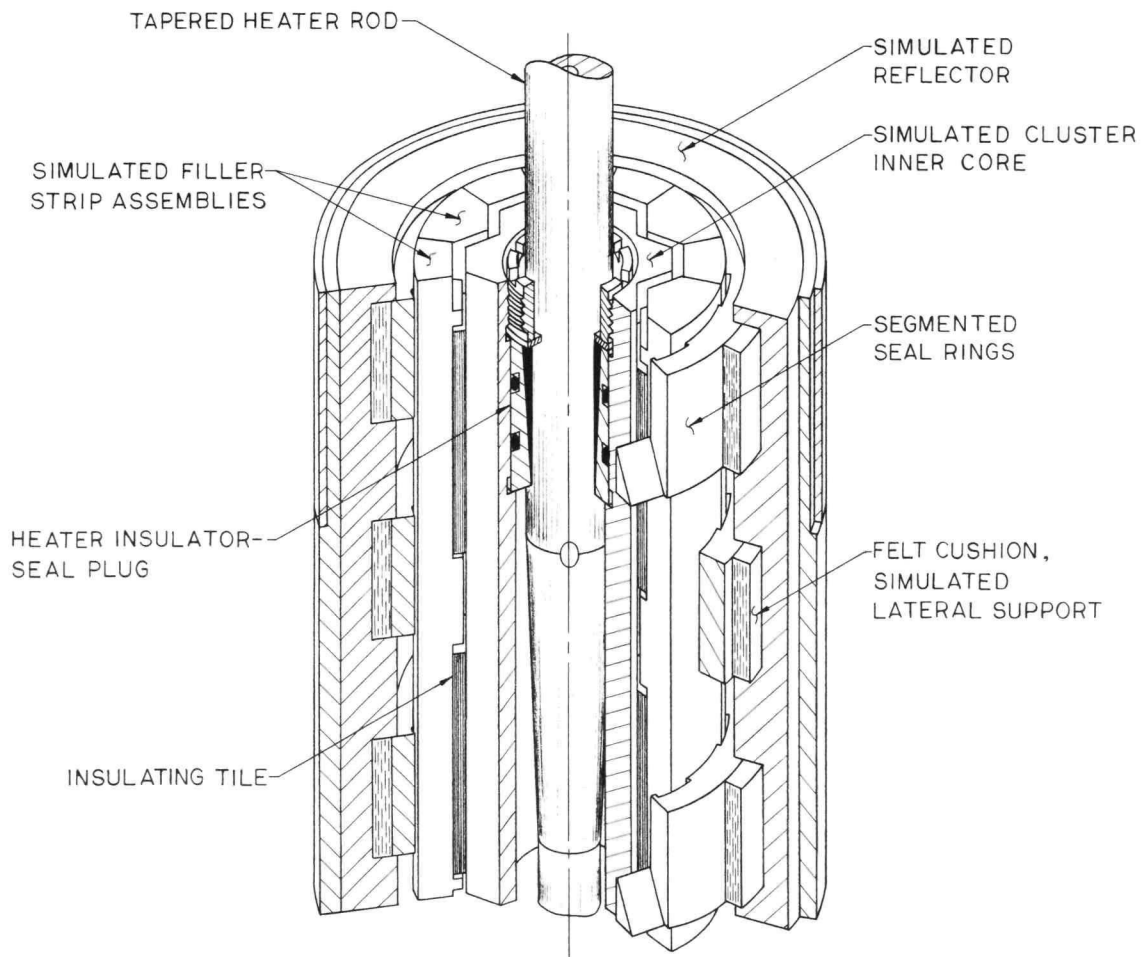
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PERIPHERAL CORROSION

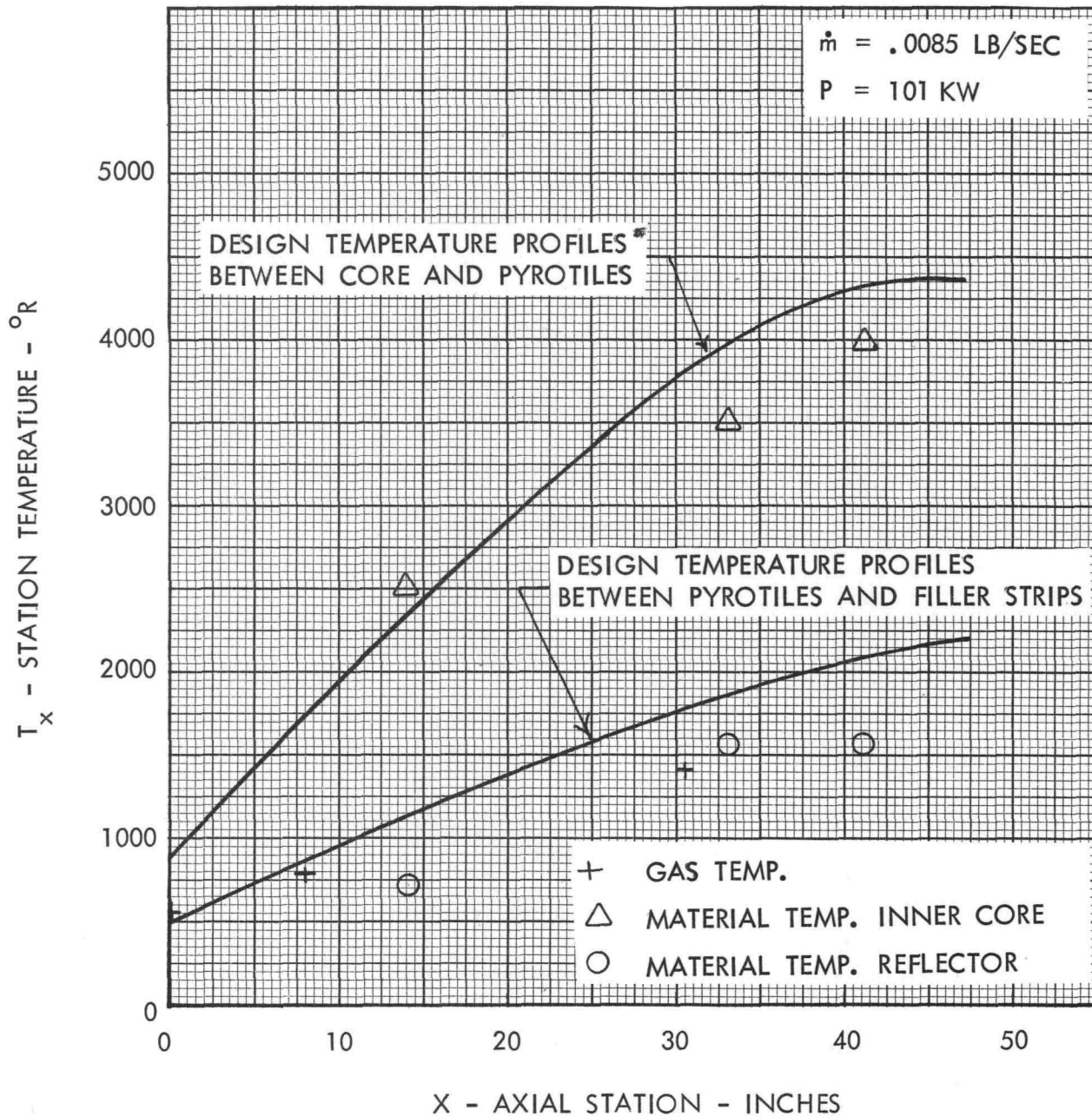




SCHEMATIC CROSS-SECTION OF PROPOSED
FILLER STRIP CORROSION TEST SECTION



SIX - INCH DIAMETER
FILLER STRIP CORROSION TEST MODEL

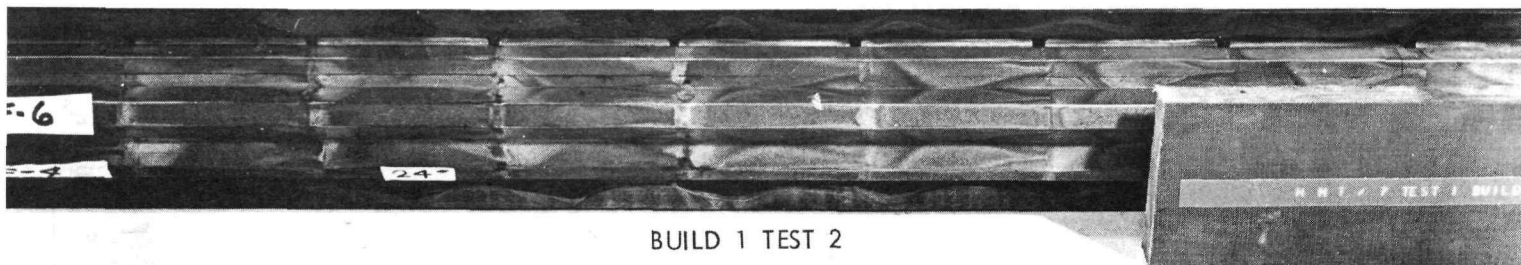


TYPICAL AXIAL TEMPERATURE MEASUREMENTS COMPARED WITH DESIGN DISTRIBUTION

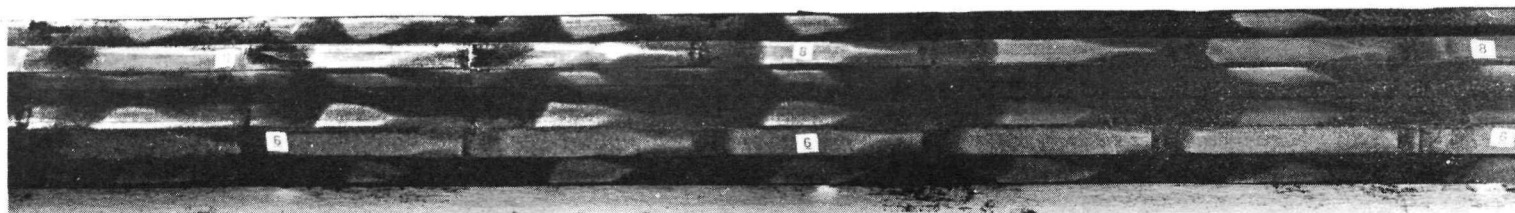
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BUILD 1 TEST 2



BUILD 1 TEST 4



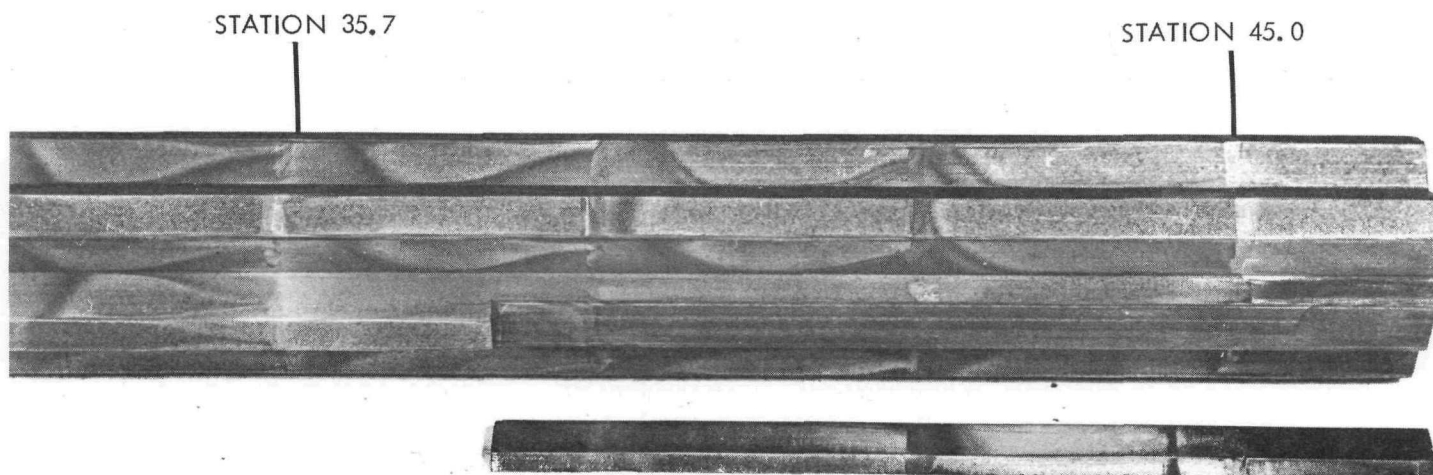
COMPARISON OF TYPICAL FLOW AND CORROSION PATTERNS OBTAINED UNDER
NRX-A TYPE FILLER STRIP ASSEMBLIES

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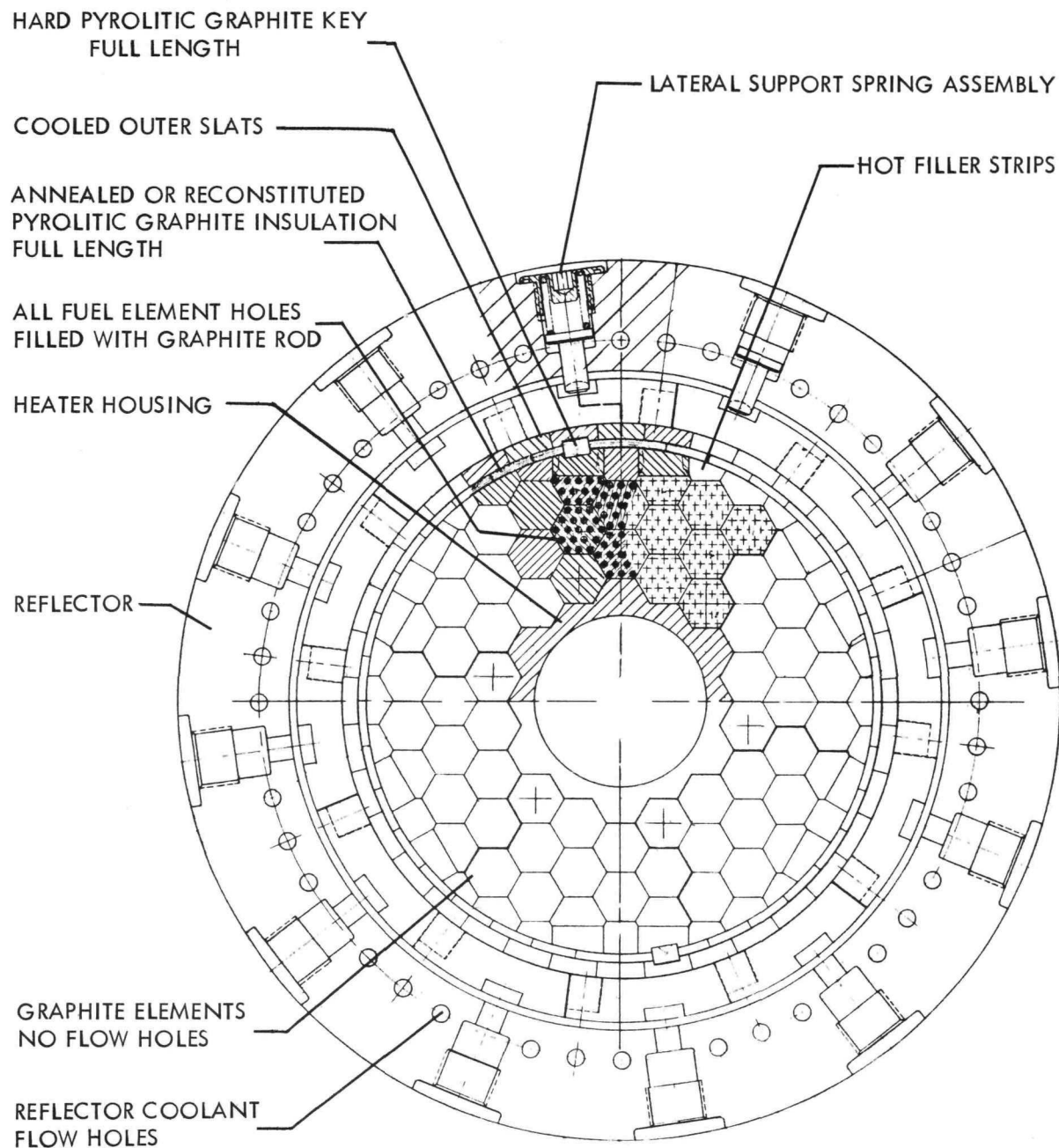
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BUILD 1 TEST 2

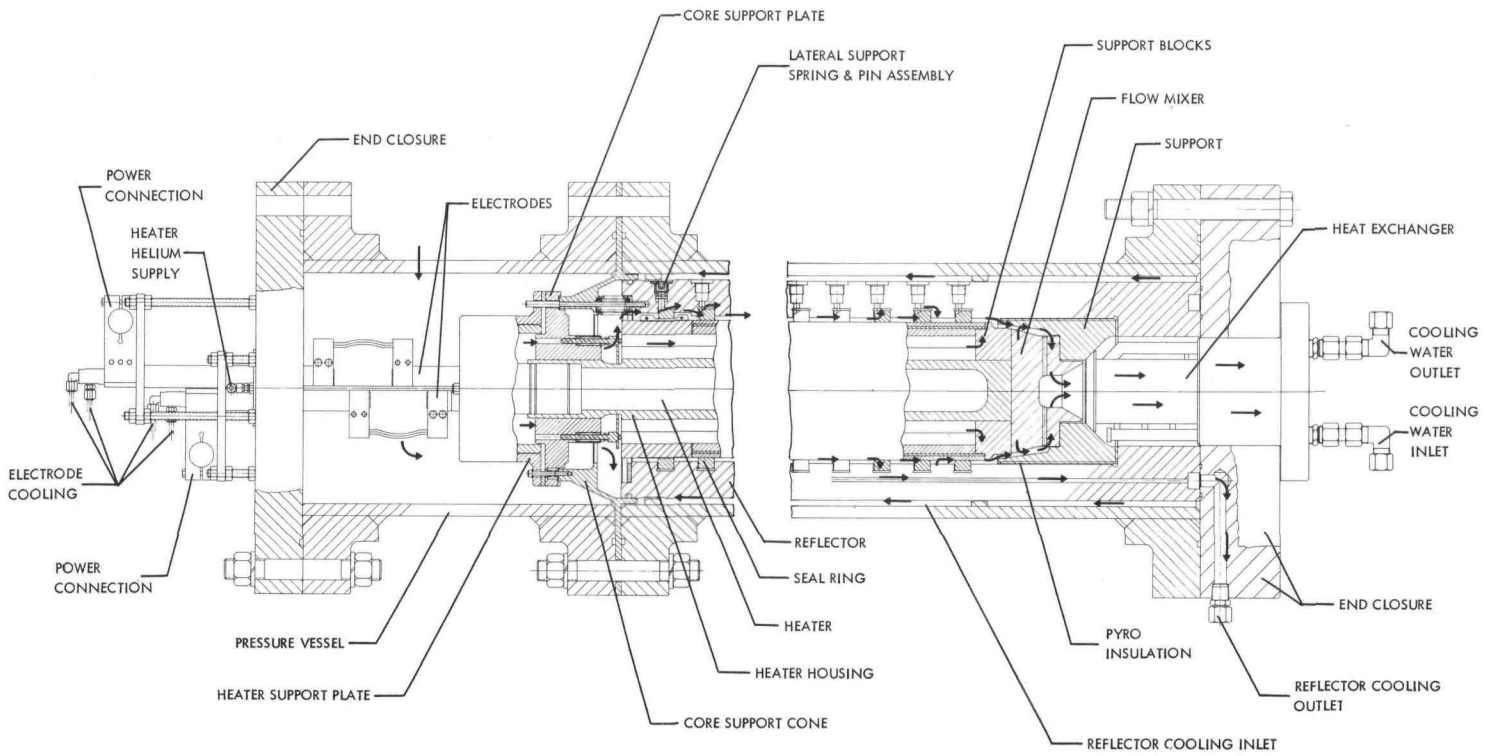


VIEW OF THE HOT END OF CORE SHOWING TEST SPECIMEN

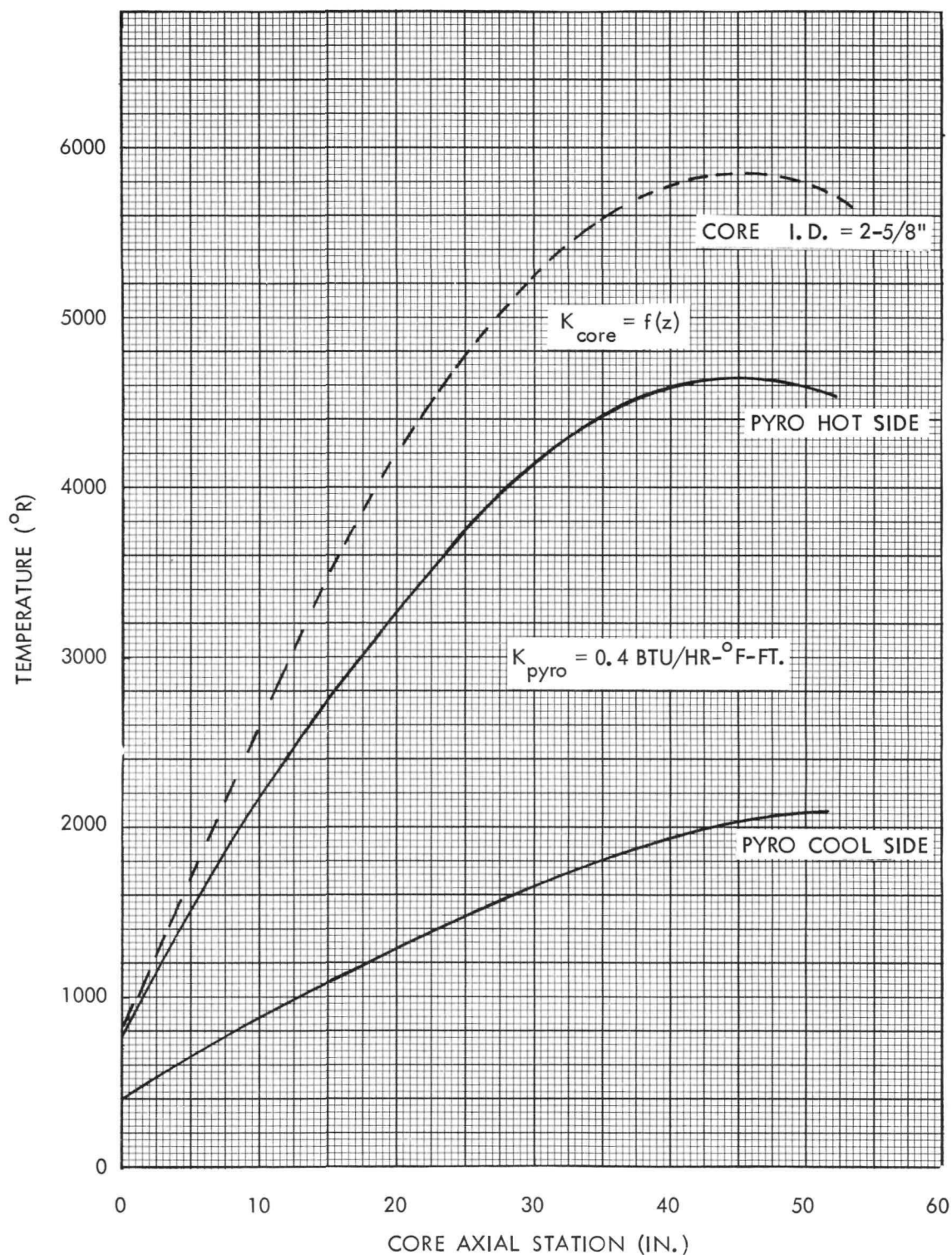


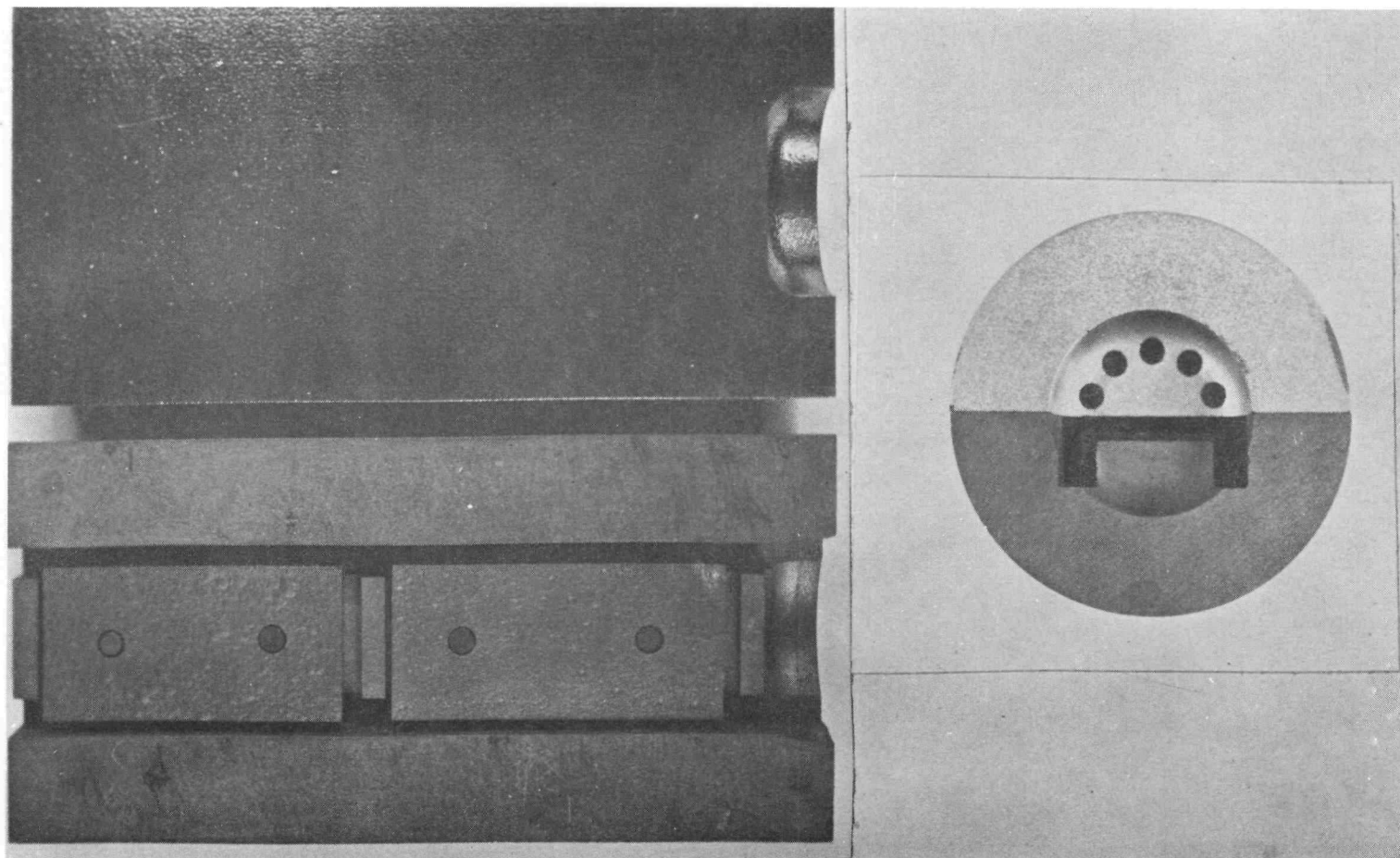
603452

HHT-15 MULTIPLE CLUSTER THERMAL TEST
CROSS SECTION OF MODEL

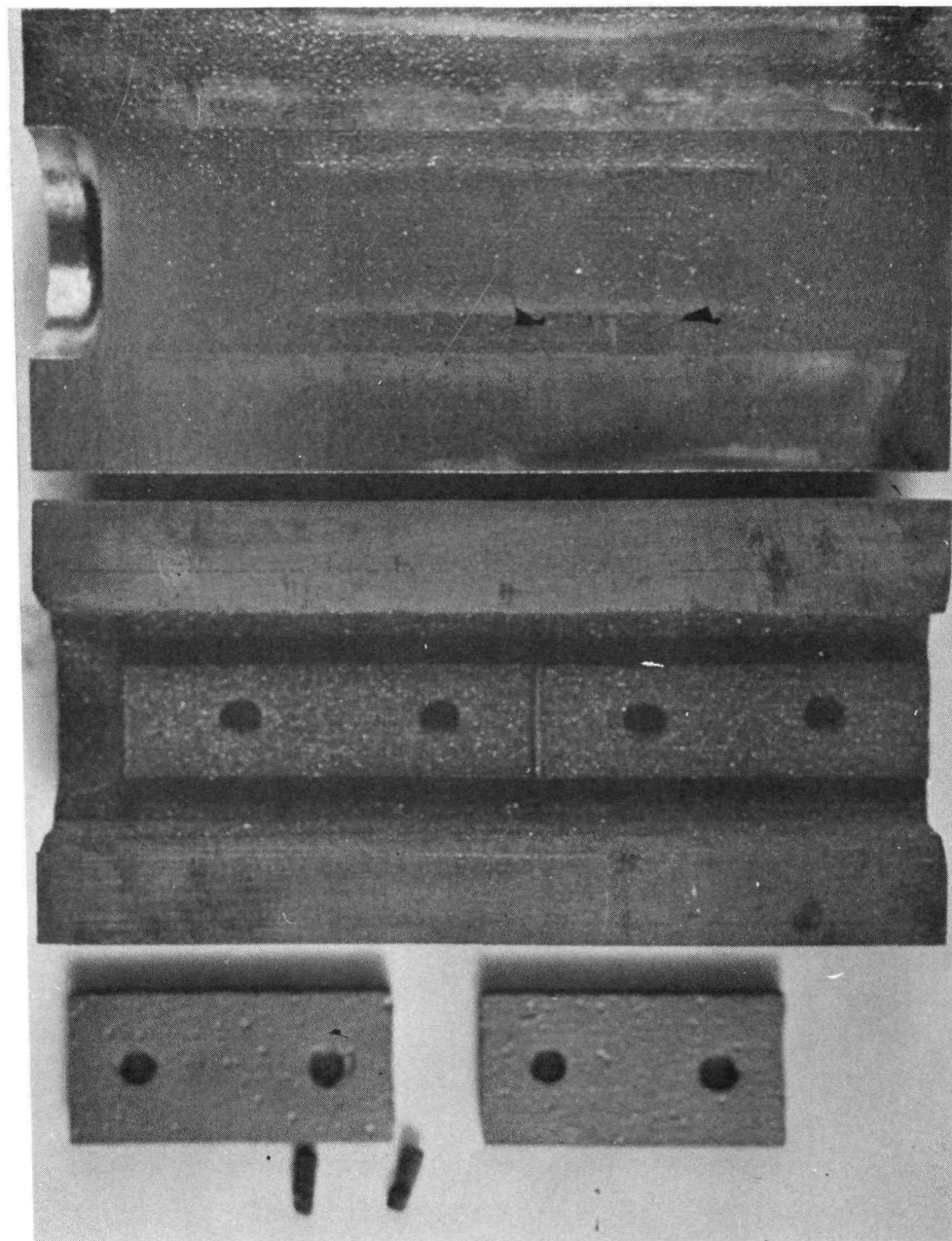


603451

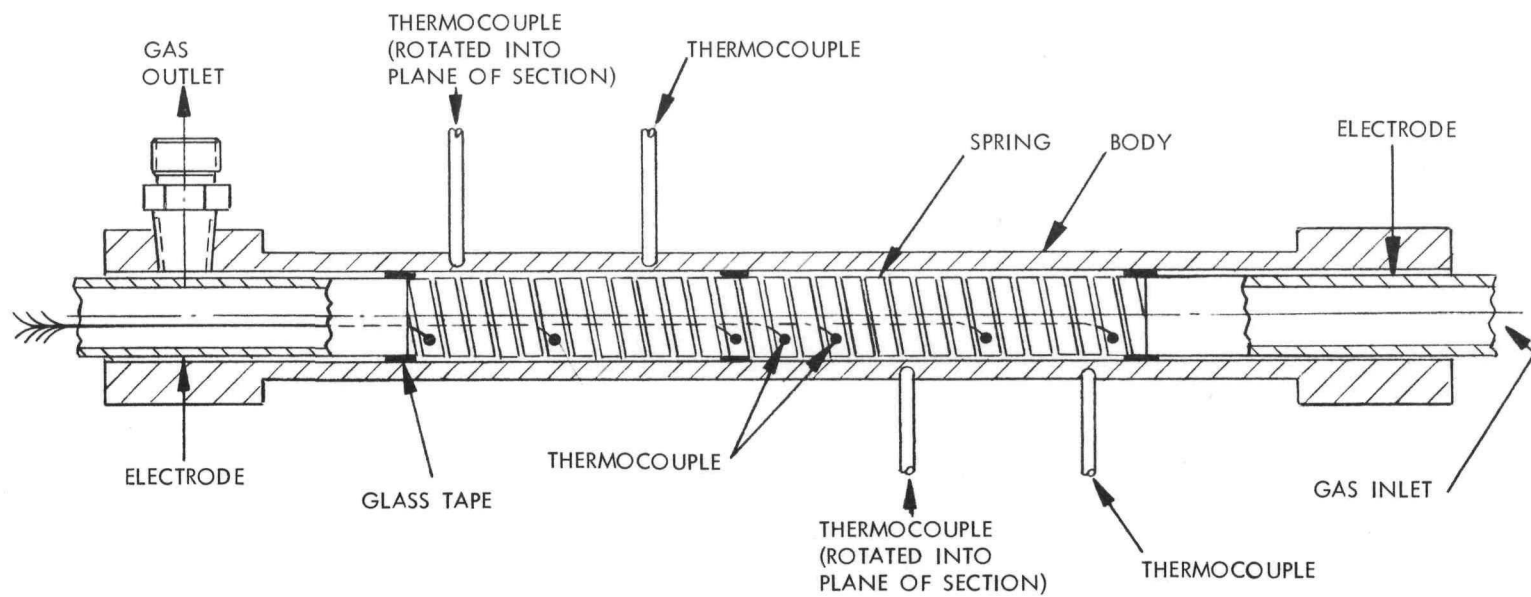
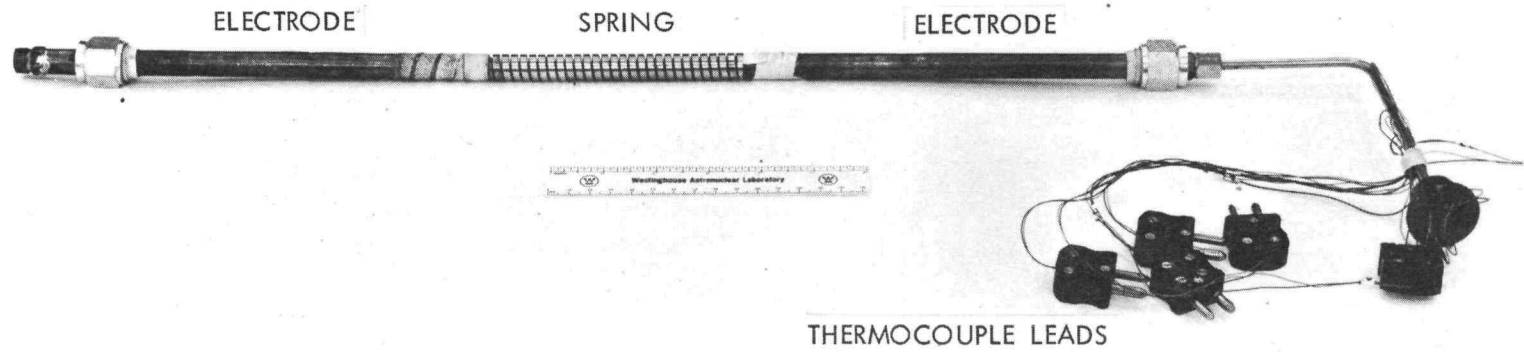




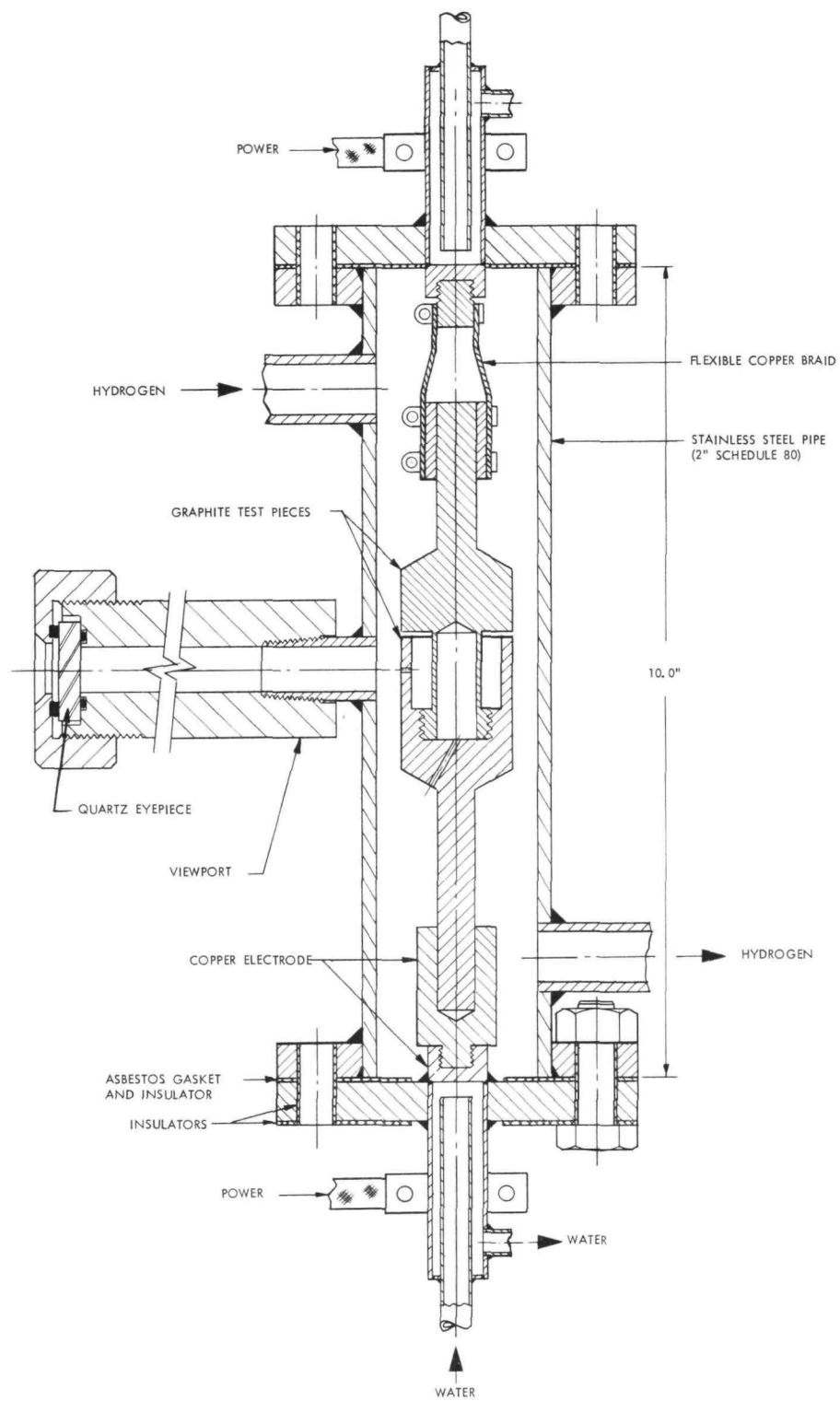
SIMULATED SUPPORT BLOCK AND FILLER STRIP ASSEMBLY

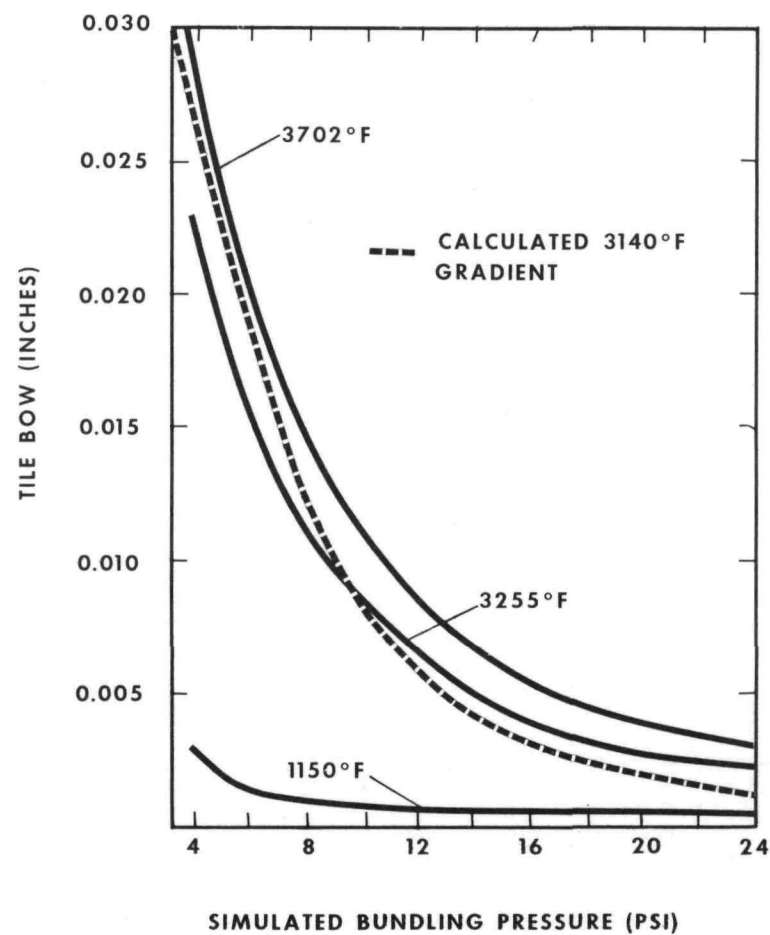
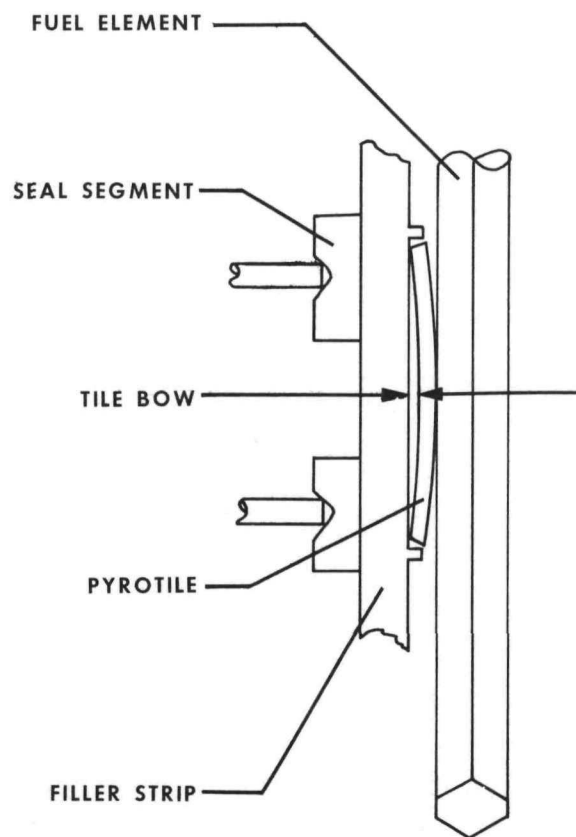


PYROLYTIC GRAPHITE PINS FOR TILE RETENTION



HOT SPRING TEST SPECIMEN

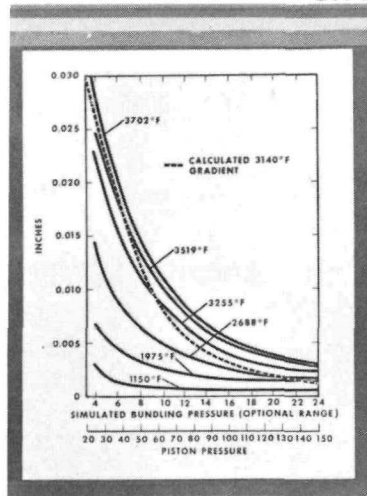




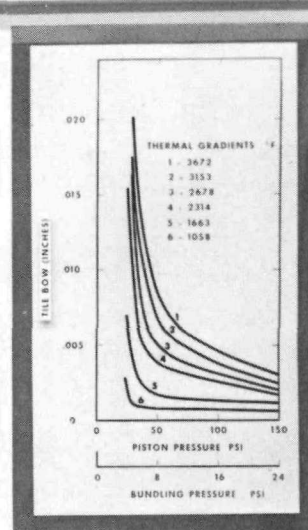
CORE INSULATION BOW TEST

603800D

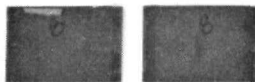
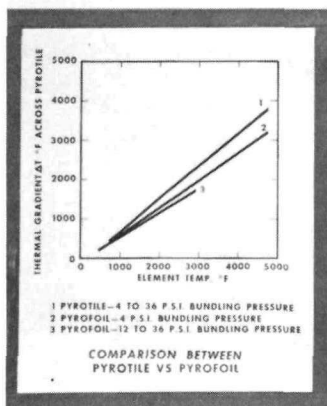
ENGINEERING MECHANICS LABORATORY



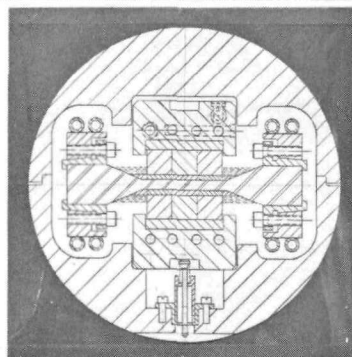
A2-A4 TILE BOW TEST



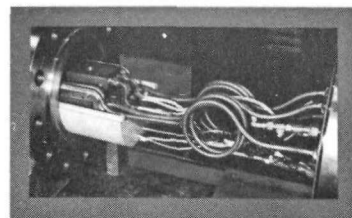
A3-A4 TILE BOW TEST



PYROFOIL INSULATION TEST

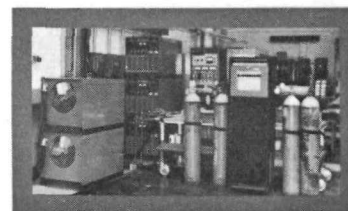


CROSS SECTION OF TEST SETUP

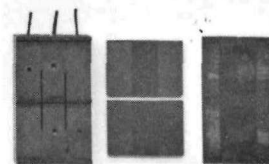


INTERNAL VIEW OF TEST SETUP

CORE INSULATION THERMAL TESTS



CORE INSULATION THERMAL TEST FACILITIES

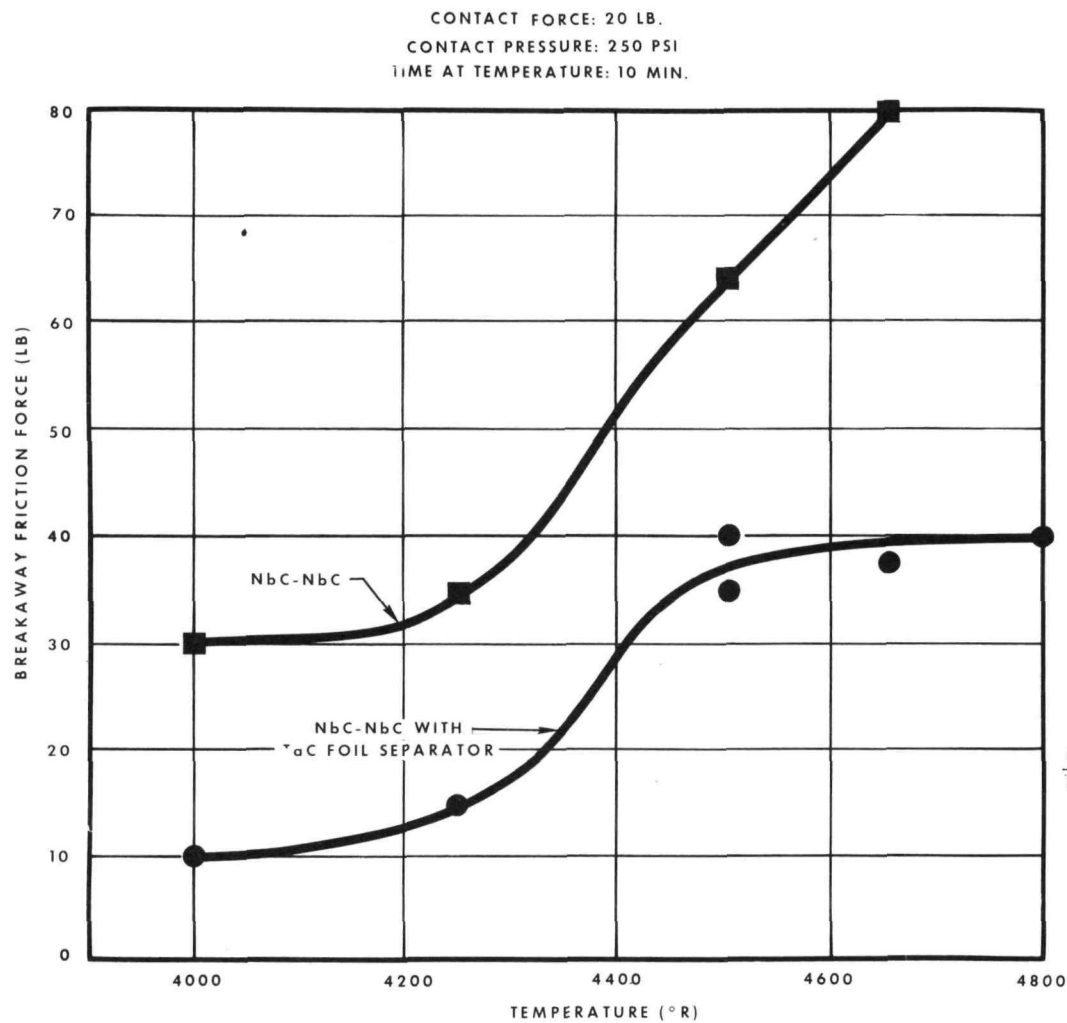


CORROSION TEST HARDWARE

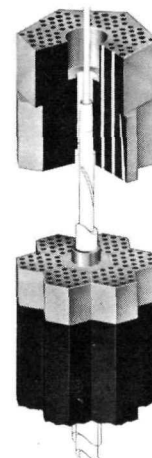
CORE PERIPHERY INTERFACE RELATIVE MOTION TESTS

Test Temperature: 4500°R
Contact Pressure: 15 psi
Time At Temperature: 10 min - 3 Cycles
Relative Displacement: 0.25 in.

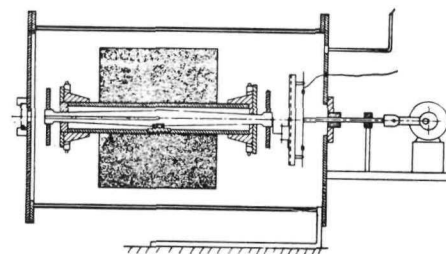
<u>INTERFACE</u>	<u>COEFFICIENT OF FRICTION</u>	<u>SURFACE CONDITION</u>
ZrC COATED F. E. - PYROTILE	1.3	SOME COATING LOST
ZrO ₂ COATED F. E. - PYROTILE	4.8	SOME COATING LOST
NbC COATED F. E. - PYROTILE	1.0	NO COATING DAMAGE
UNCOATED F. E. - PYROTILE	0.3	F. E. SURFACE CORRODED
NbC COATED F. E. - NbC COATED F. E.	2.6	NO COATING DAMAGE
NbC COATED F. E. - UNCOATED F. E.	4.5	F. E. SURFACE CORRODED
UNCOATED F. E. - UNCOATED F. E.	0.3	SURFACES CORRODED



FUEL ELEMENT SUPPORT BLOCK INTERFACE
 BONDING AND COATING WEAR TESTS



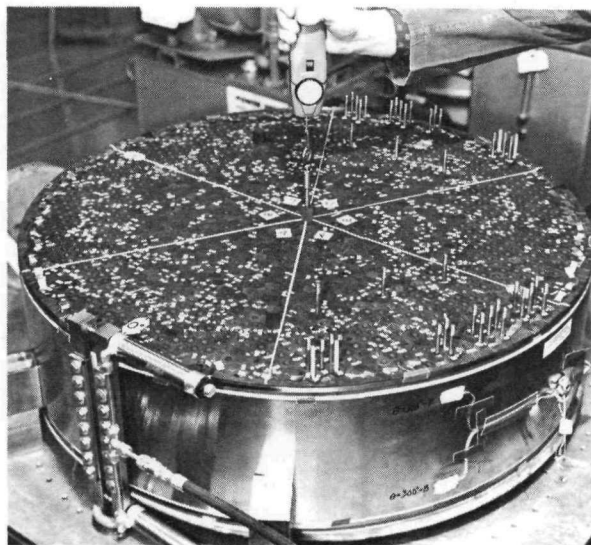
FUEL ELEMENT CLUSTER-
 SUPPORT BLOCK INTERFACE



SCHEMATIC OF TEST SETUP

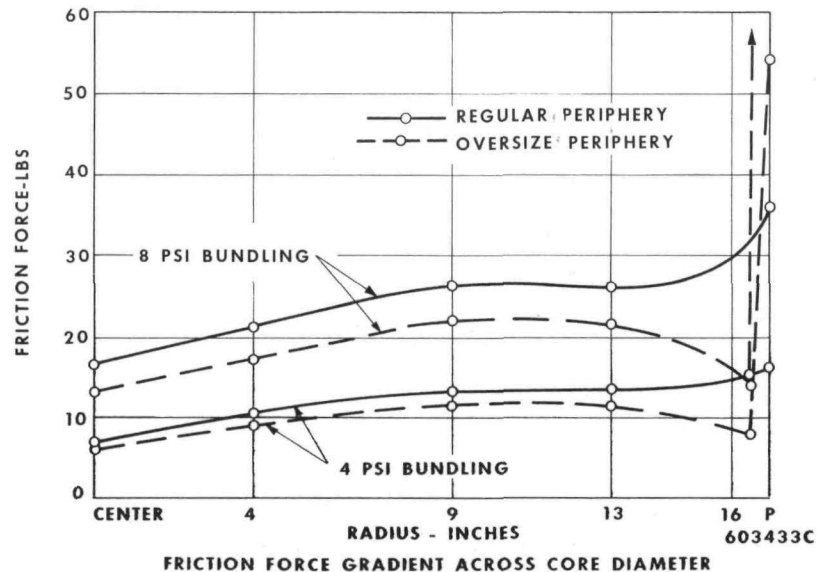
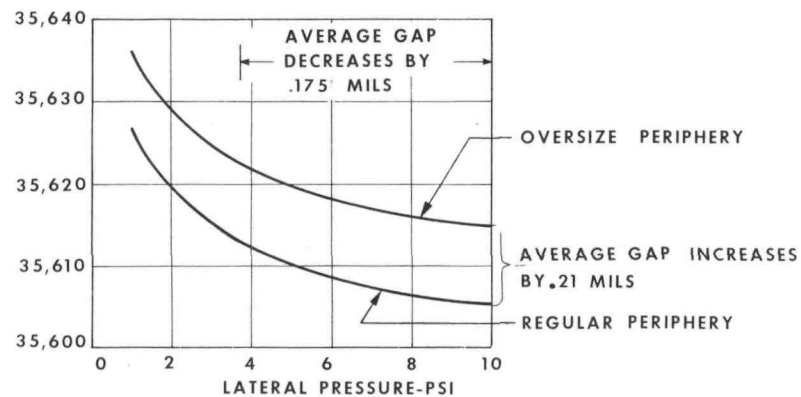
603480D

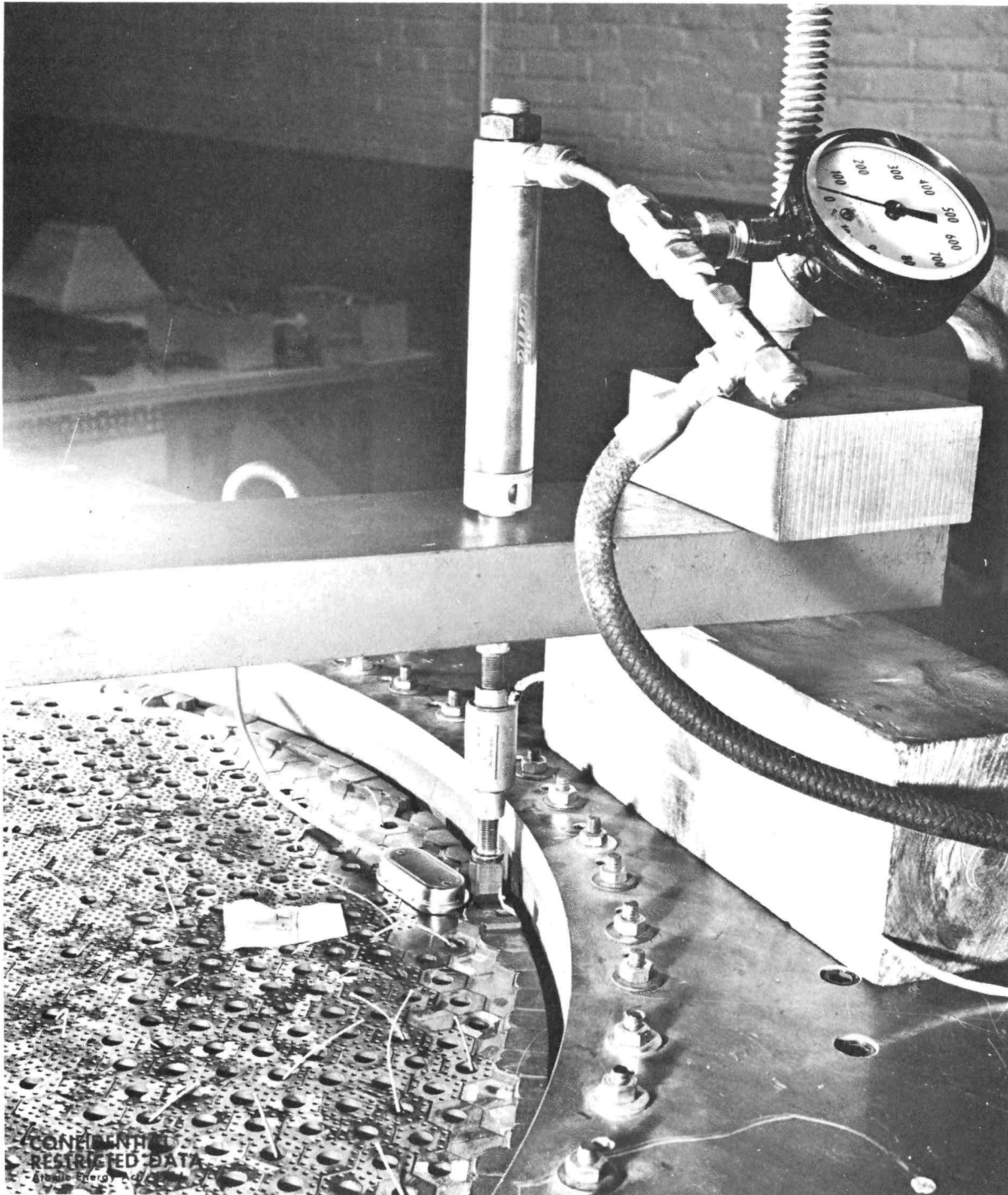
EFFECT OF COATED PERIPHAL ELEMENTS ON CORE PACKING



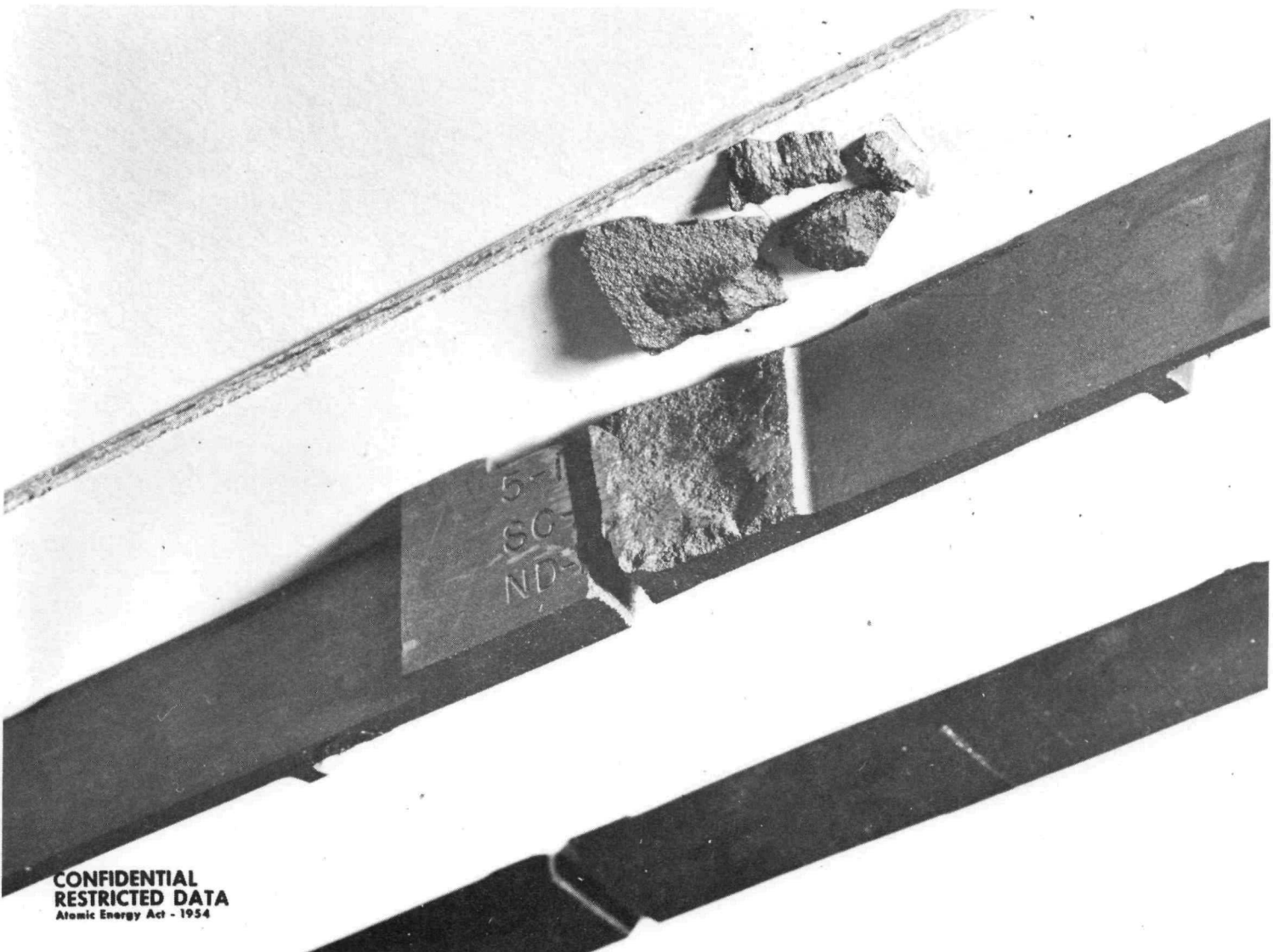
TEST SETUP

CORE DIAMETER VS BUNDLING PRESSURE





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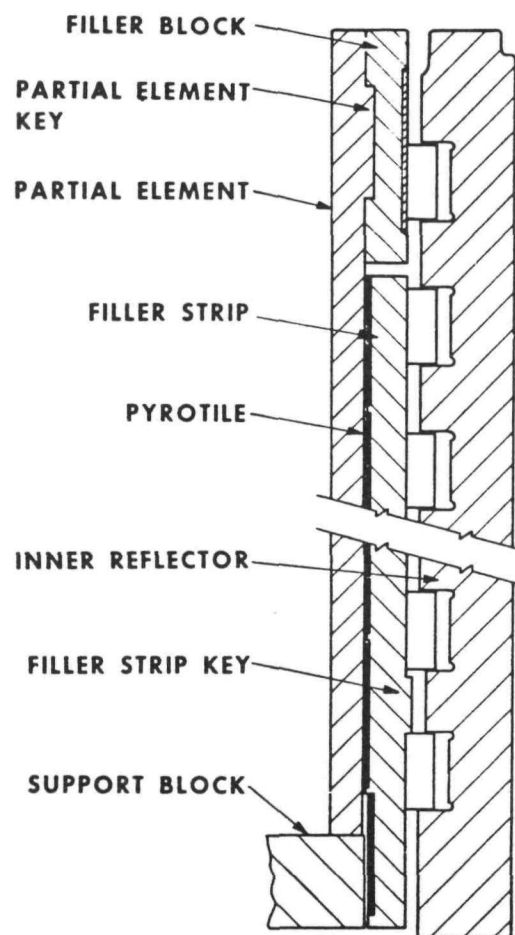
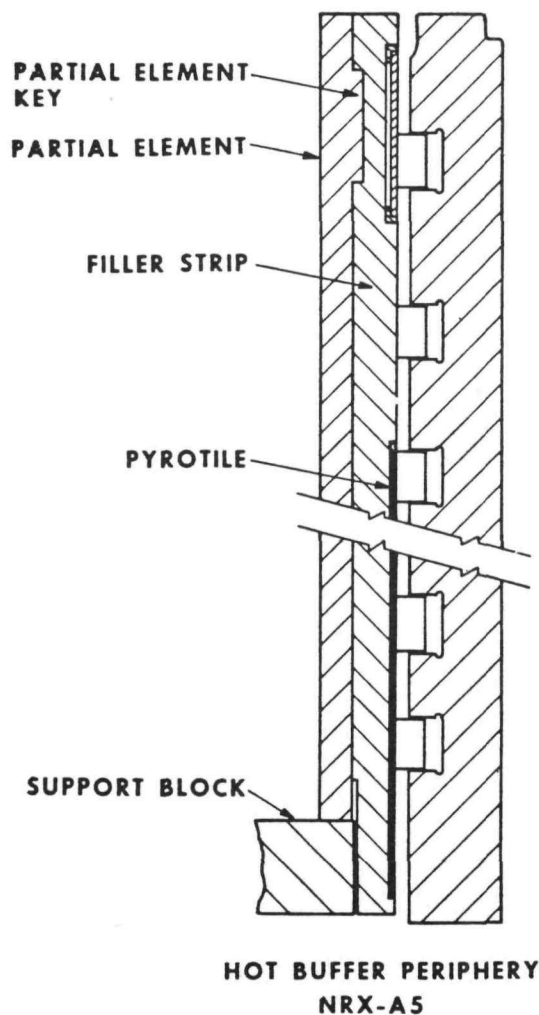


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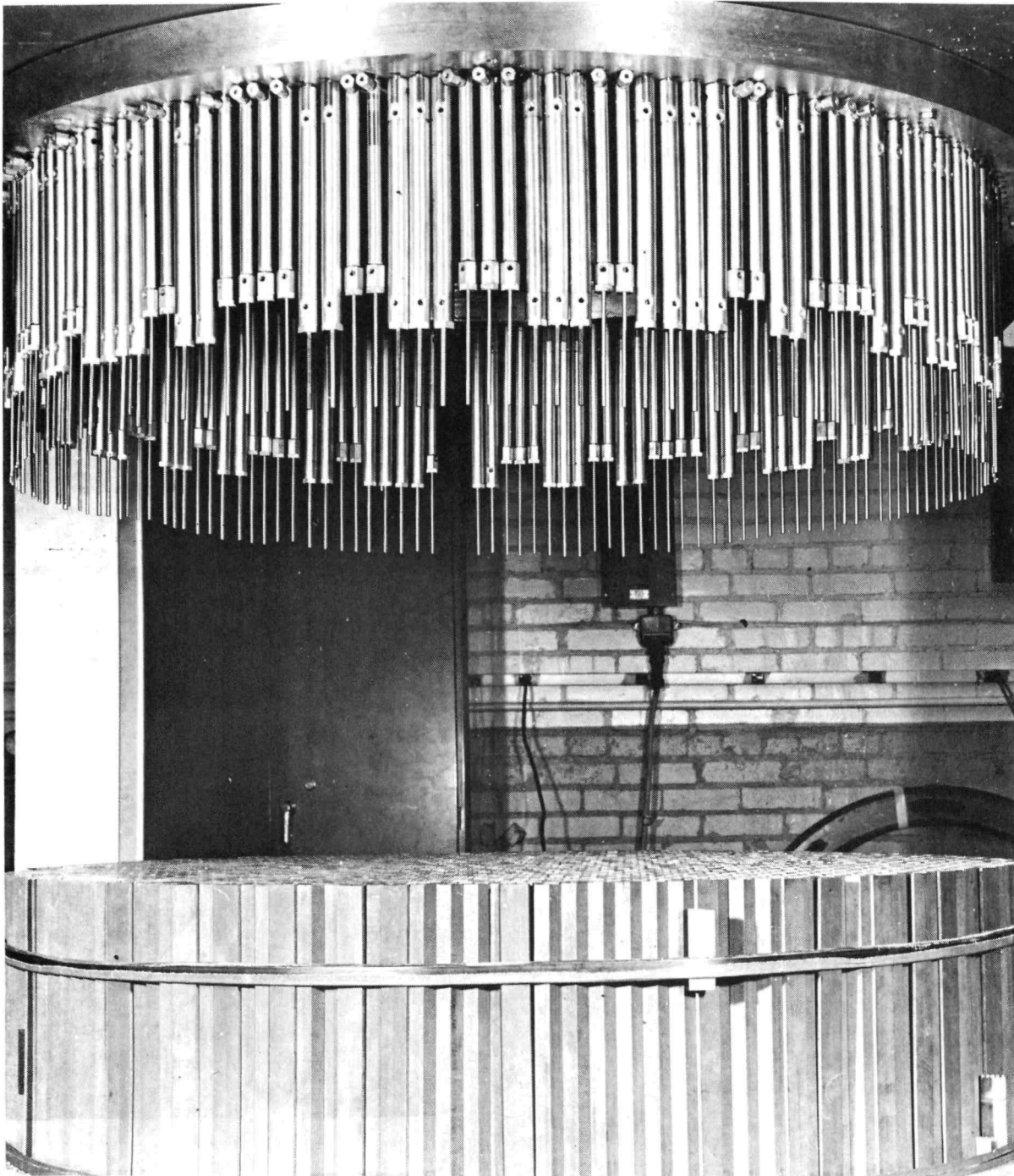
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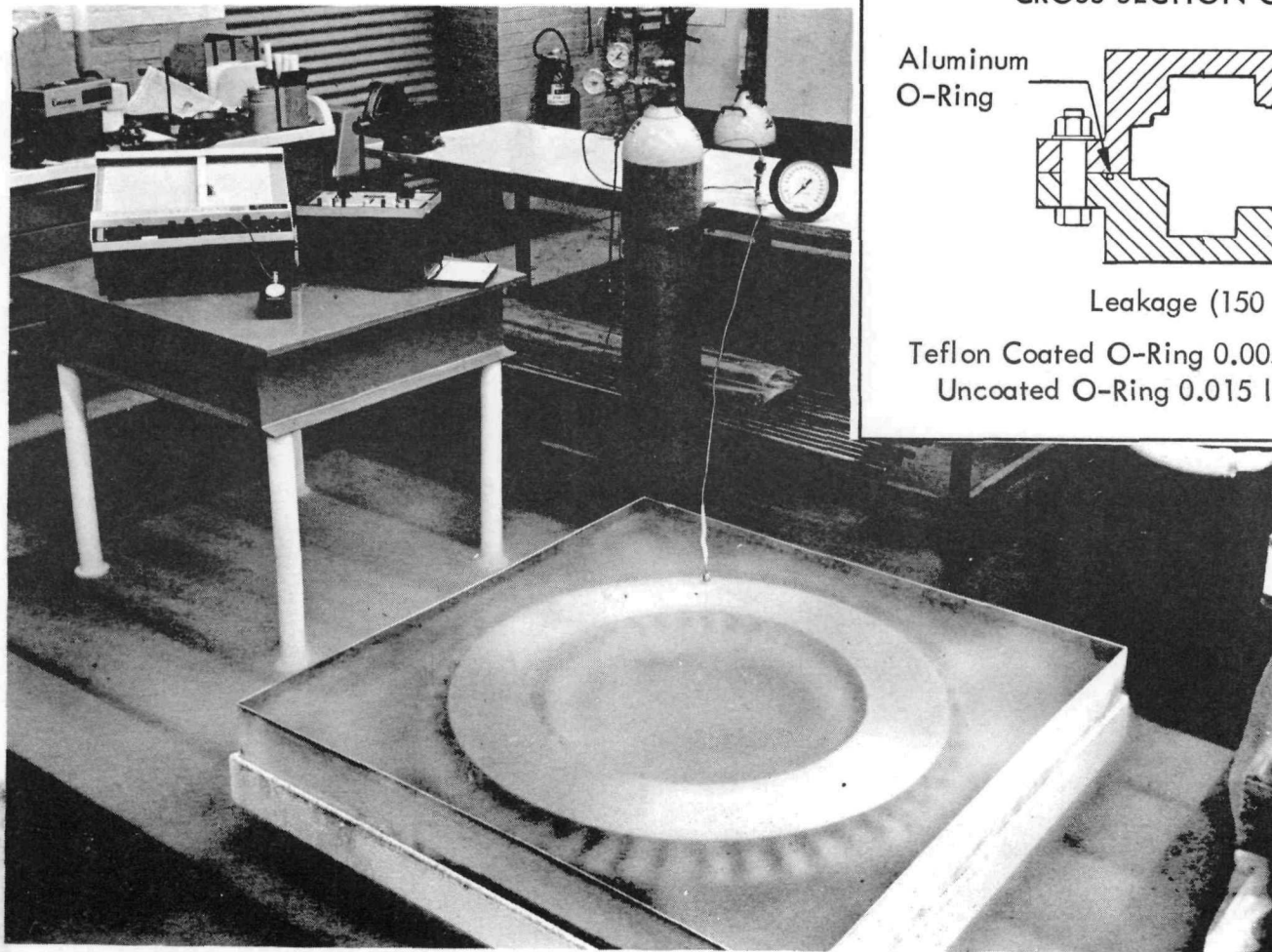
SUMMARY OF FILLER STRIP TEST RESULTS

FILLER STRIP IDENTIFICATION NUMBER	FRACTURE LOAD (LB)
10	650
10	575
10	620
1	560
26	750
6	380
16	750
	612 LB. AVERAGE
DESIGN LOAD CRITERIA	400 LB.



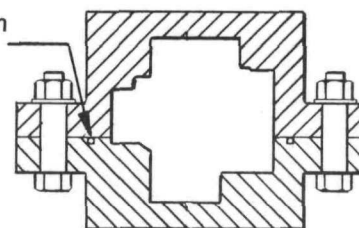
CORE PERIPHERY





CROSS SECTION OF TORUS

Aluminum
O-Ring



Leakage (150 psig)

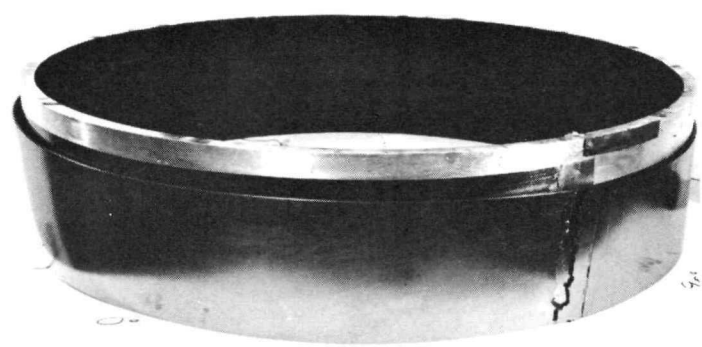
Teflon Coated O-Ring 0.005 lb/hr/ft of Seal

Uncoated O-Ring 0.015 lb/hr/ft of Seal

603460

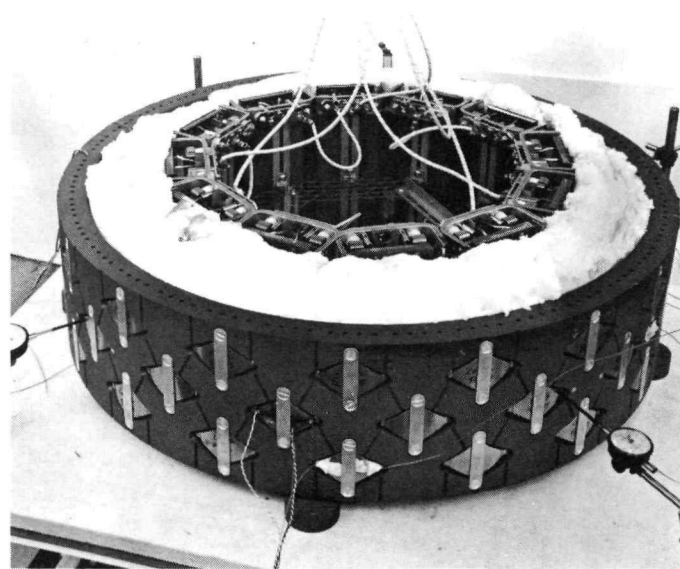
DOM E N D S E A L T E S T R I G T O R U S S E A L T E S T

PYROFOIL WRAPPER TESTING

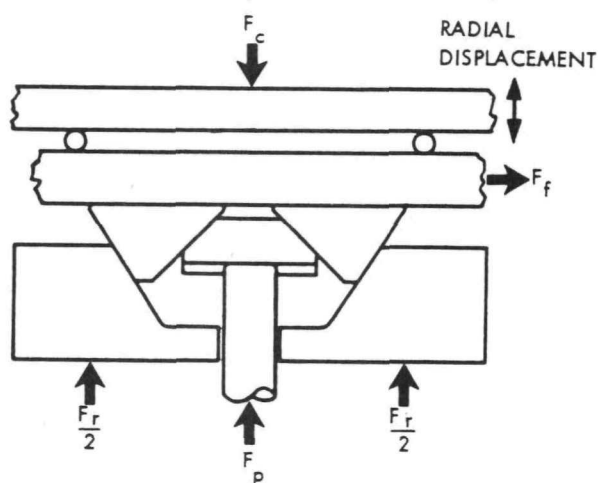


RESULTS OF 1% AVG. STRAIN

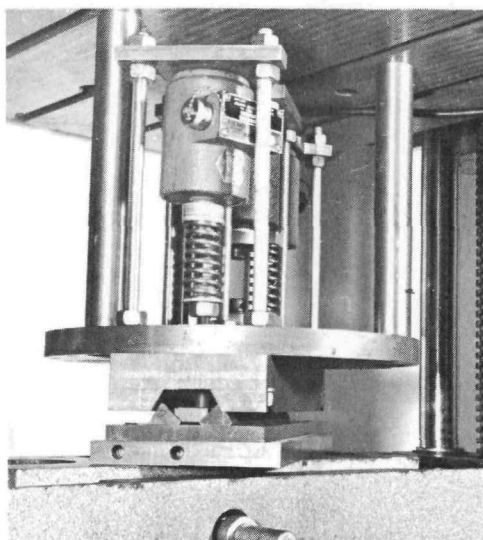
TEST SETUP



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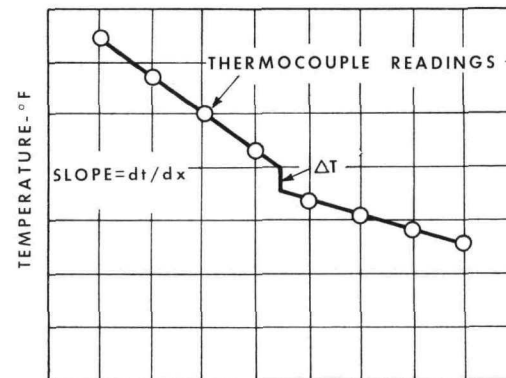
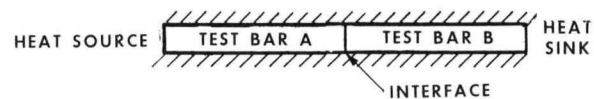
LOADING DIAGRAM



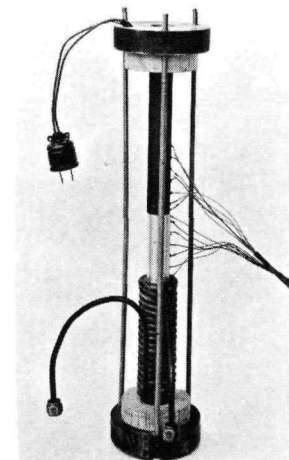
TEST SETUP

CORE PERIPHERY SEAL FRICTION AND LOCKUP TEST

THERMAL CONTACT PROPERTIES TESTING

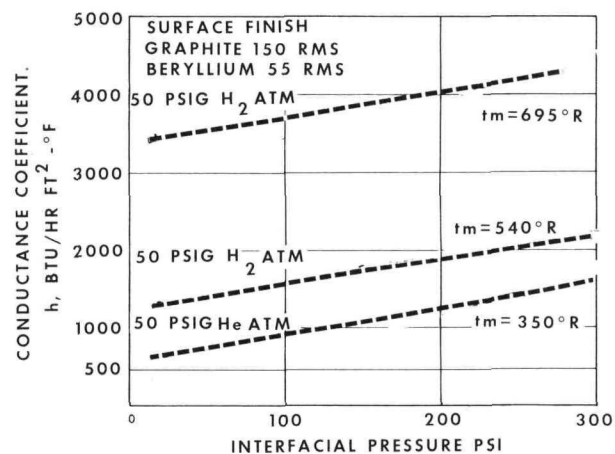


DISTANCE - FEET



TEST ASSEMBLY

EFFECT OF INTERFACE TEMPERATURE
AND CONTACT PRESSURE



THERMAL CONTACT RESISTANCE TESTS

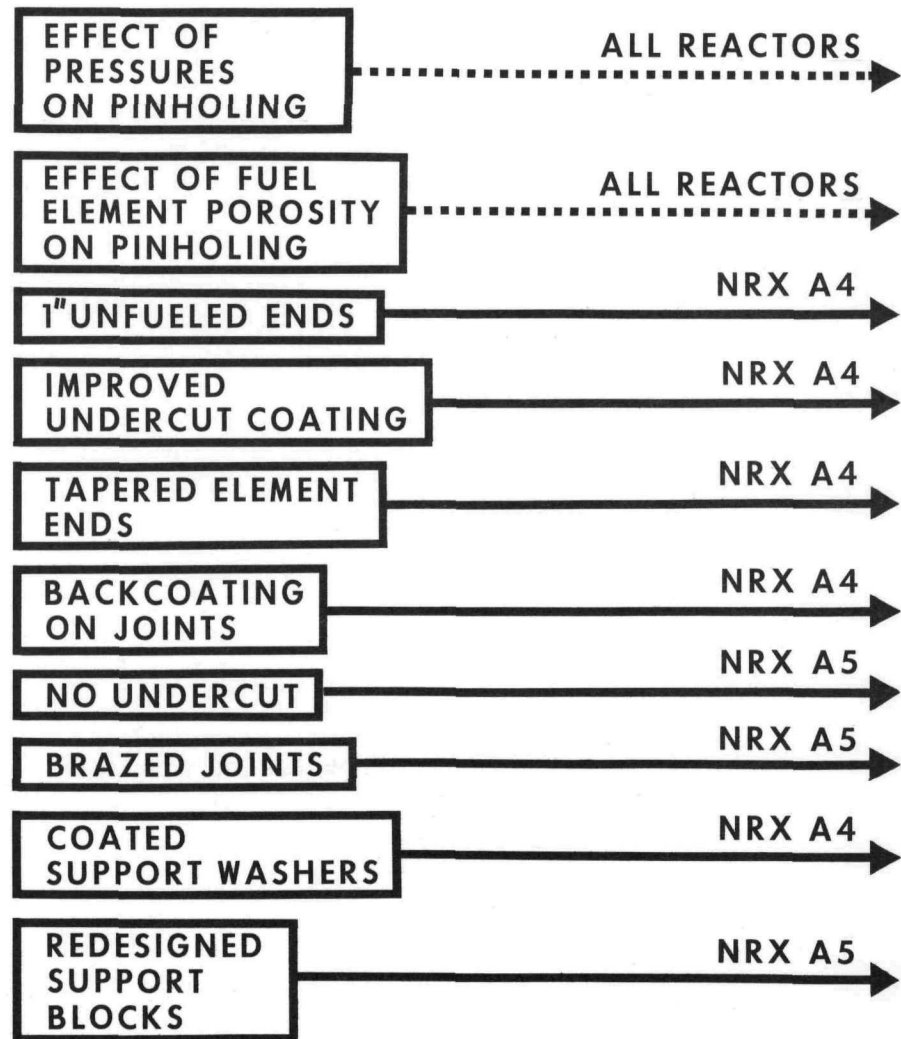
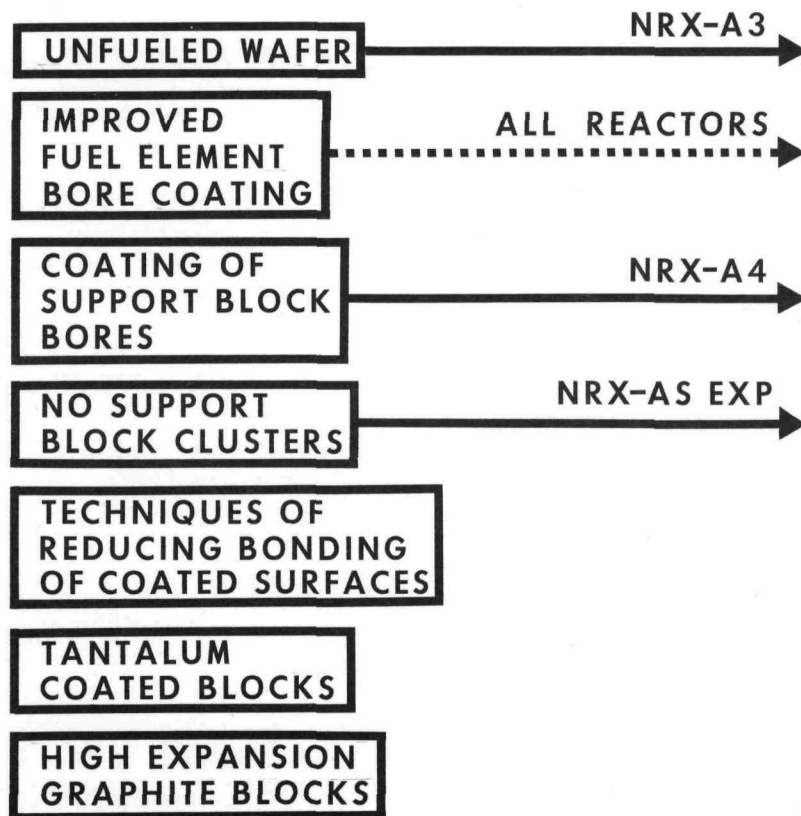
		TEMP °R	INTERFACIAL PRESS PSI	GAS
1	ALUMINUM GRAPHITE			
		350	10	H ₂
		540	50	He
		700	300	
2	BERYLLIUM GRAPHITE			
		350	10	H ₂
		540	50	He
		700	300	

603434C

FUEL ELEMENT & SUPPORT BLOCK CORROSION

NRX A2

NRX A3



SINGLE ELEMENT HYDROGEN CORROSION TEST SUMMARY
OCTOBER 1, 1964 TO AUGUST 15, 1965

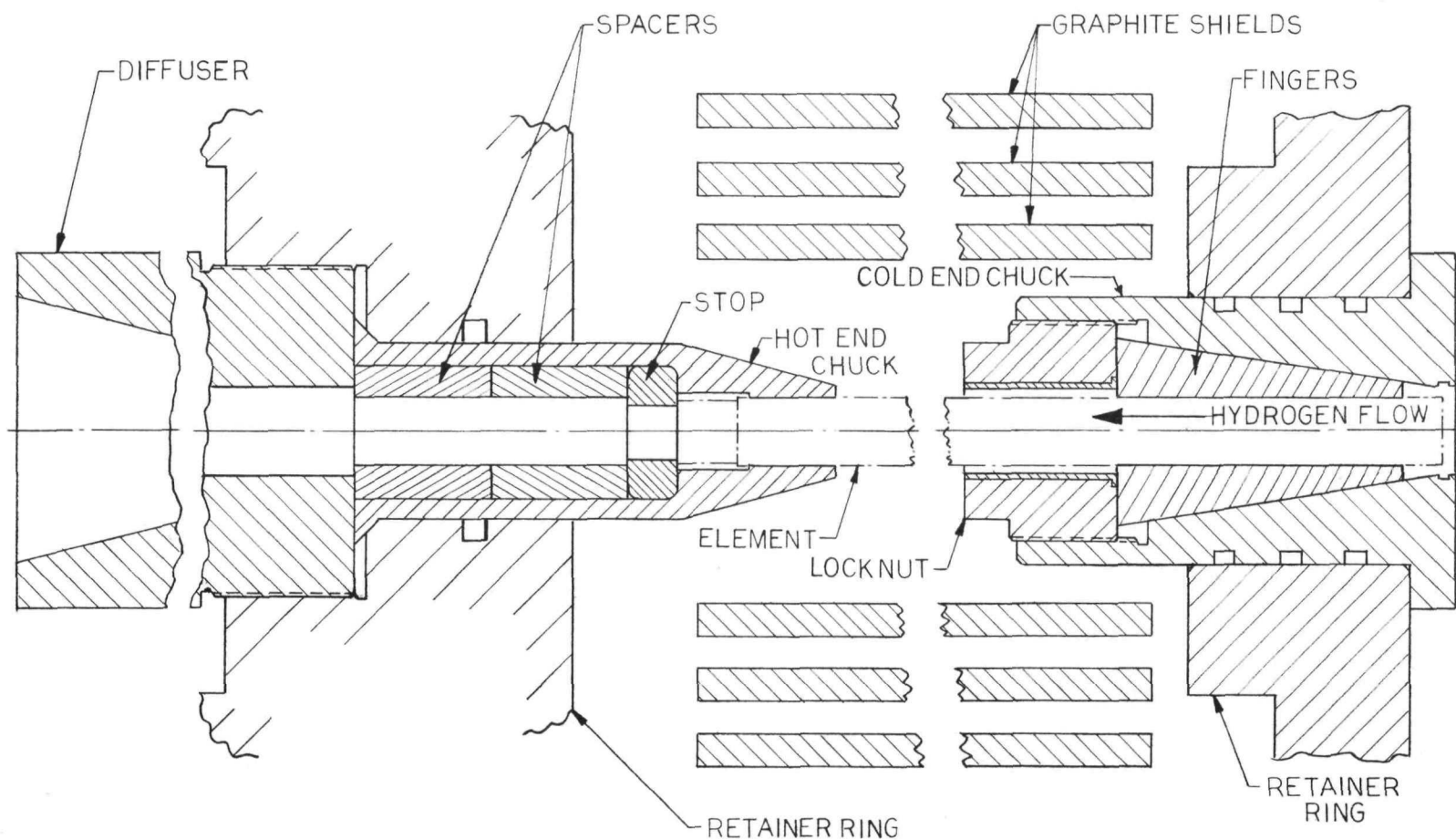
<u>Test</u>	<u>WANL</u>	<u>WANHES</u>	<u>Total</u>
1A5	168	2	170
1A10	1	---	1
1B10	211	82	293
1B5+5	165	194	359
1B15	5	3	8
1B20	131	129	260
1B30	5	18	23
1C10	33	---	33
1C10+10	1	1	2
1C20	37	11	48
1C30	4	---	4
Misc.	<u>35</u>	<u>17</u>	<u>52</u>
Total	796	457	1,253

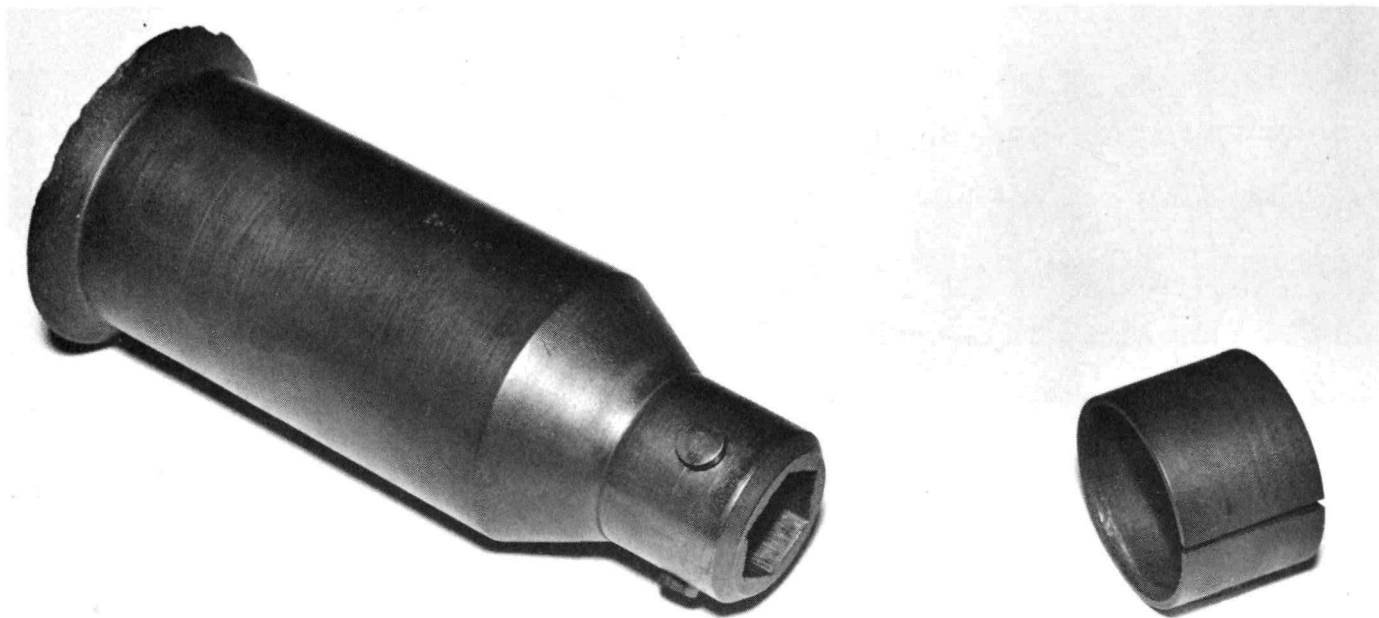
TEMPERATURE CONDITIONS

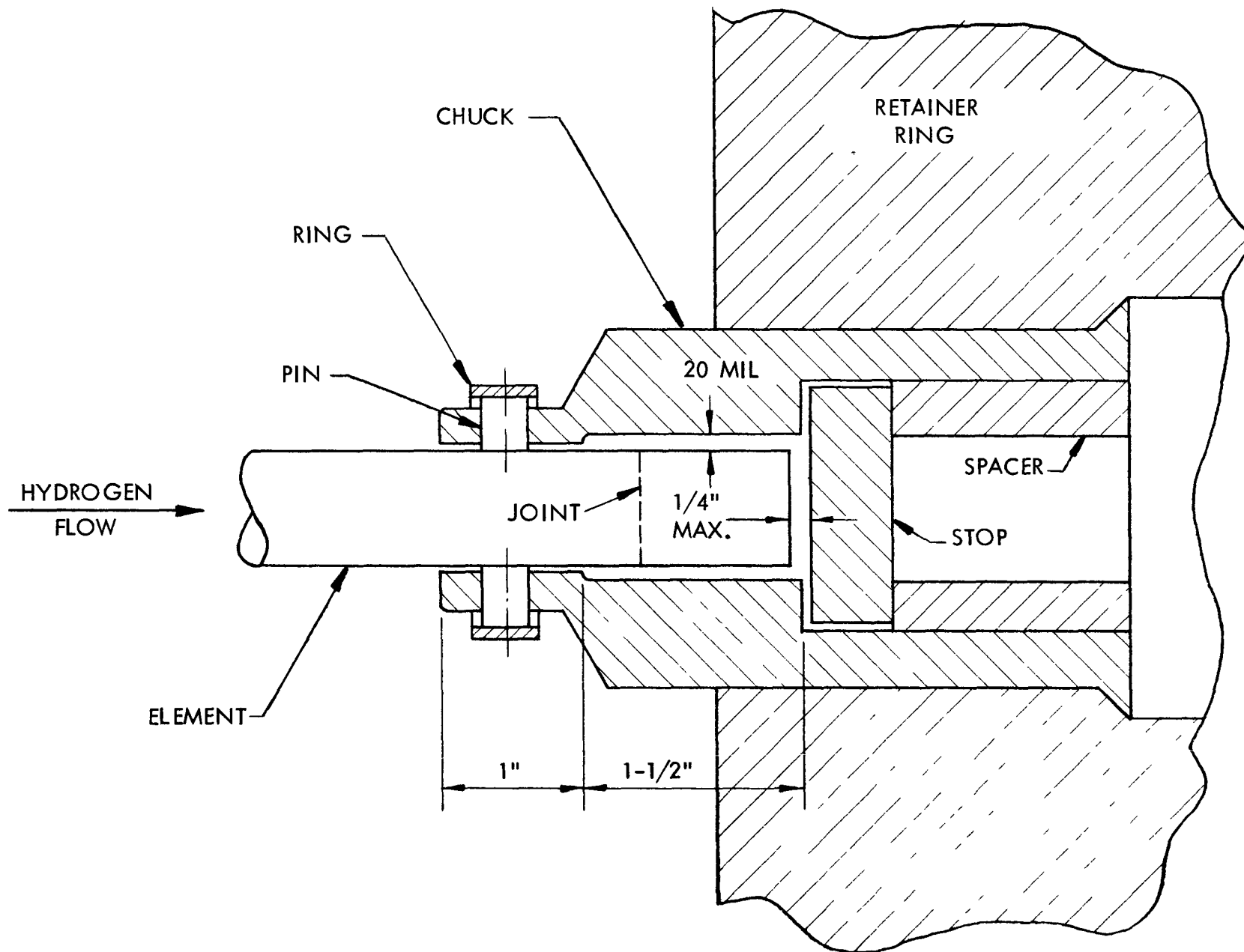
1A = surface temperature 4600°R , int. temp. = 5150

1B = surface temperature 4450°R , int. temp. = 4900

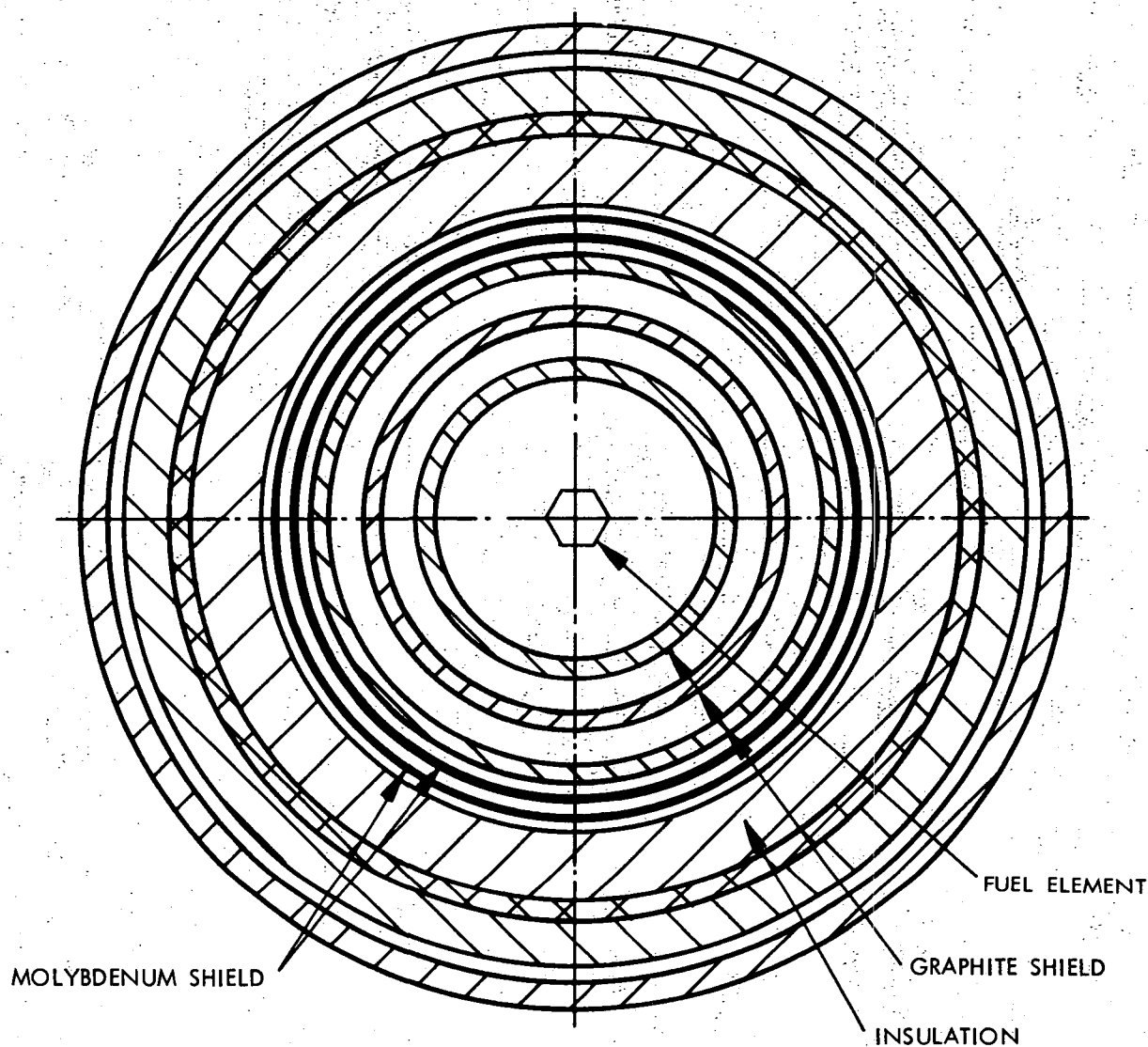
1C = surface temperature 4300°R , int. temp. = 4660



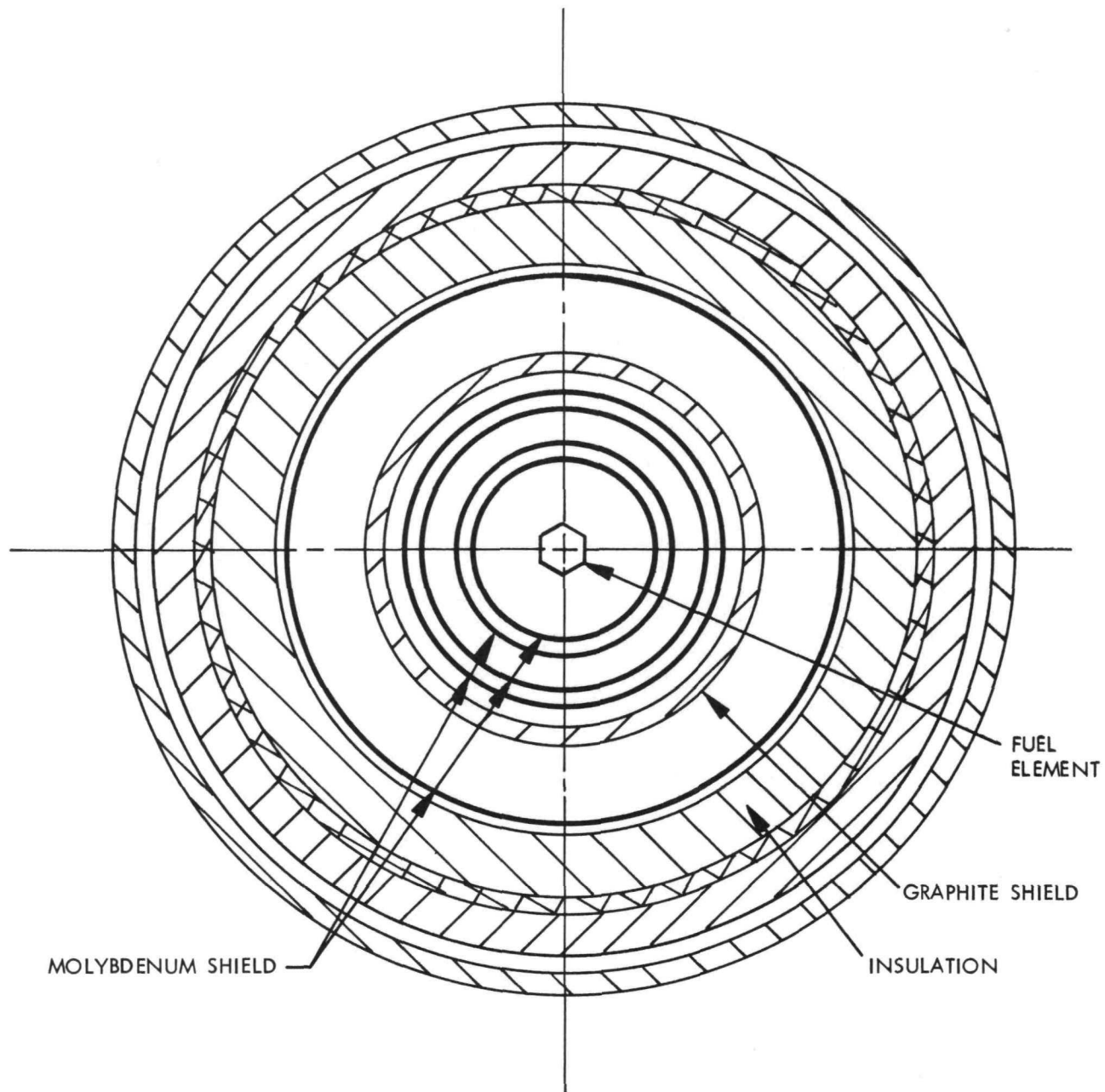




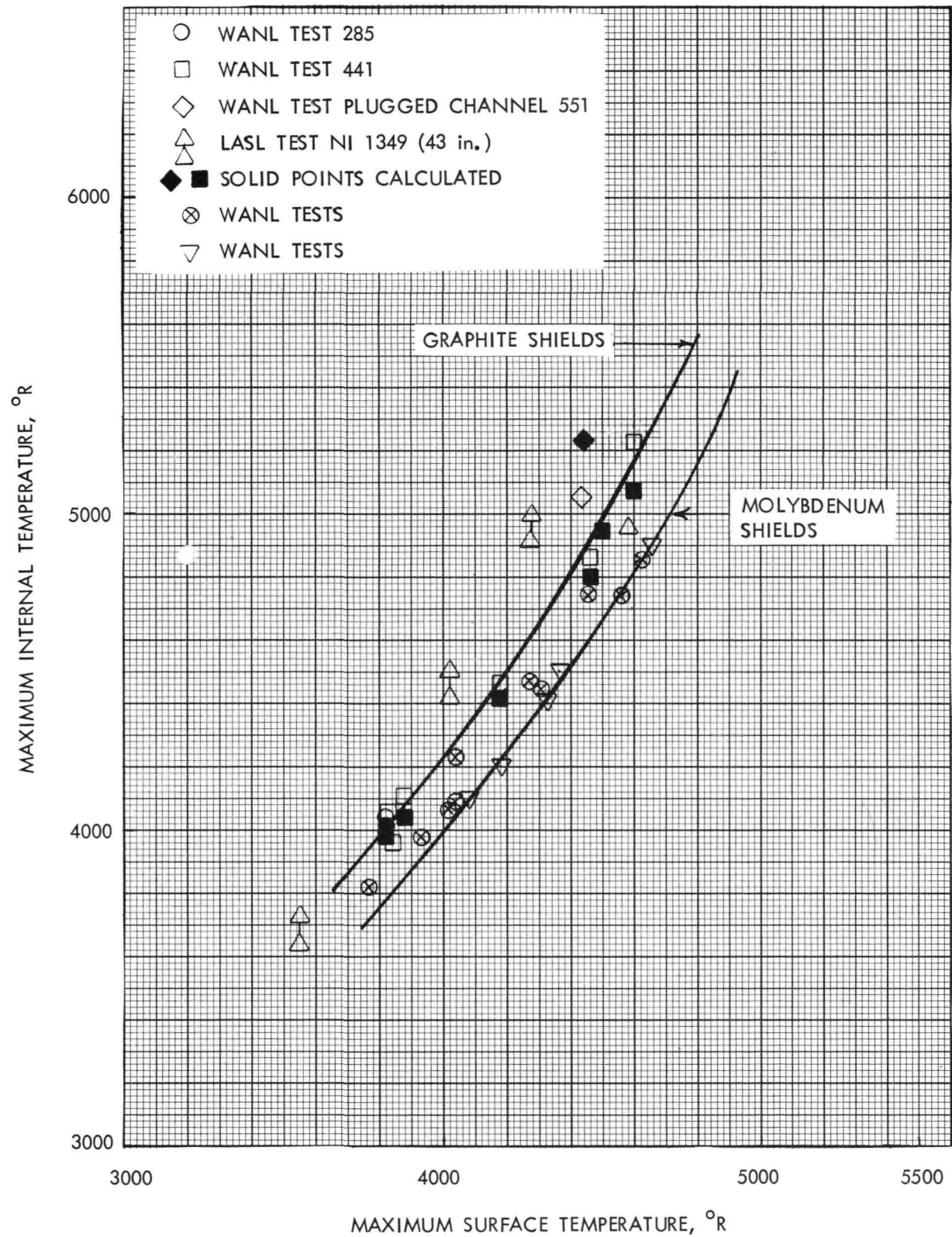
THREE PIN HOT END CHUCK ARRANGEMENT

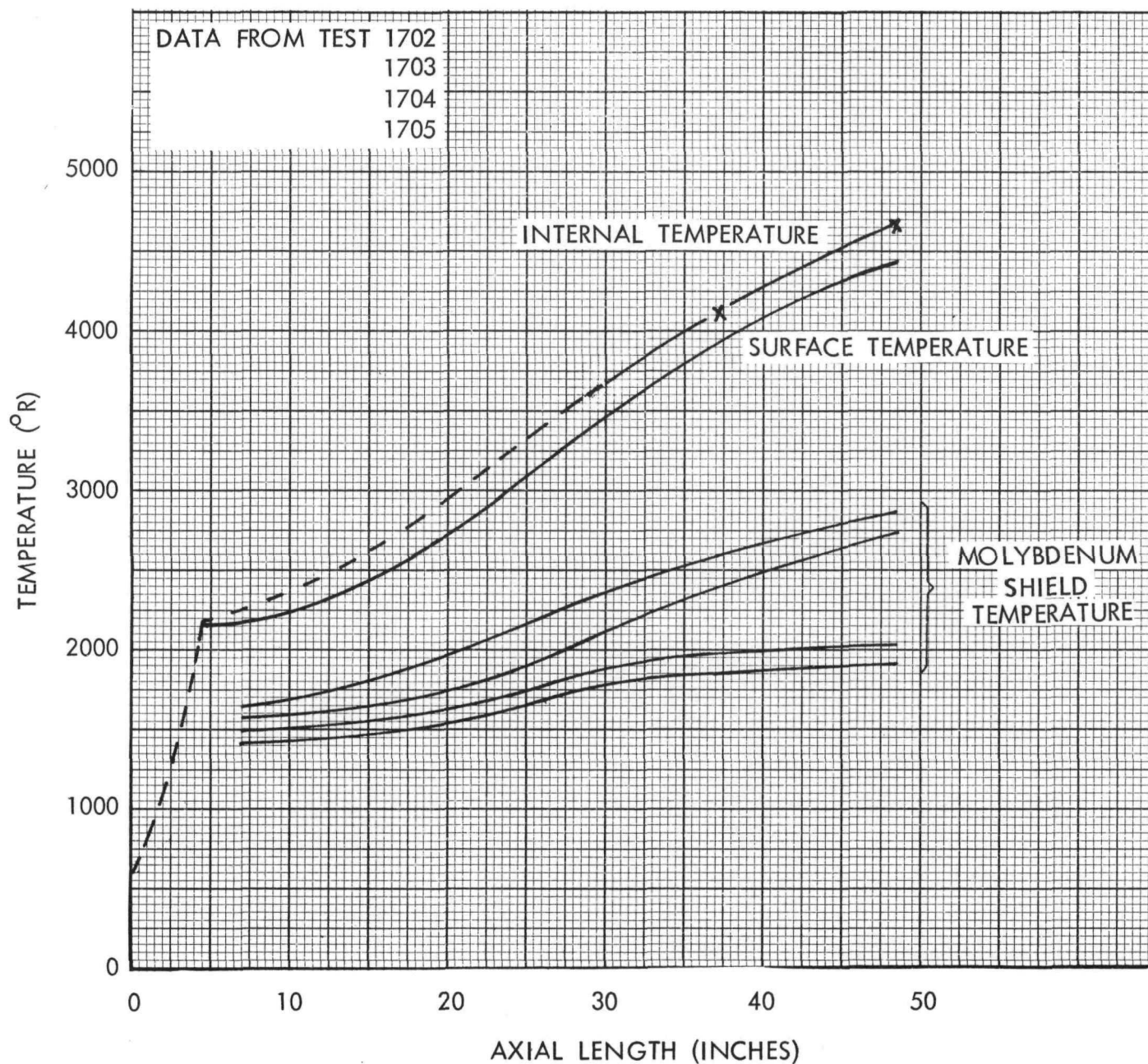


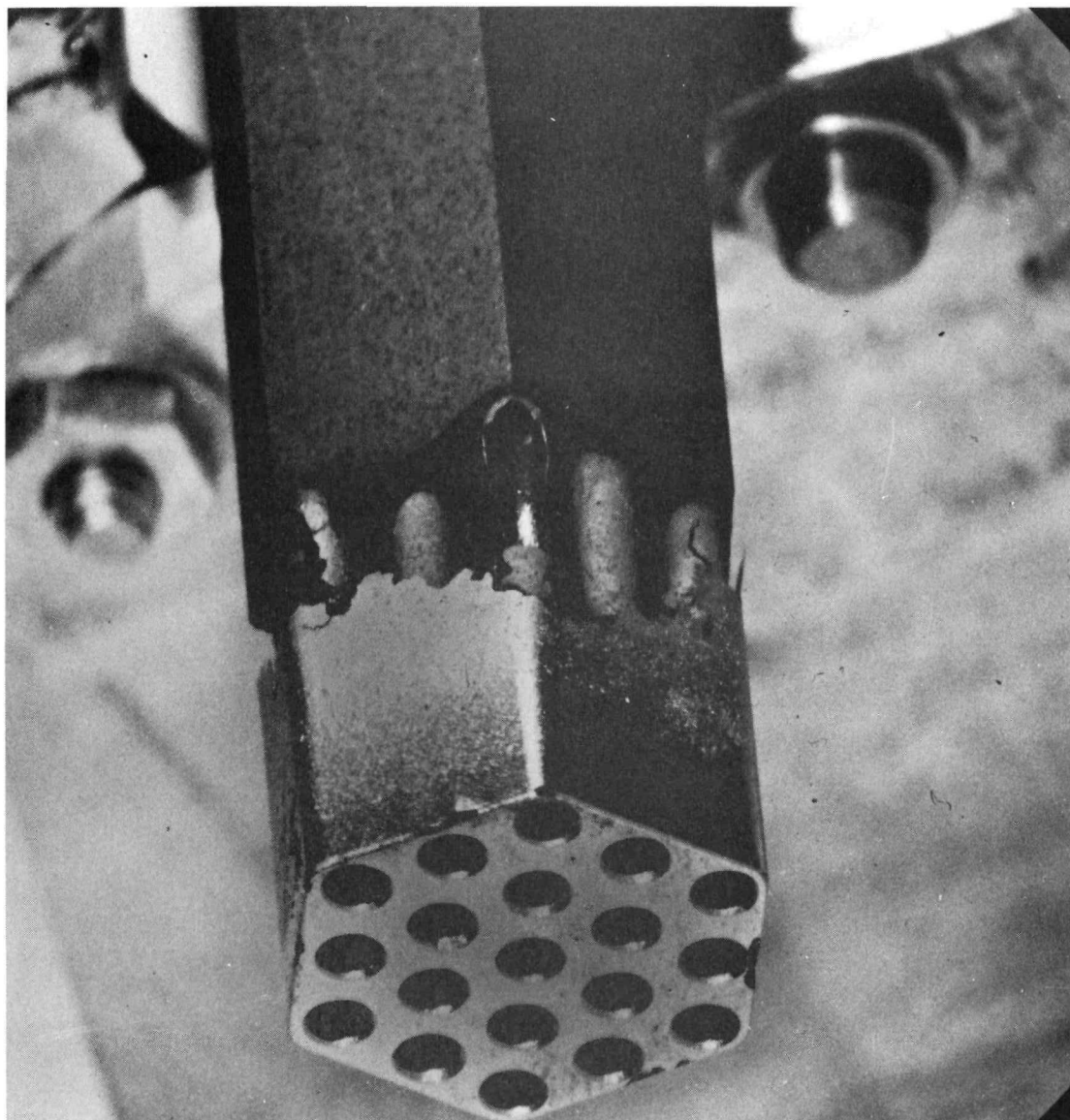
NRX-A2, A3 AND A4 RADIATION SHIELDING GEOMETRY
HYDROGEN CORROSION FURNACE

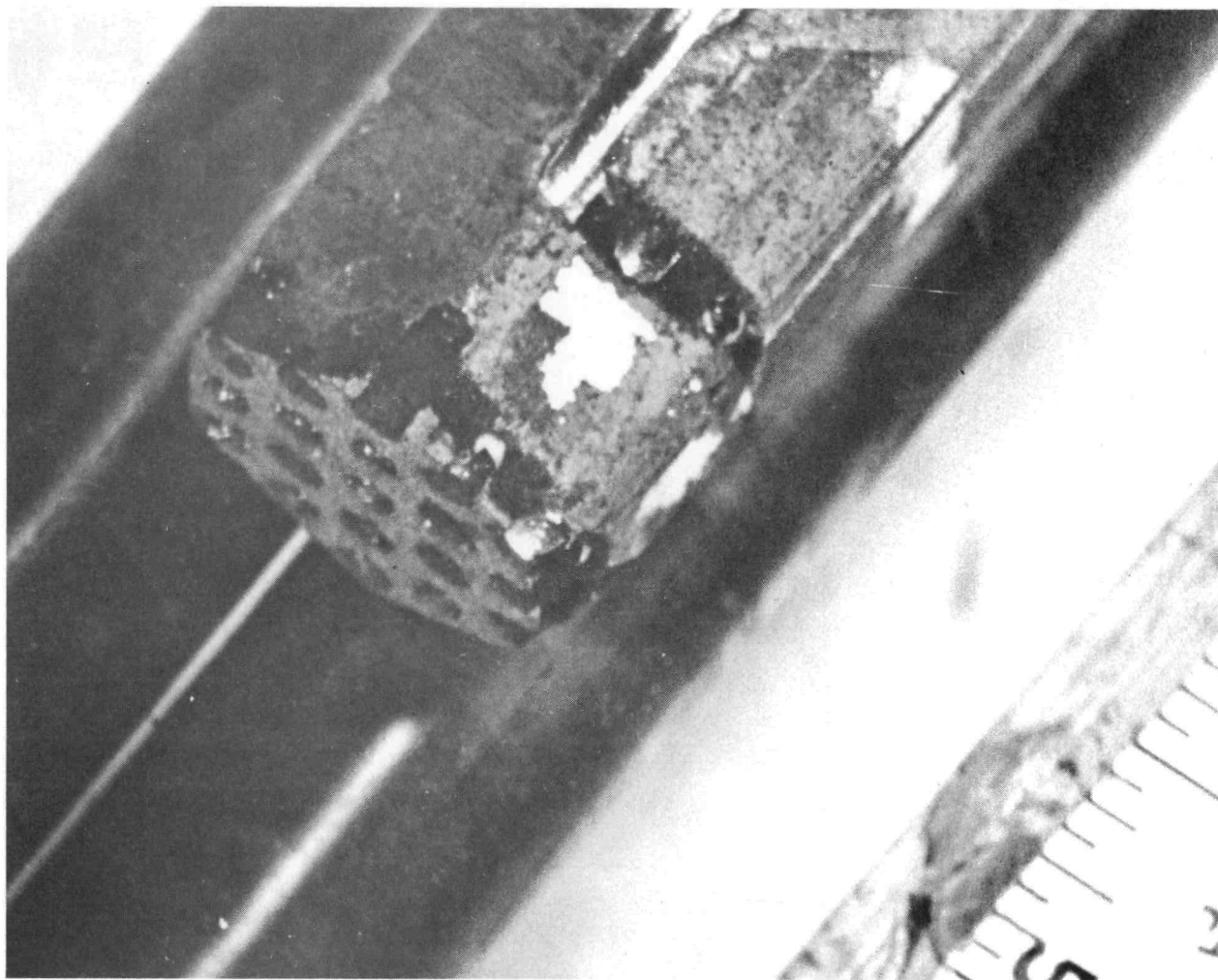


PROPOSED NRX-A5 RADIATION SHIELDING SCHEMATIC
HYDROGEN CORROSION FURNACE



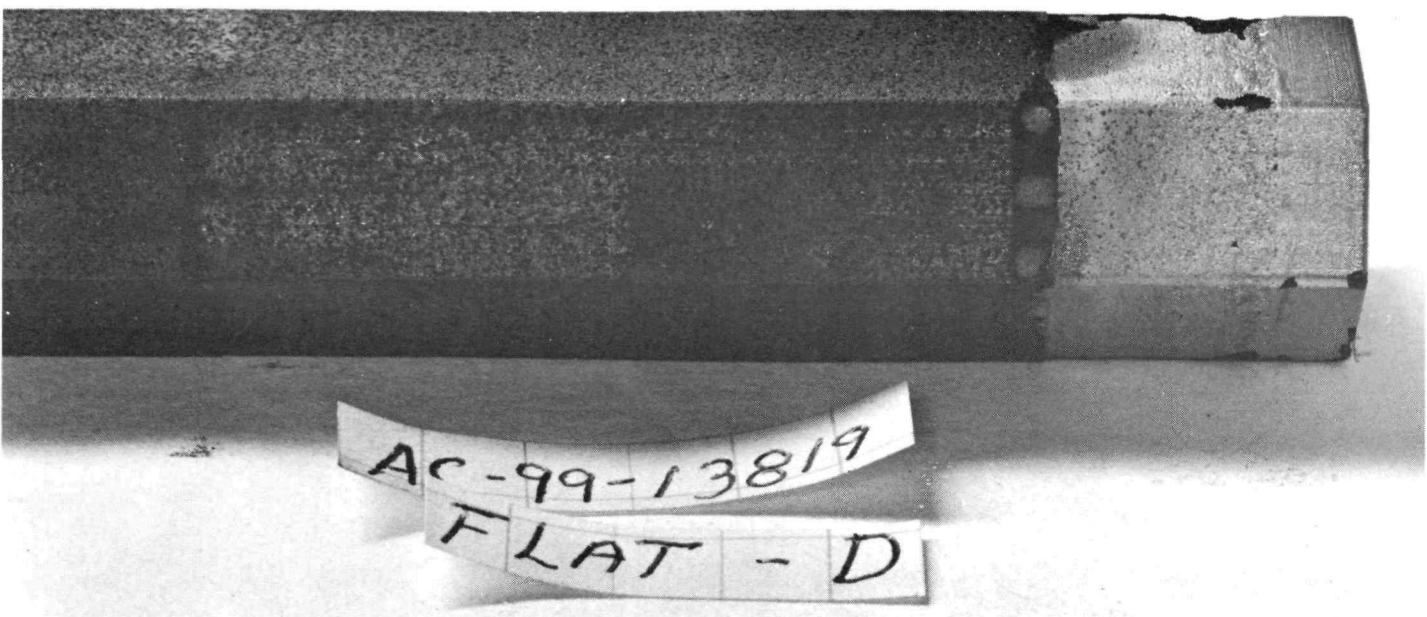






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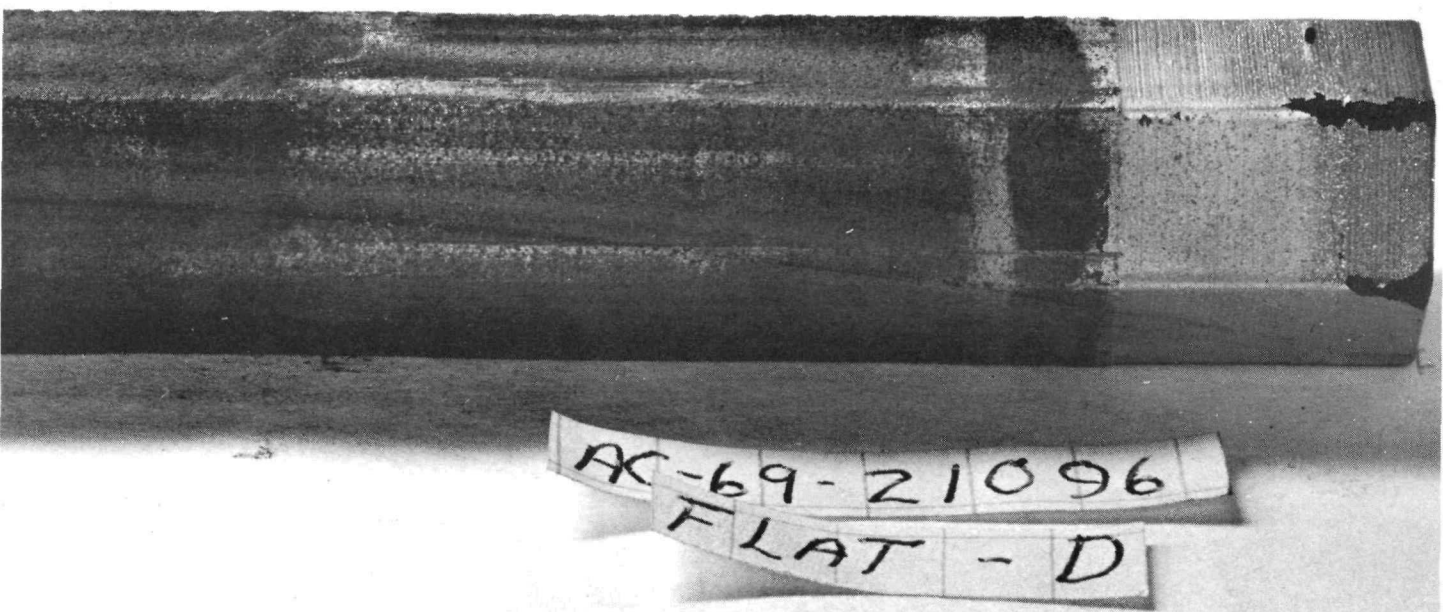
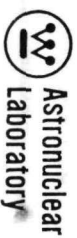
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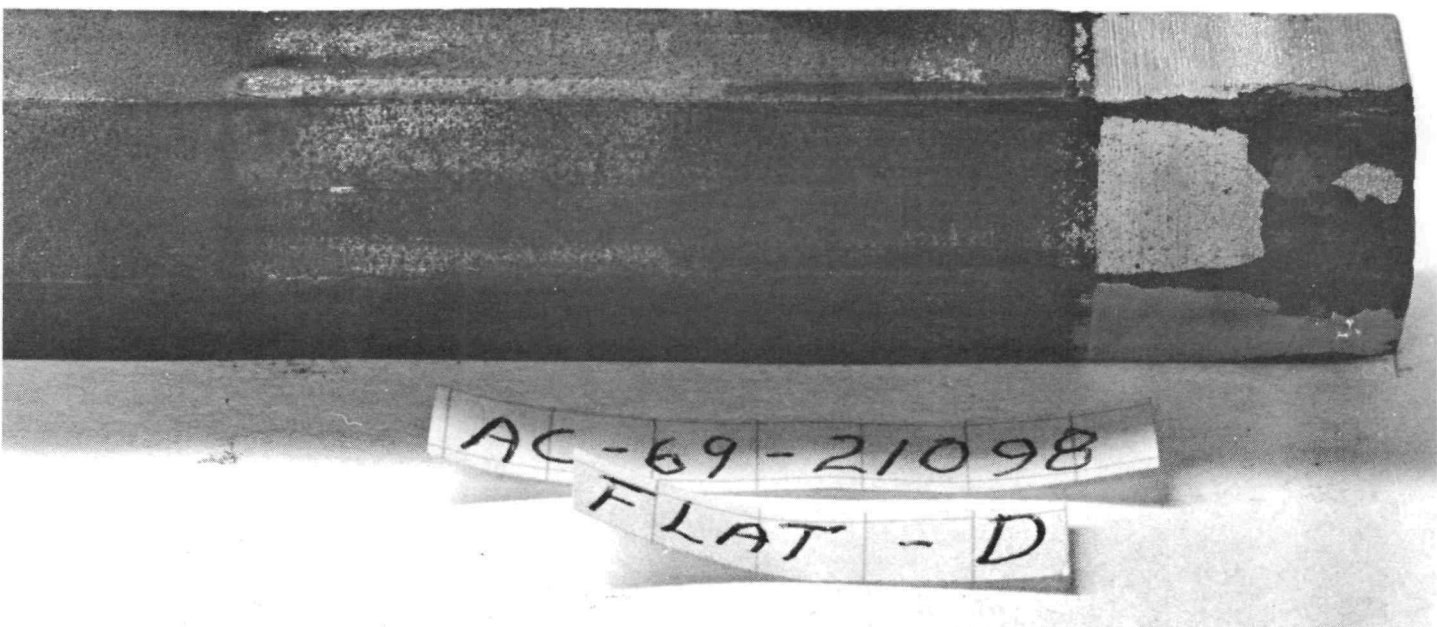
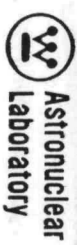
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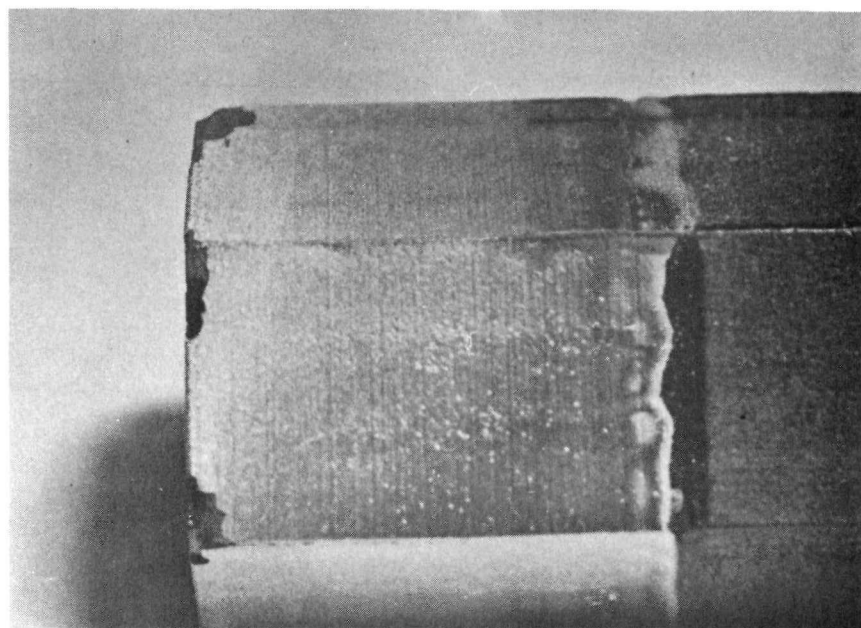
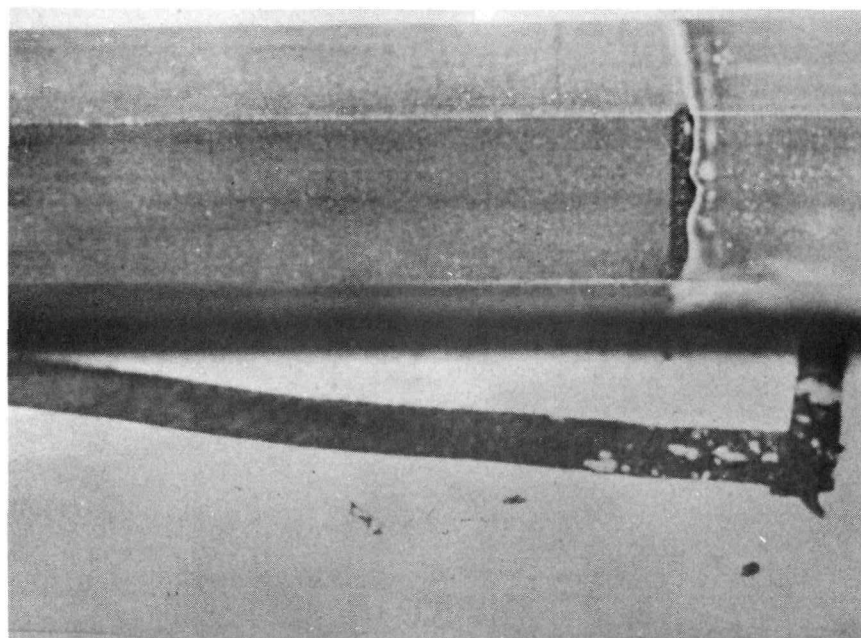
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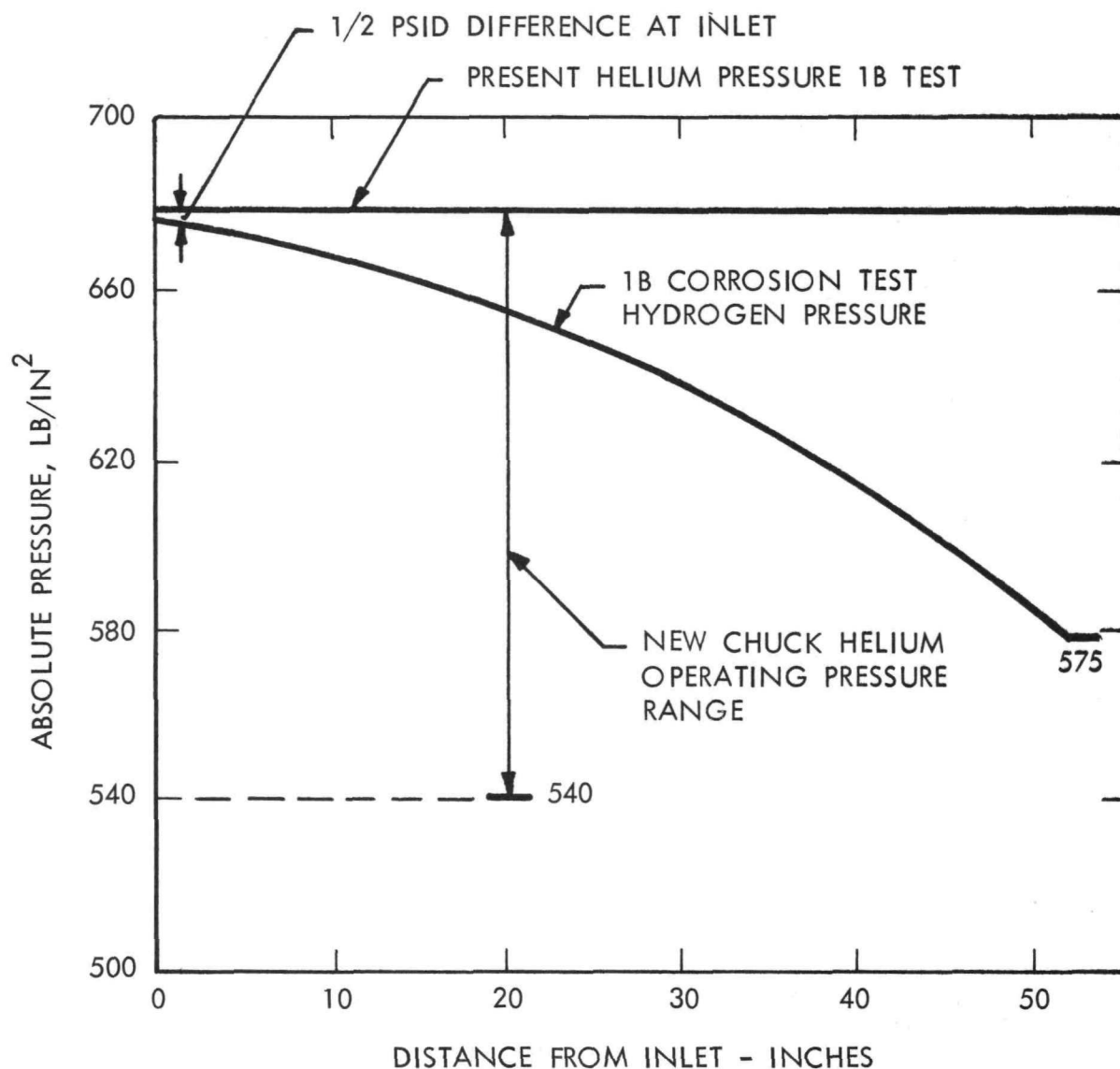


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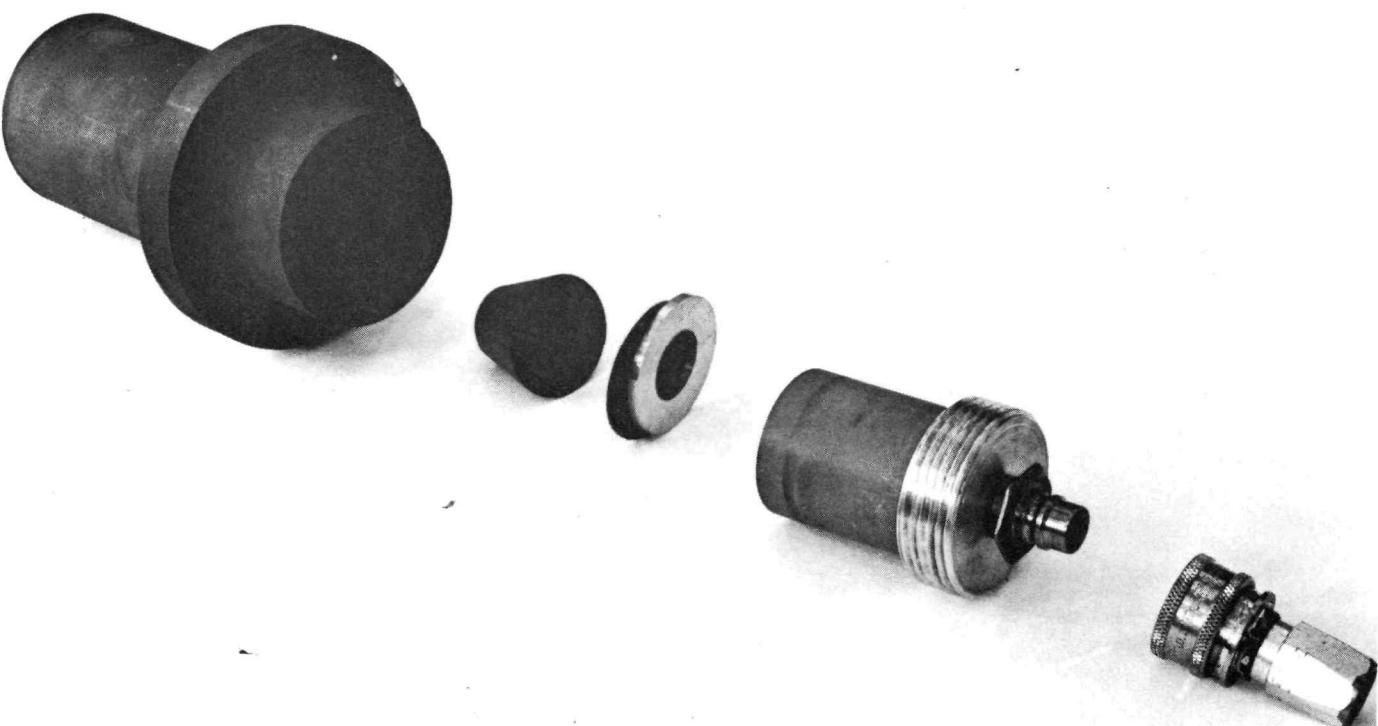
**TYPICAL CORROSION BEHAVIOR
OF FUEL ELEMENT IN CLUSTER TESTS**



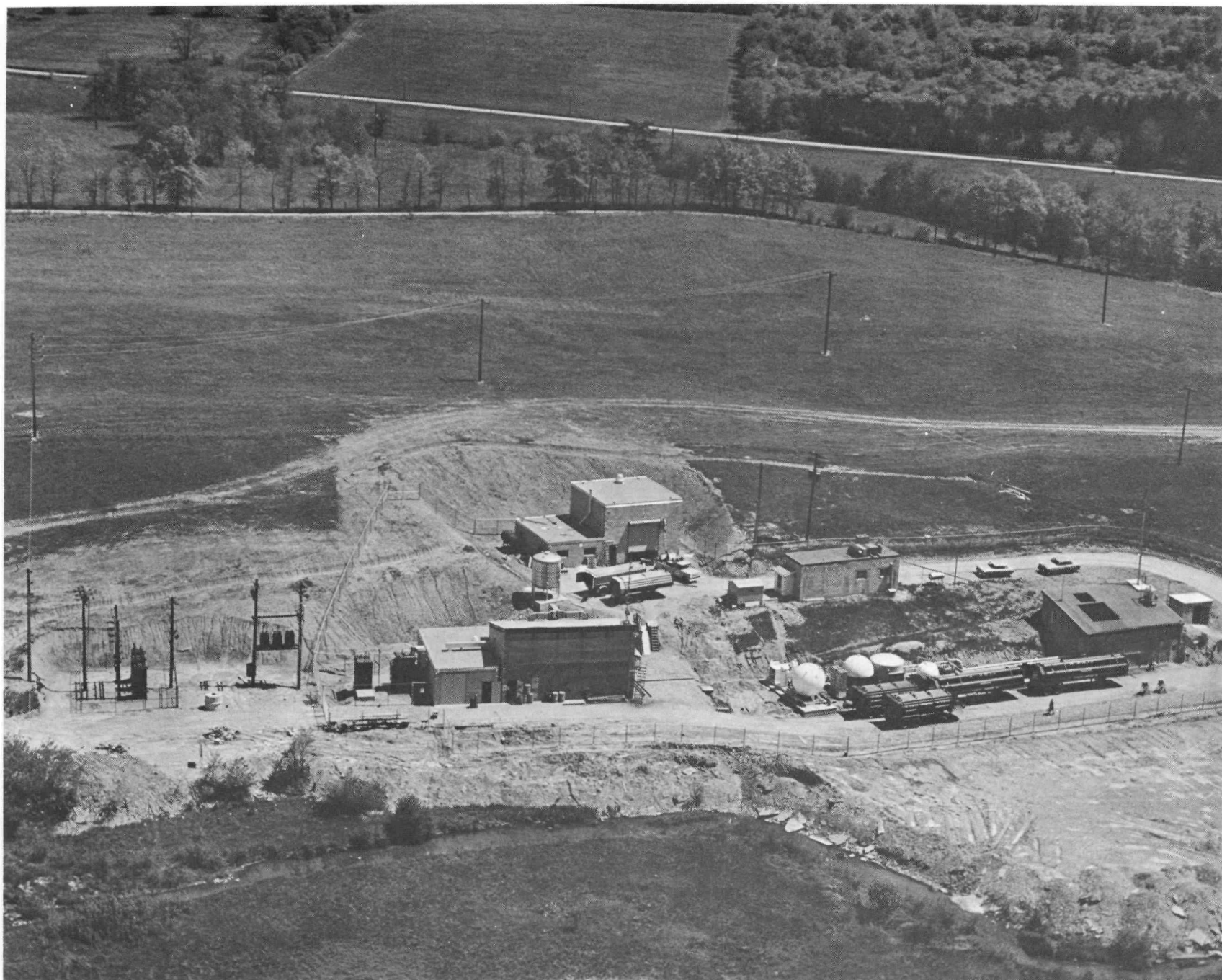
PRESSURE DISTRIBUTION - 1B HYDROGEN CORROSION TEST

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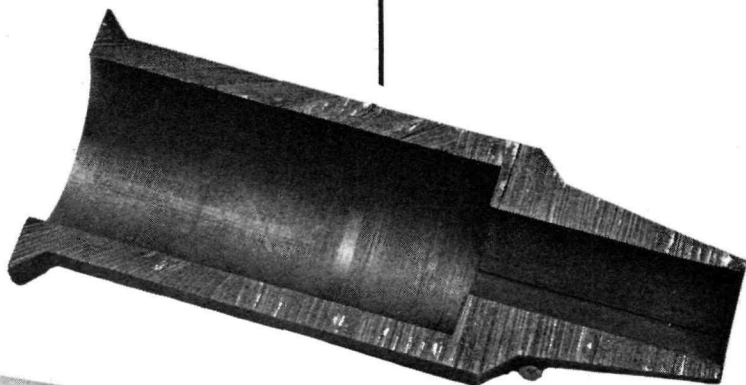
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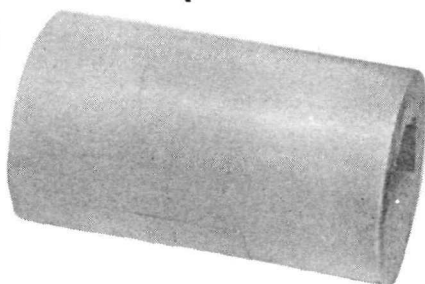


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Laboratory

FUEL ELEMENT HOT END CHUCK



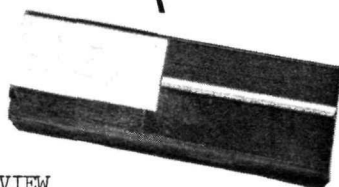
LOBE HOLDER



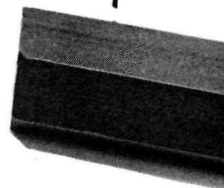
SPACER



LOBE



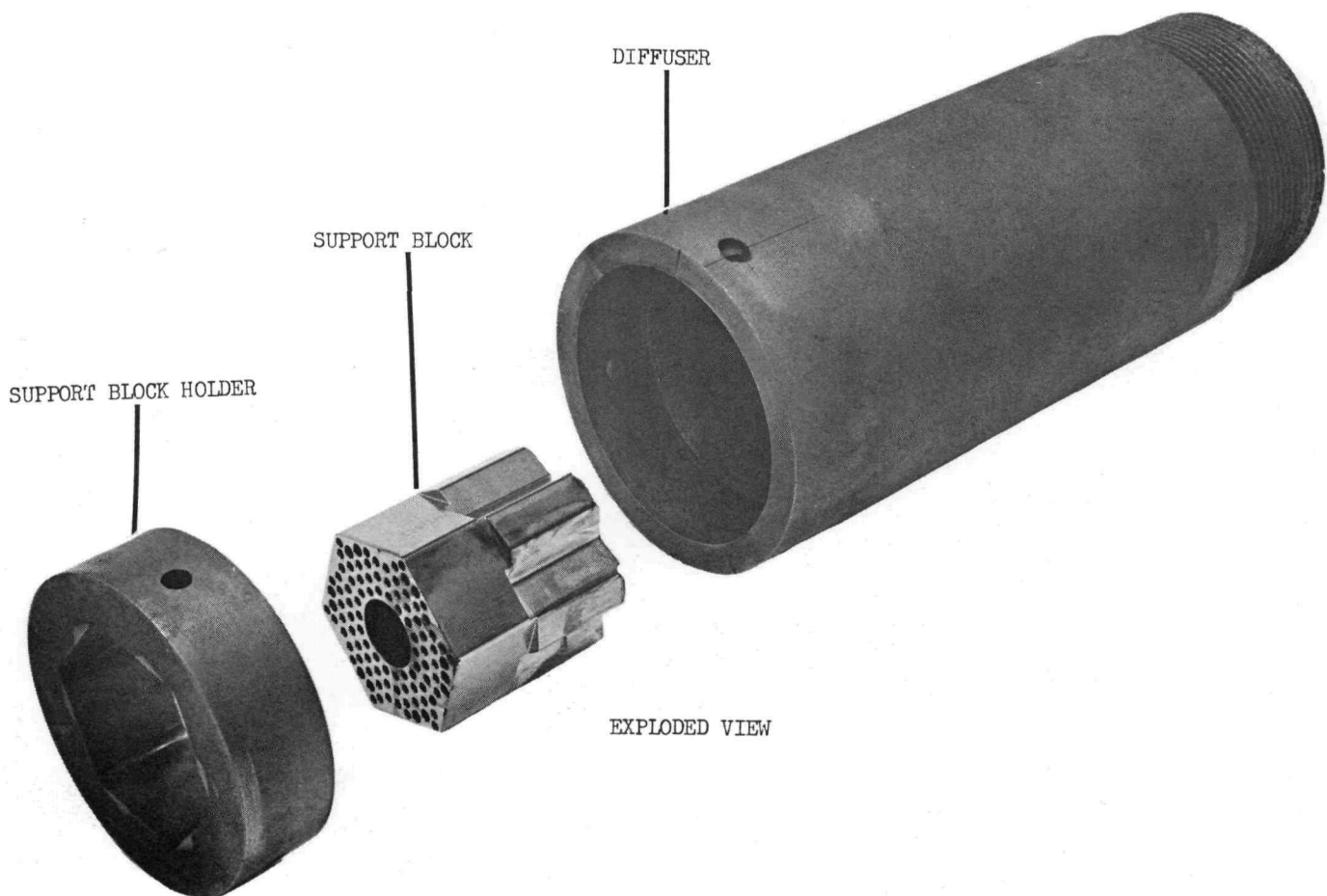
FUEL ELEMENT



EXPLODED VIEW

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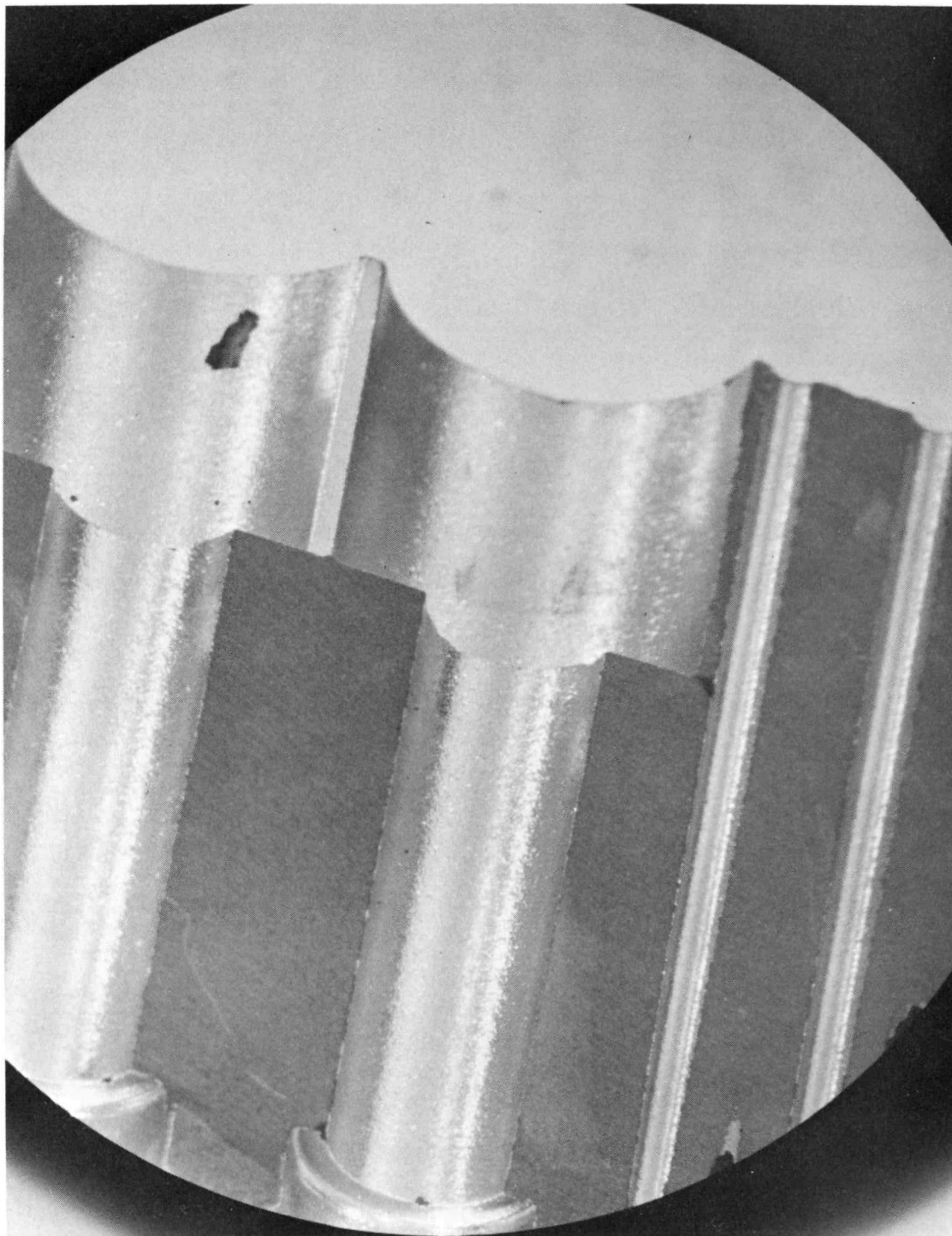
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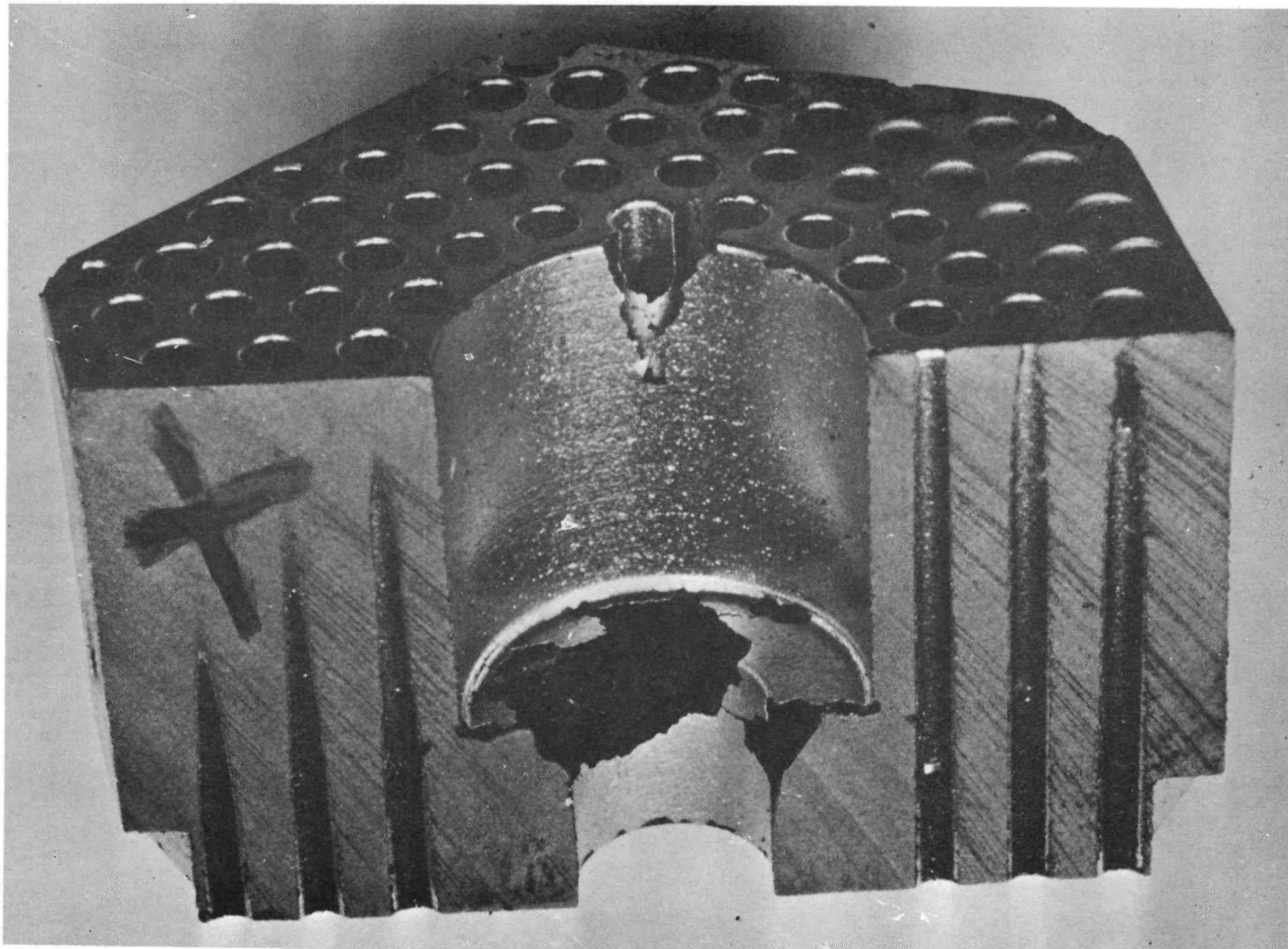
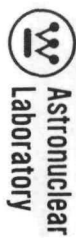


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Atomic Energy Act - 1954



SUPPORT BLOCK SERIAL NO. A-21952
CRACK LOADED AND CORROSION TESTED
MAGNIFIED 1.5X

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Atomic Energy Act - 1954

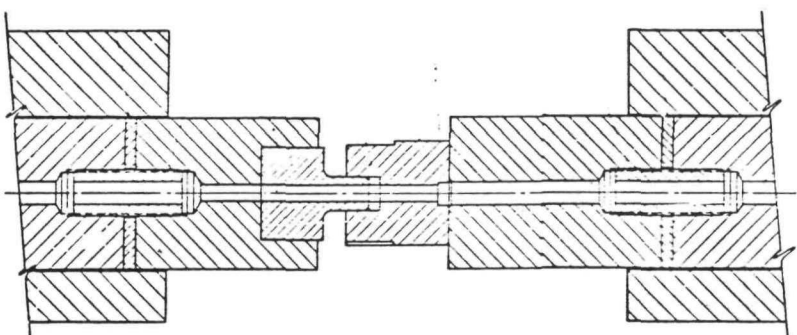
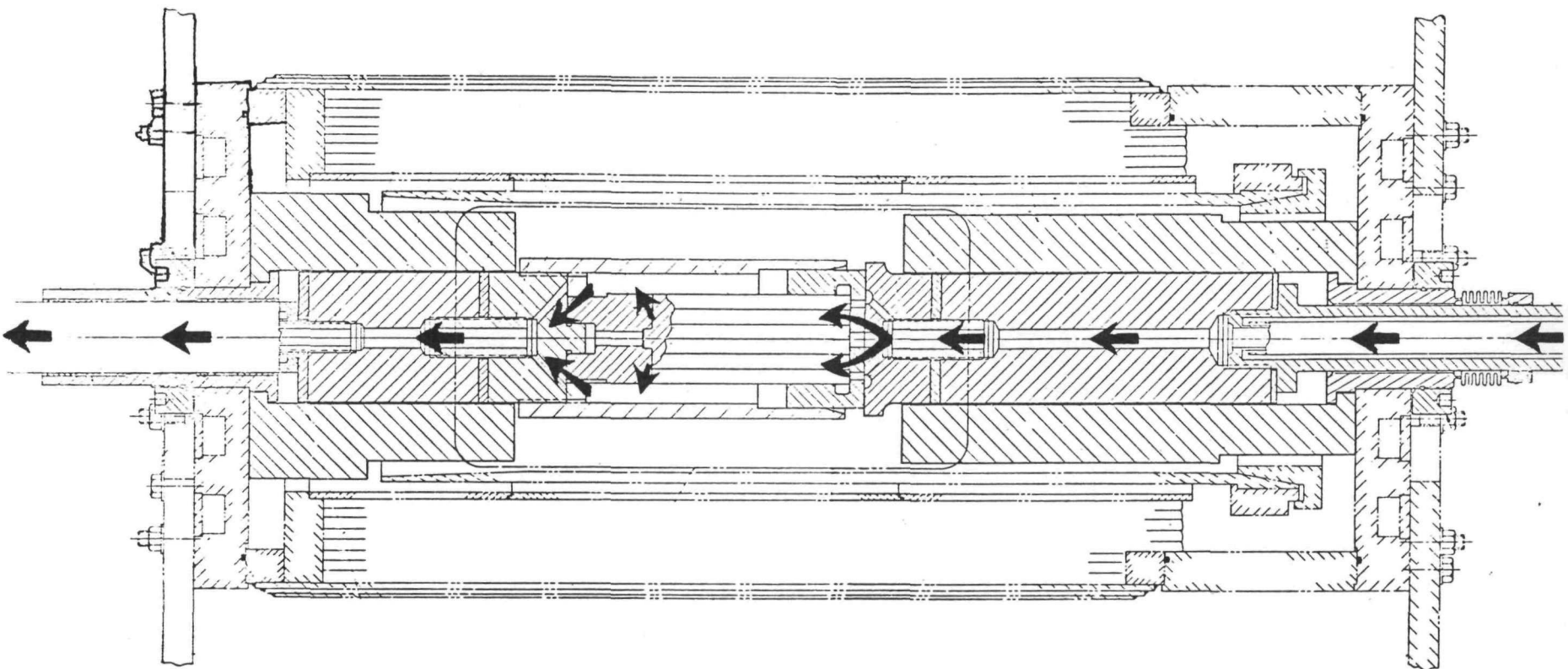


THE 10 INCH FURNACE IN OPERATION



FURNACE OPENED FOR SPECIMEN REMOVAL

603807C



FUELED ELEMENTS TESTED 40 MINUTES IN 5 SCFM HYDROGEN

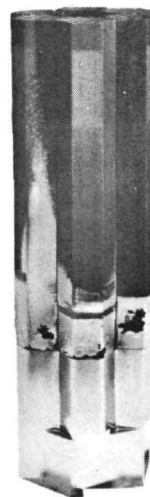
NRX-A3
ELEMENTS



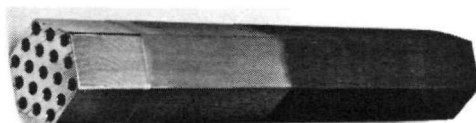
BEFORE TEST



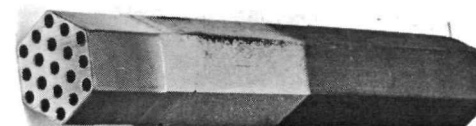
AFTER TEST



NRX-A4
BACKCOATED
ELEMENTS

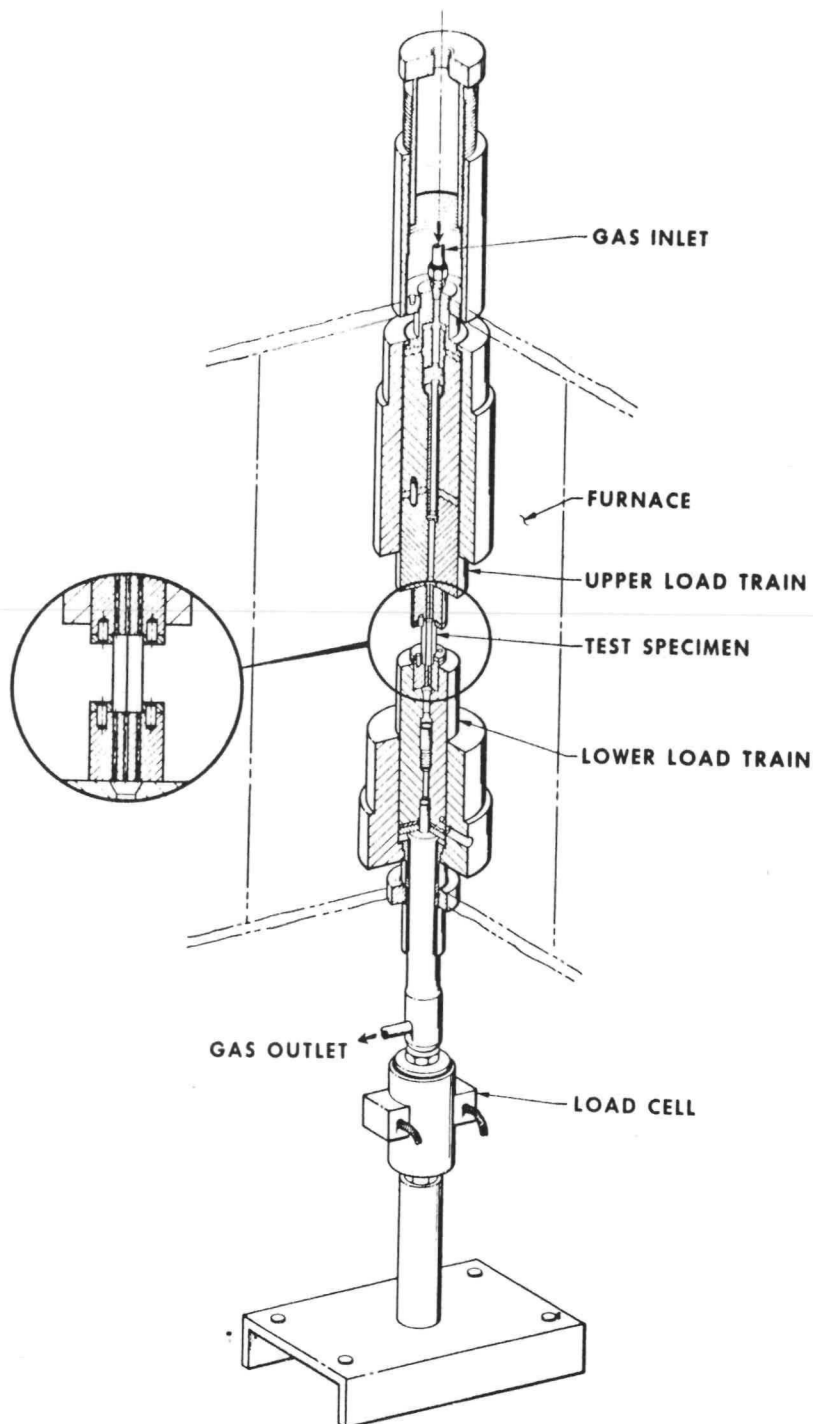


BEFORE TEST

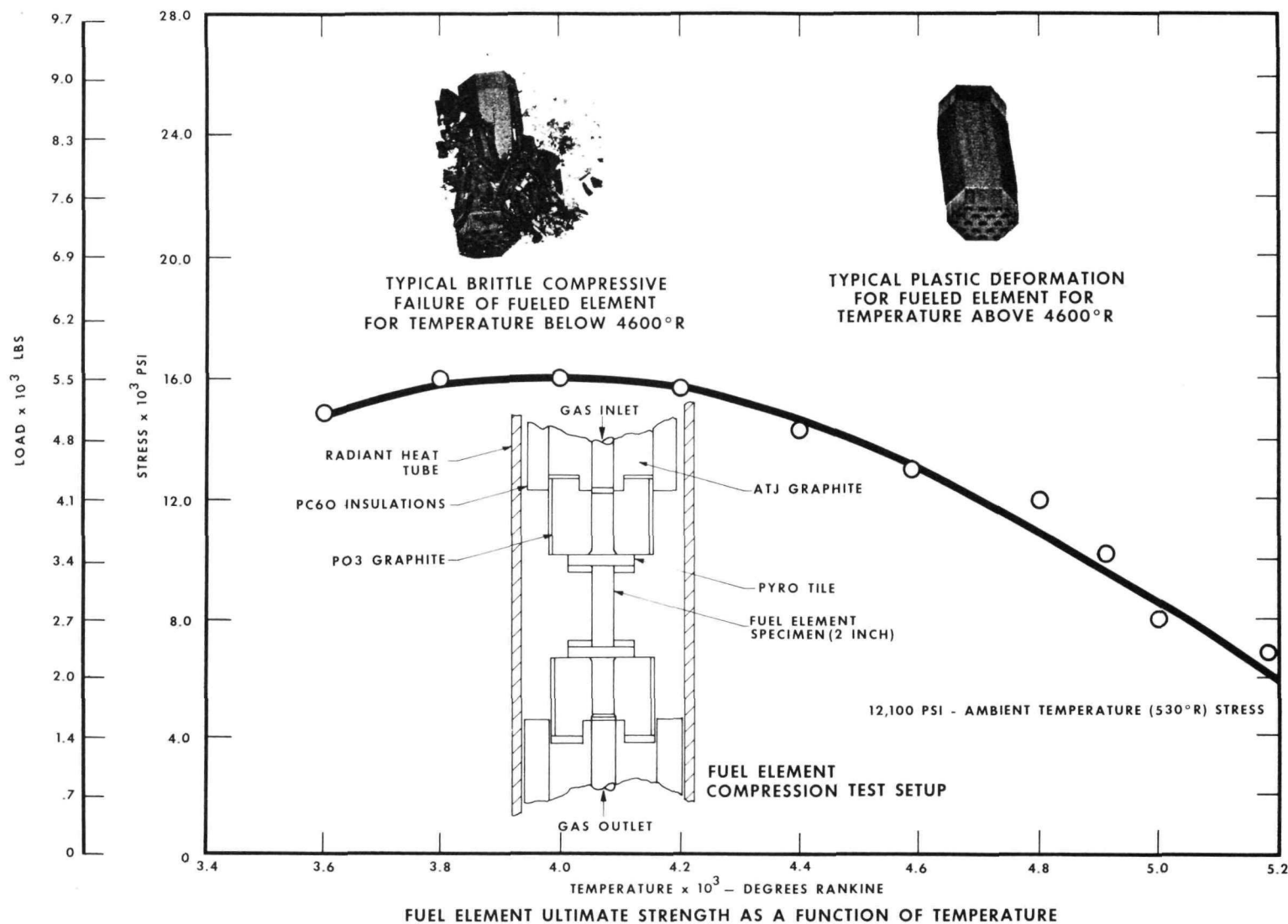


AFTER TEST

603473D



**FUEL ELEMENT HIGH TEMPERATURE
COMPRESSION TEST LOADING SYSTEM**



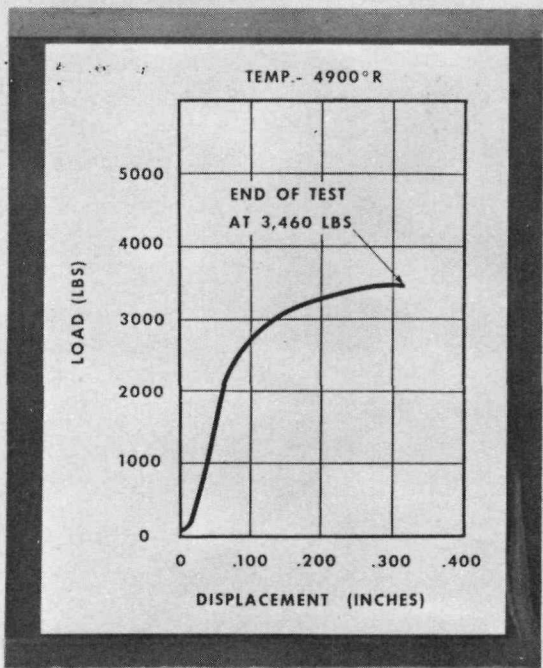
FUEL ELEMENT ULTIMATE STRENGTH AS A FUNCTION OF TEMPERATURE

FUEL ELEMENT TEST RESULTS

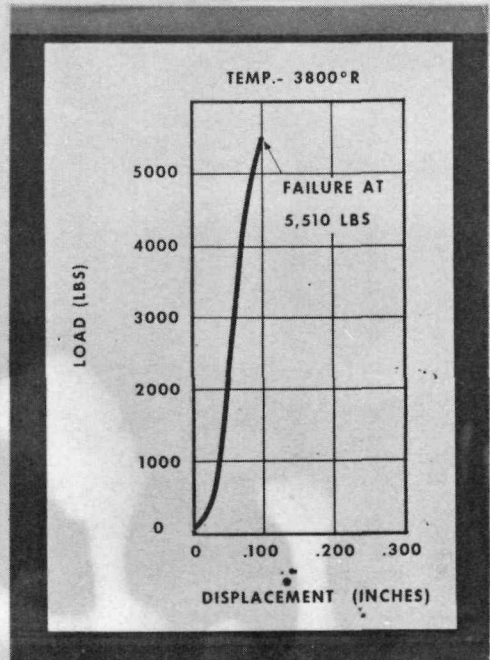
603492D

ENGINEERING MECHANICS LAB.

ENRICHED



TYPICAL PLASTIC DEFORMATION FAILURE

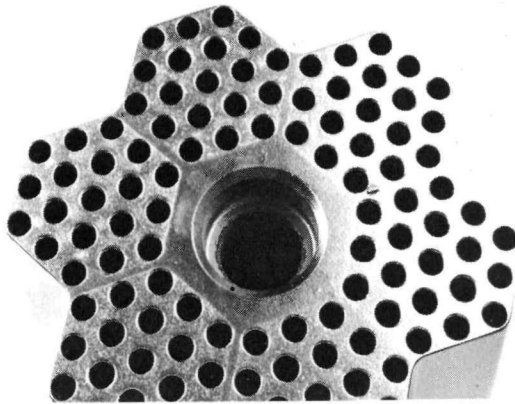


TYPICAL BRITTLE FAILURE

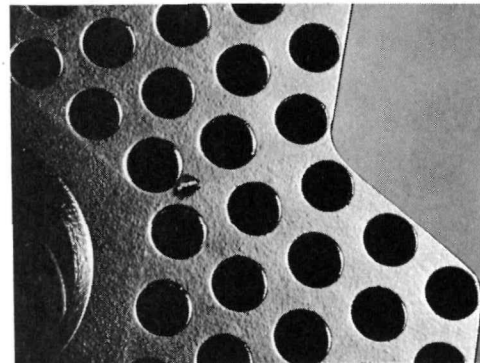
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FUEL ELEMENT COMPRESSION TEST RESULTS

NbC WELDING TESTS



END VIEW OF SUPPORT BLOCK
AFTER SEPARATION. THE FROSTED
APPEARANCE INDICATES THE POINTS
OF WELDING



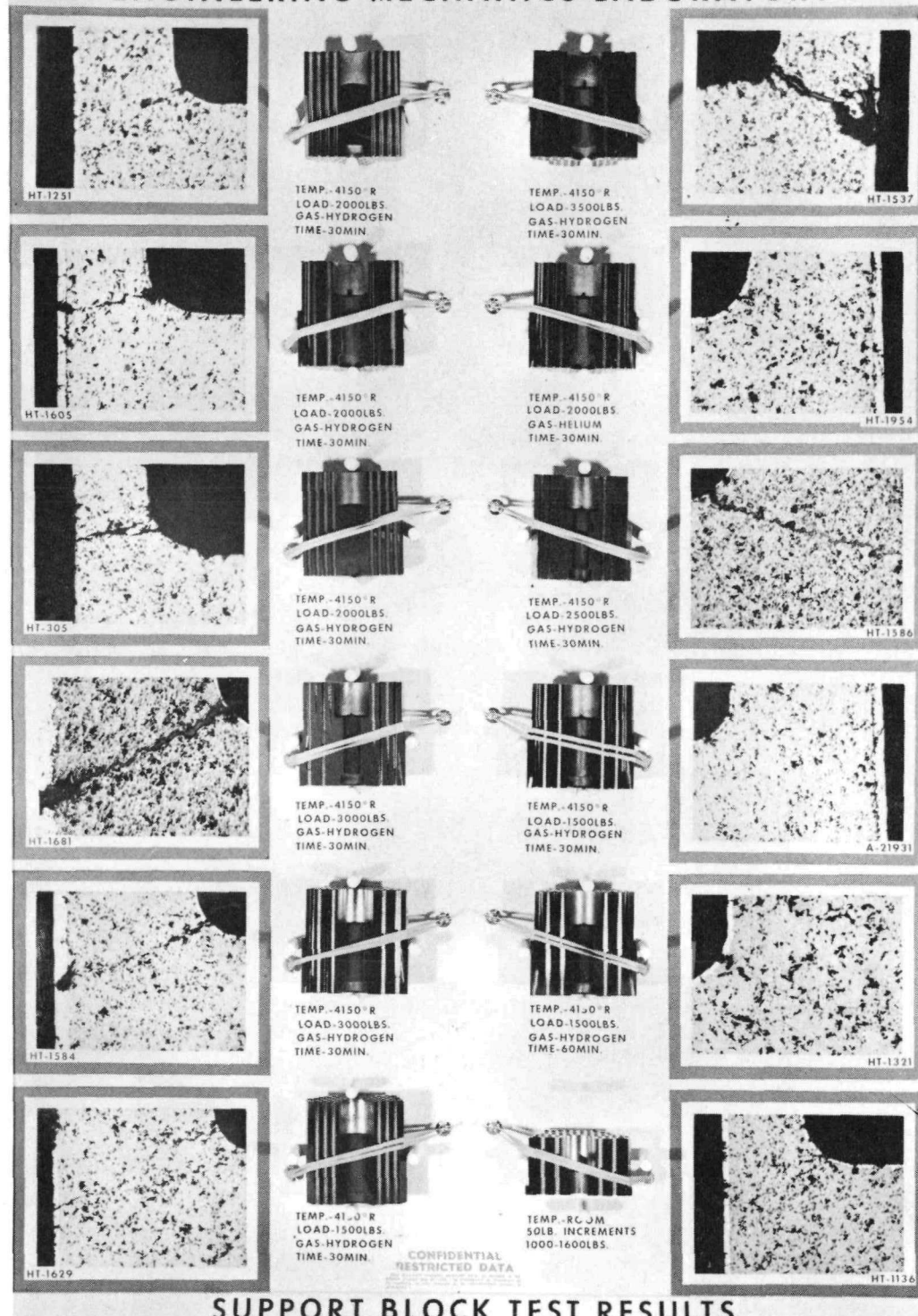
ENLARGED VIEW OF CHIP
PULLED FROM ELEMENT
ON SEPARATION.



AN EXAMPLE OF COMPLETE
WELDING OF NbC COATINGS

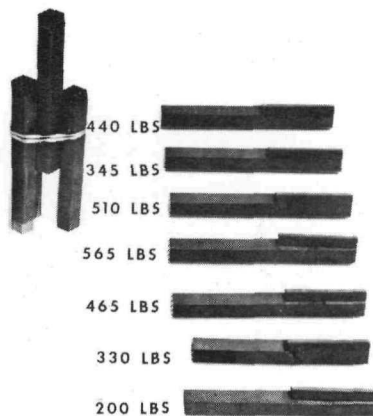
603478D

ENGINEERING MECHANICS LABORATORY

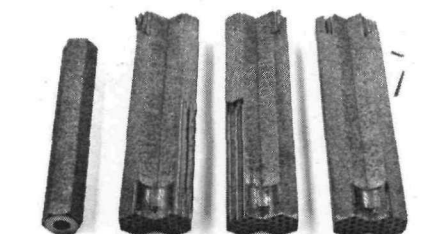
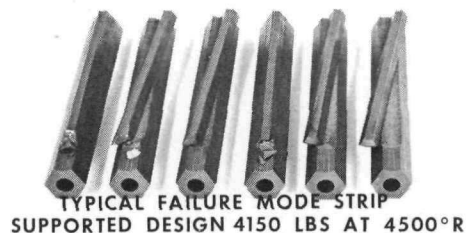


SUPPORT BLOCK TEST RESULTS

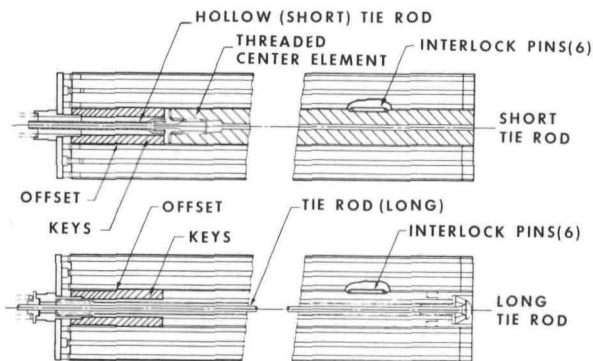
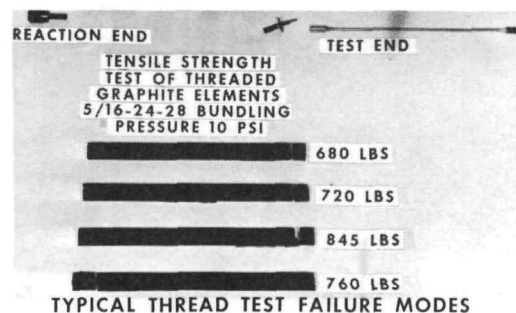




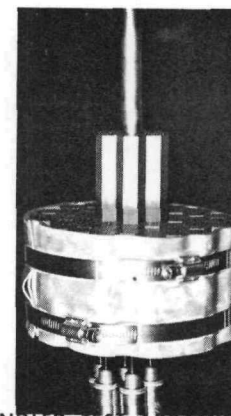
TYPICAL KEY FAILURE MODES



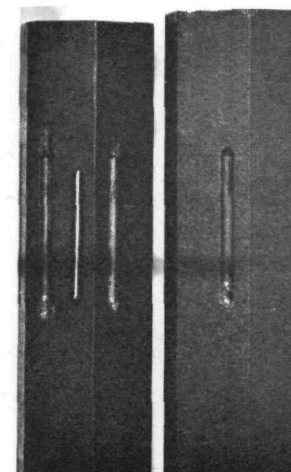
CLUSTER DEVELOPMENT TEST RESULTS



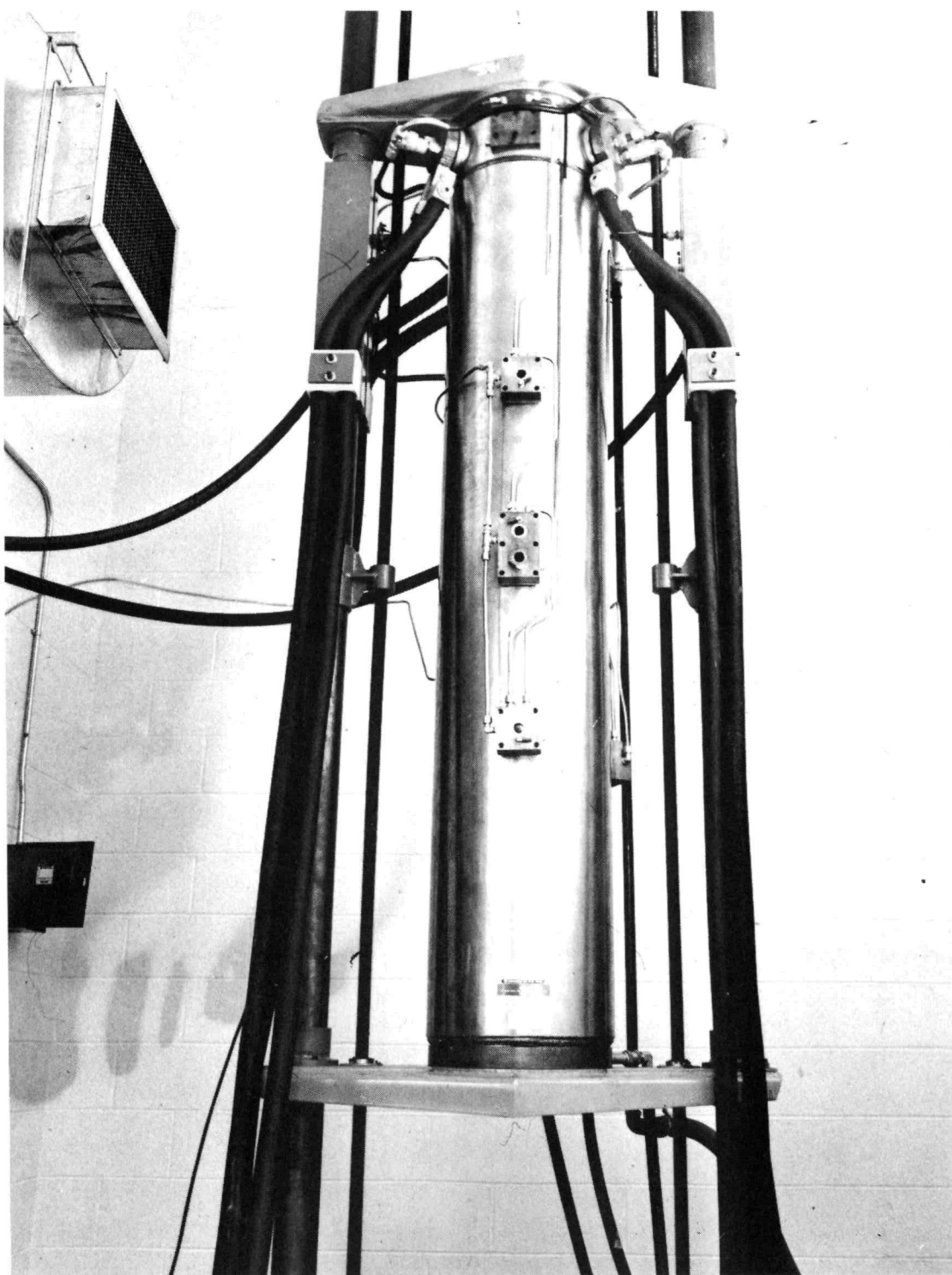
DRAWING OF COLD-END SUPPORTED DESIGN



BUNDLED TEST ARRANGEMENT



TYPICAL FAILURE MODE
FOR HOT-END PIN DESIGN 603474D



CENTRAL ELEMENT AND TIE ROD INTEGRITY

NRX A2

O RING SEAL → NRX A3

CENTRAL ELEMENT
INSTRUMENT
INSTALLATION REDESIGN → NRX A3

COATING OF
CENTRAL BORE → NRX A4

PYROFOIL SEAL → NRX A4

2 AND 4 HOLE
CONTROL ELEMENTS → NRX A4

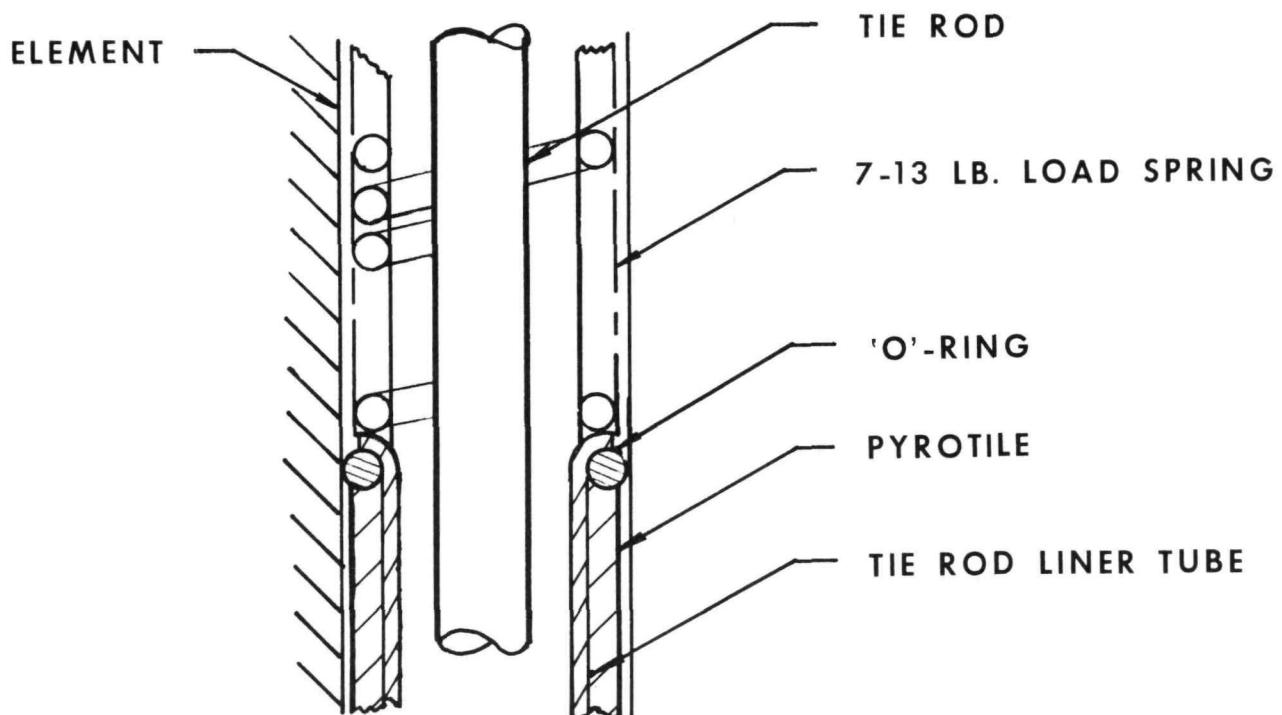
TANTALUM CARBIDE
IMPREGNATED ELEMENTS → NRX A4

HOLLOW TIE ROD
INSTRUMENT INSTALLATION → NRX A5

NRX A3

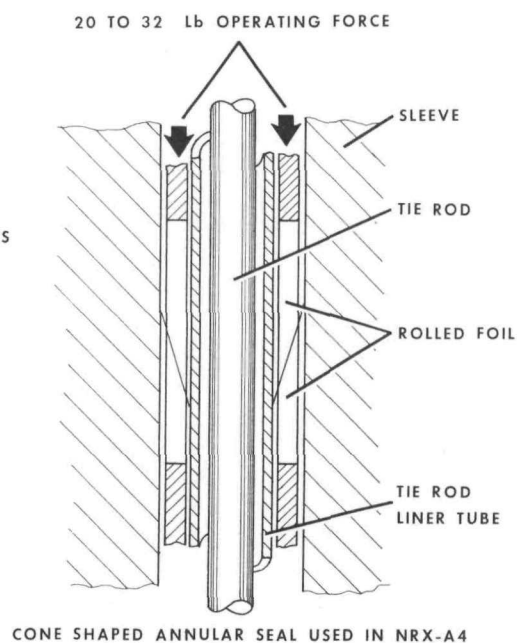
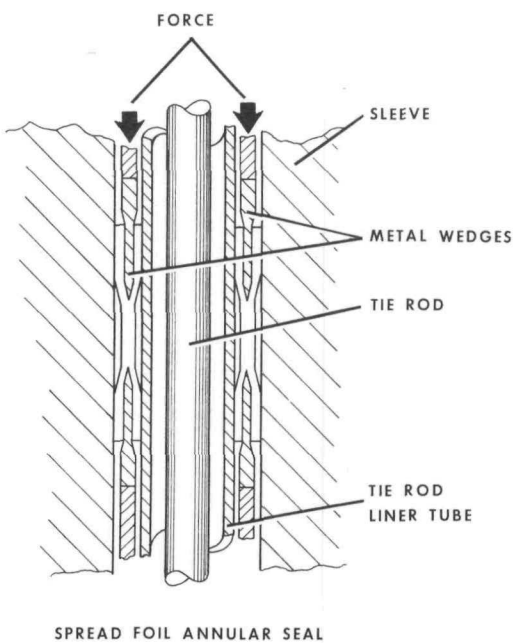
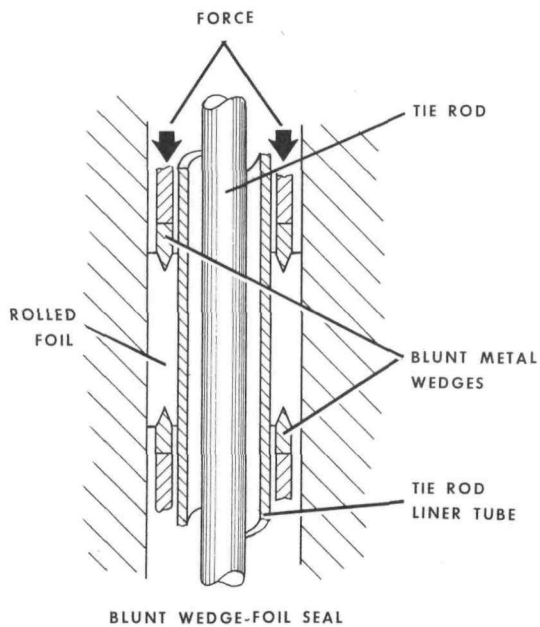
FLOW SCRAM
INVESTIGATION → NRX A3

COUNTER FLOW
TIE RODS → NRX A6



'O'-RING SEAL USED IN NRX-A3

603125B

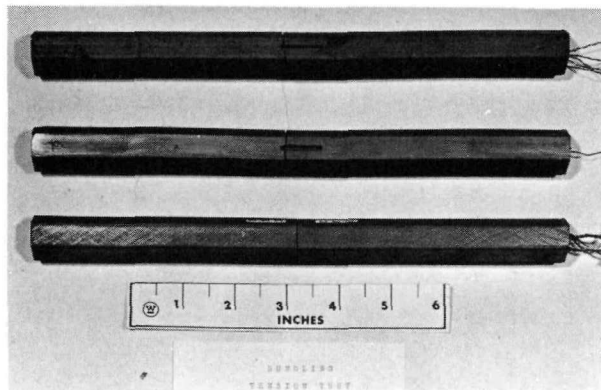


EVOLUTION OF A CENTRAL ELEMENT LINER TUBE SEALING DEVICE

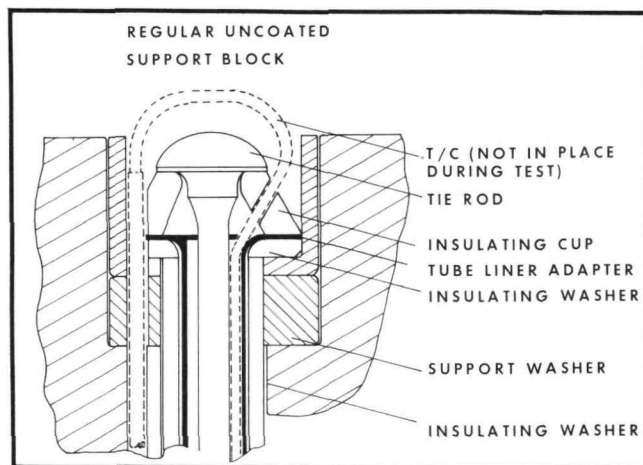
603127B

THERMOCOUPLE INSTRUMENTED COMPONENTS EFFECT ON STRENGTH

FUEL ELEMENTS WITH T/C SLOTS



	PLAIN	SLOTTED	REDUCTION %
TENSILE, PSI	4600	3840	17
FLEXURE, PSI	4420	3330	25



	STANDARD	MODIFIED
ULTIMATE FRACTURE LBS	4330	4340

HOLLOW INSTRUMENT TIE RODS

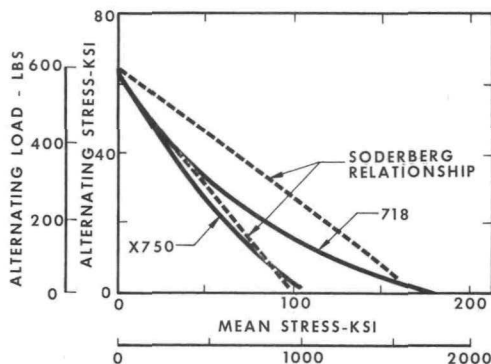
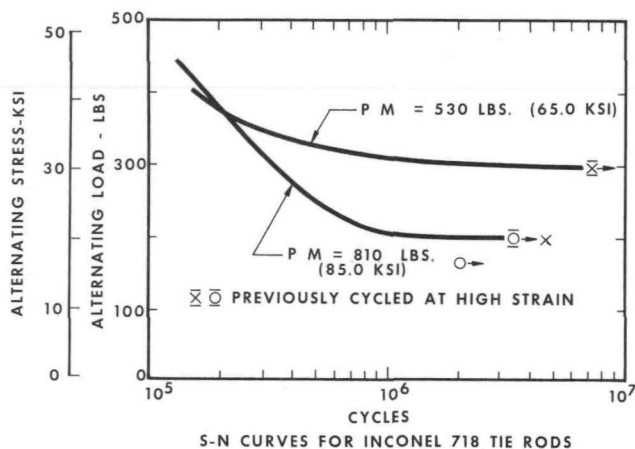
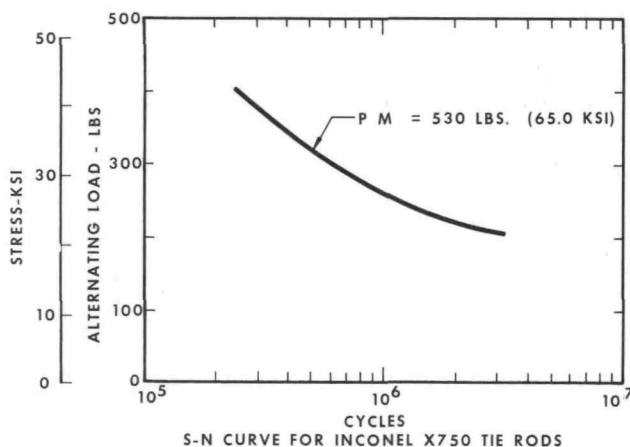
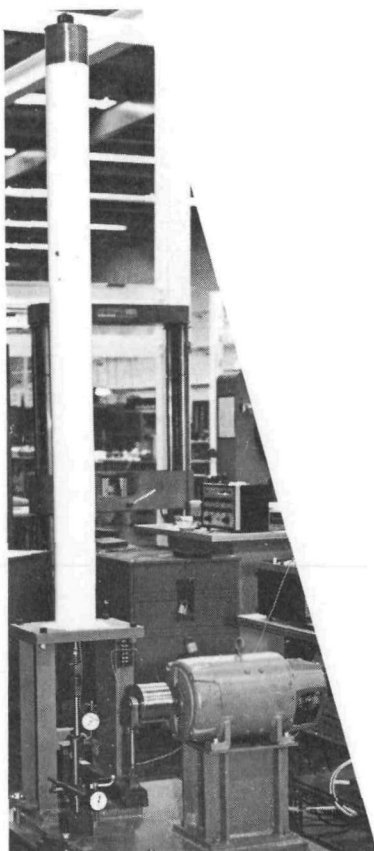


	INCONEL 718	
	SOLID TIE ROD	HOLLOW
TENSILE STRENGTH PSI	214,000	229,000
YIELD STRENGTH PSI	173,000	200,000
ELONGATION-%	10.2	5.6
CYCLE LIFE AT 800 ± 200 LBS.	3 × 10 ⁶	0.65 × 10 ⁶

THESE RODS DISQUALIFIED FOR NRX-A4

603444 D

TIE ROD FATIGUE TESTS




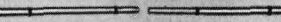





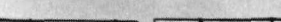
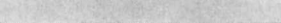



CONSTANT LIFE DIAGRAM FOR NRX-A TIE RODS

SUMMARY OF TEST DATA

TIE ROD ULTIMATE STRENGTHS-		X750	169,000psi (1600 lbs)
		718	214,000psi (2200 lbs)
TIE ROD DESIGN LOADS (STEADY)		530 TO 810 lbs	
FATIGUE STRENGTH (2×10^6 CYCLES)	Pm=530lbs	x750	$\pm 23,000$ psi (± 220 lbs)
		718	$\pm 31,600$ psi (± 300 lbs)
	Pm=810lbs	x750	9,000psi (± 85 lbs)
		718	21,000psi (± 200 lbs)

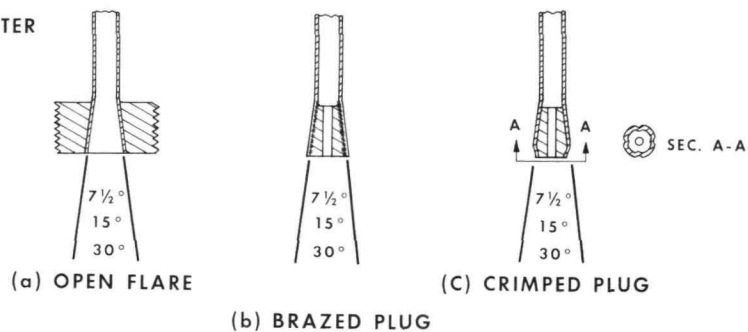
603438D

COUNTERFLOW TIE TUBE GEOMETRY DEVELOPMENT

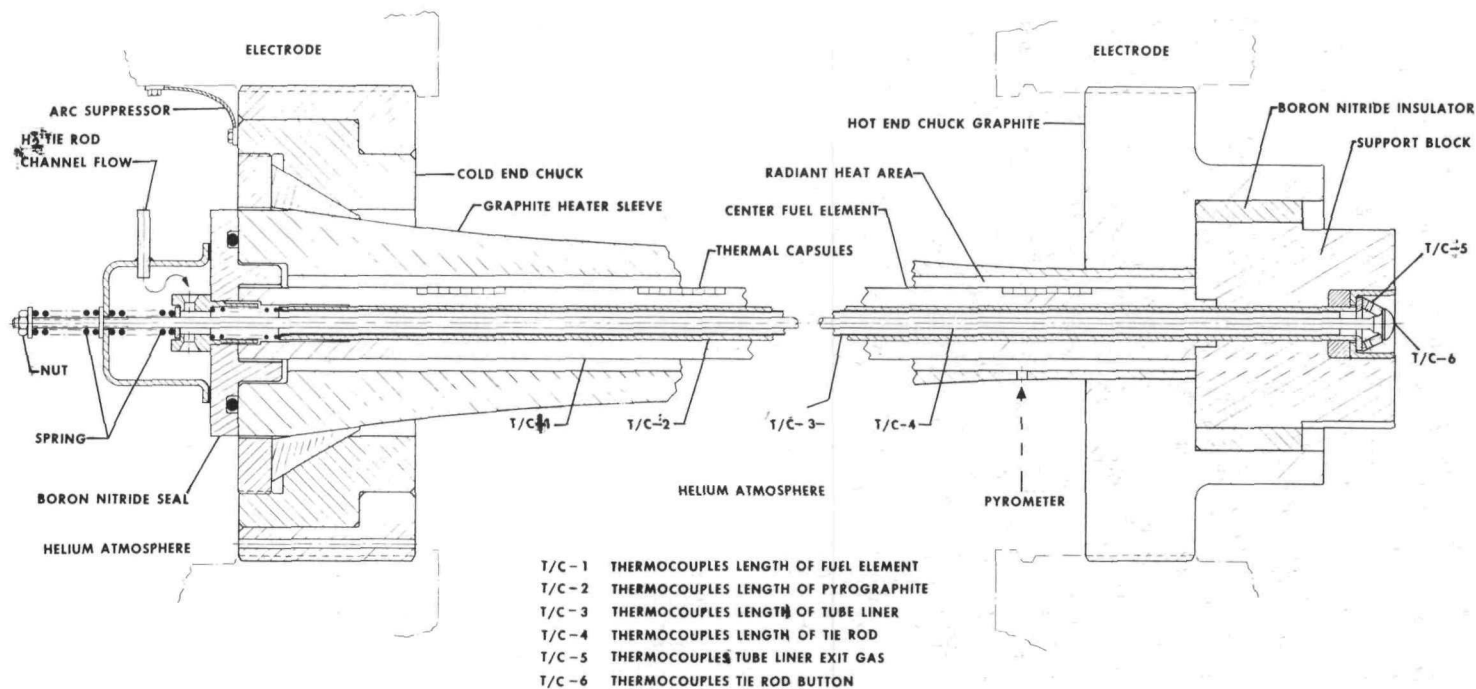
	7½°		390 LBS.
OPEN FLARE	15°		655 LBS.
	30°		606 LBS.
	7½°		685 LBS.
BRAZED PLUG	15°		669 LBS.
	30°		637 LBS.
	7½°		582 LBS.
CRIMPED PLUG	15°		644 LBS.
	30°		567 LBS.
	15°		487 LBS.
ANNEALED OPEN FLARE	15°		567 LBS.
	15°		532 LBS.

TIE TUBE STRUCTURAL EVALUATION OF GEOMETRY EFFECT

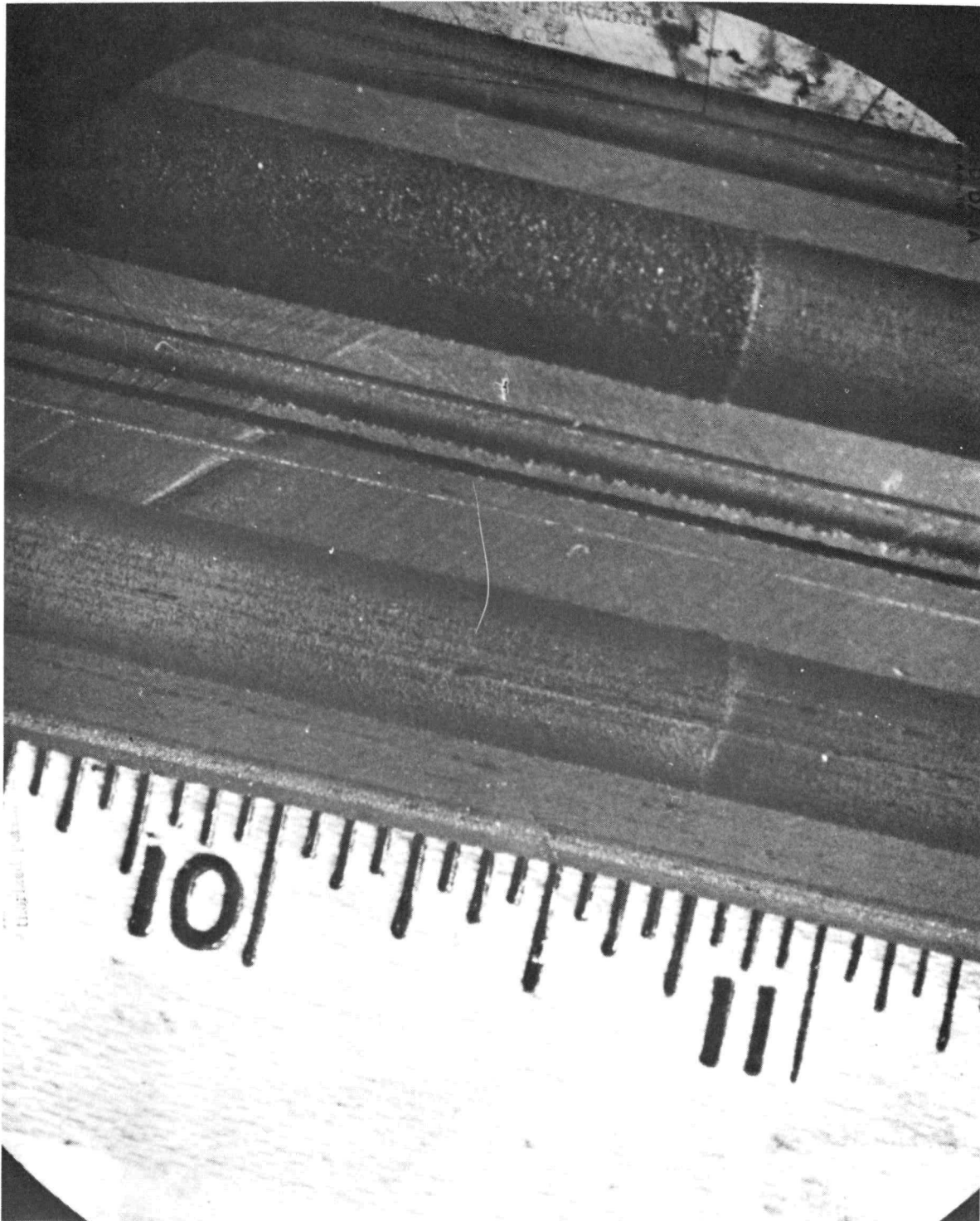
LOADING ADAPTER



603439 D



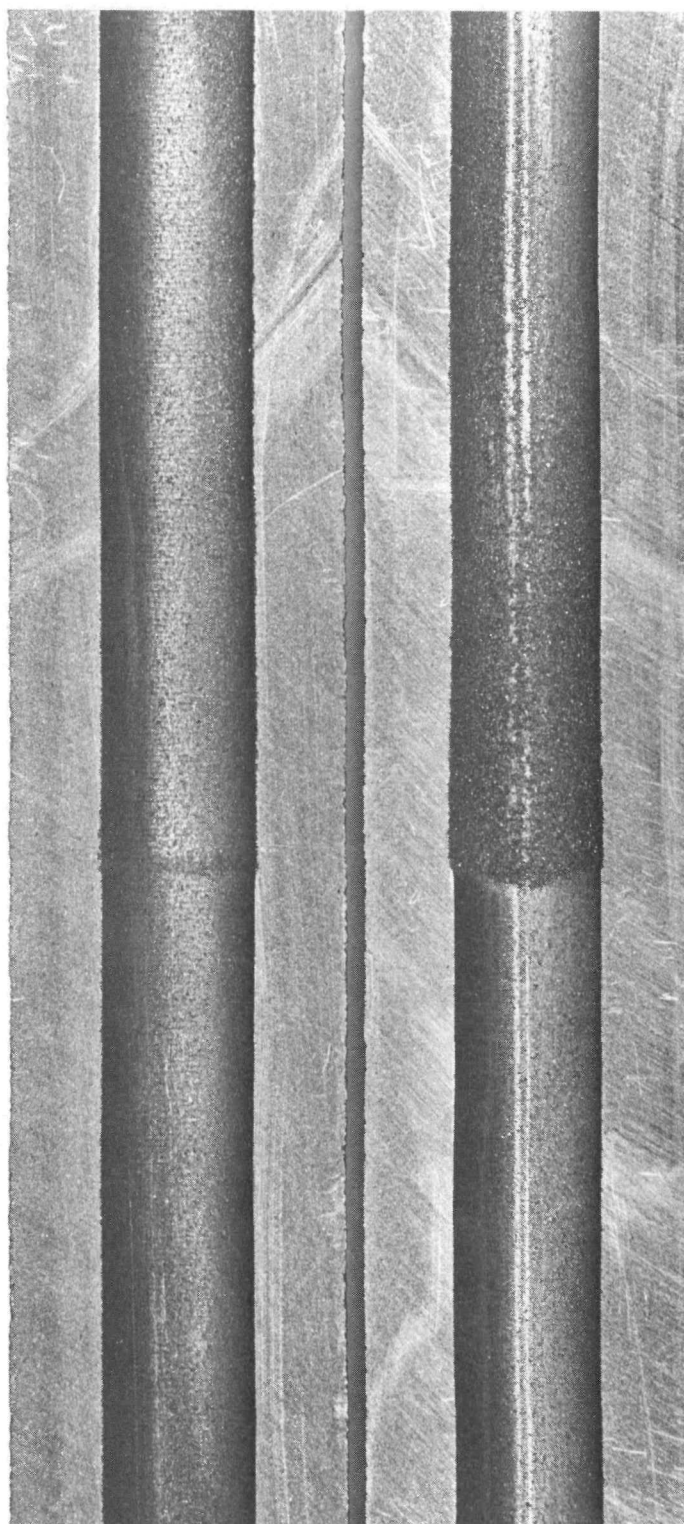
RADIANT HEATING CENTER FUEL ELEMENT



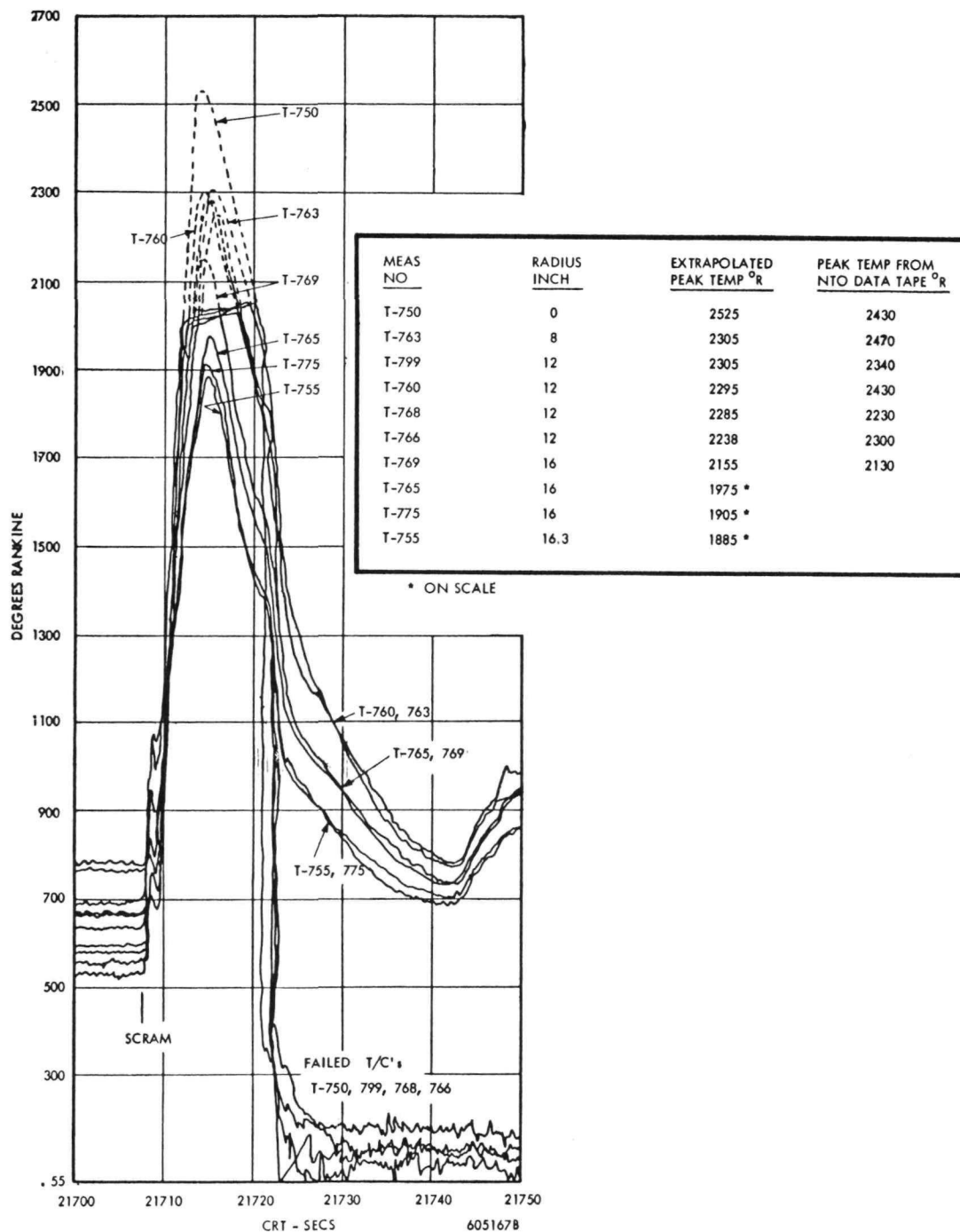
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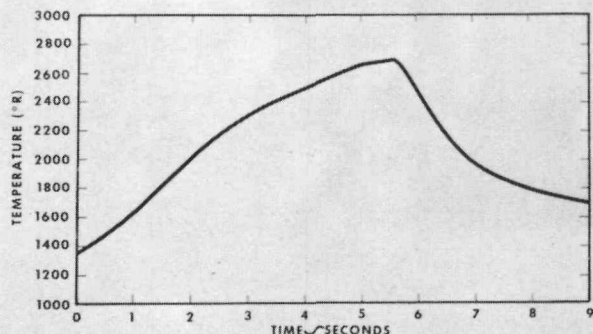


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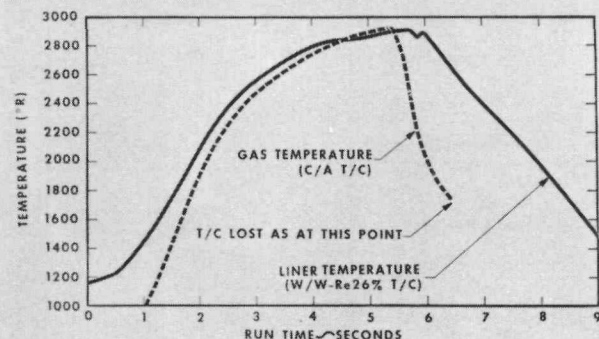
NRX-A3 EP IV TIE ROD EXIT GAS TEMPERATURES DURING SCRAM

ENGINEERING MECHANICS LABORATORY



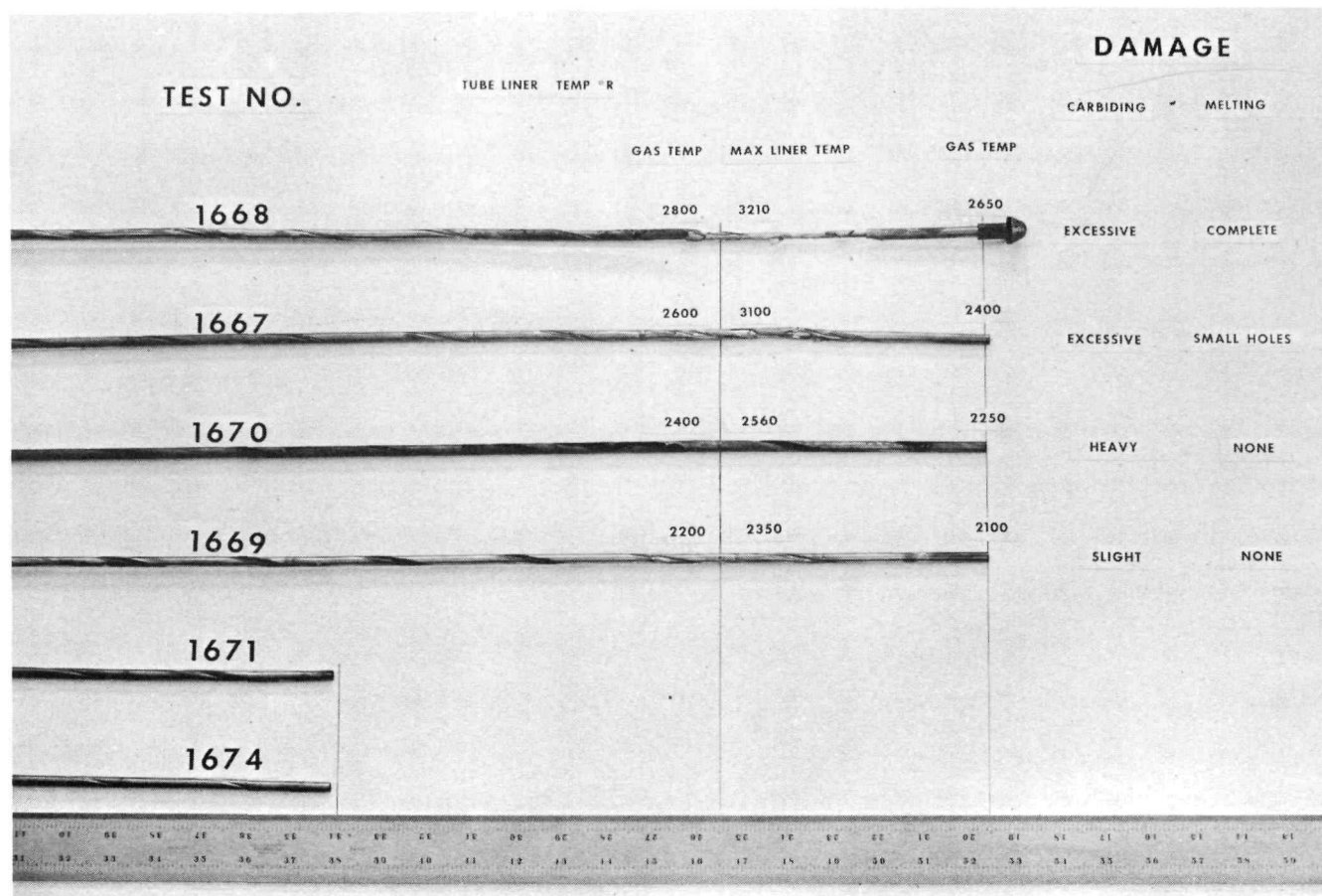
TRANSIENT TEST
BELOW
MELTING TEMP.

TRANSIENT TEST
ABOVE
MELTING TEMP.



STAINLESS LINER TUBE TRANSIENT TEST RESULTS

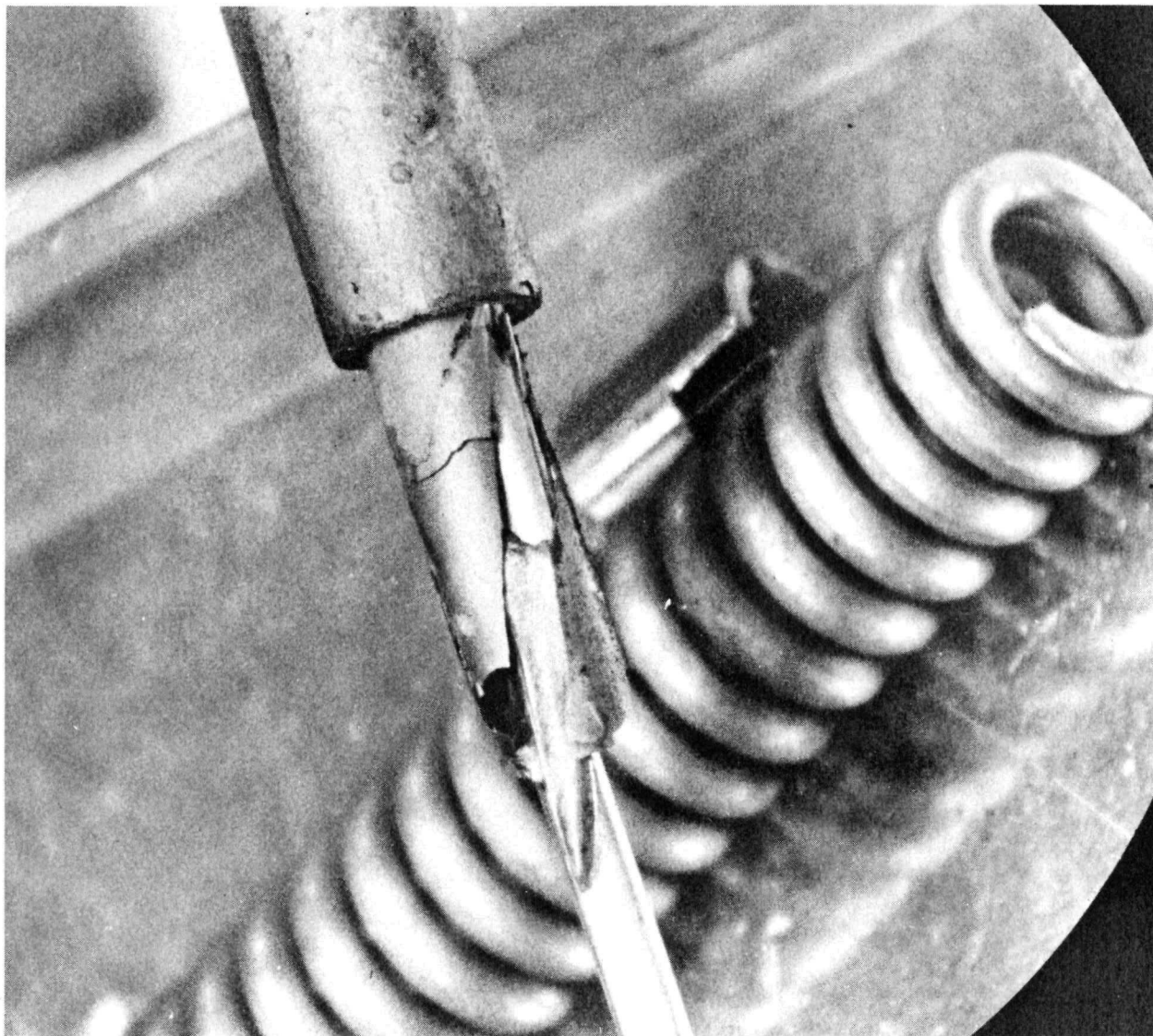
TEST NO.	TUBE LINER TEMP °R		DAMAGE	
	GAS TEMP	MAX LINER TEMP	CARBIDING	MELTING
1668	2800	3210	EXCESSIVE	COMPLETE
1667	2600	3100	EXCESSIVE	SMALL HOLES
1670	2400	2560	HEAVY	NONE
1669	2200	2350	SLIGHT	NONE
1671				
1674				



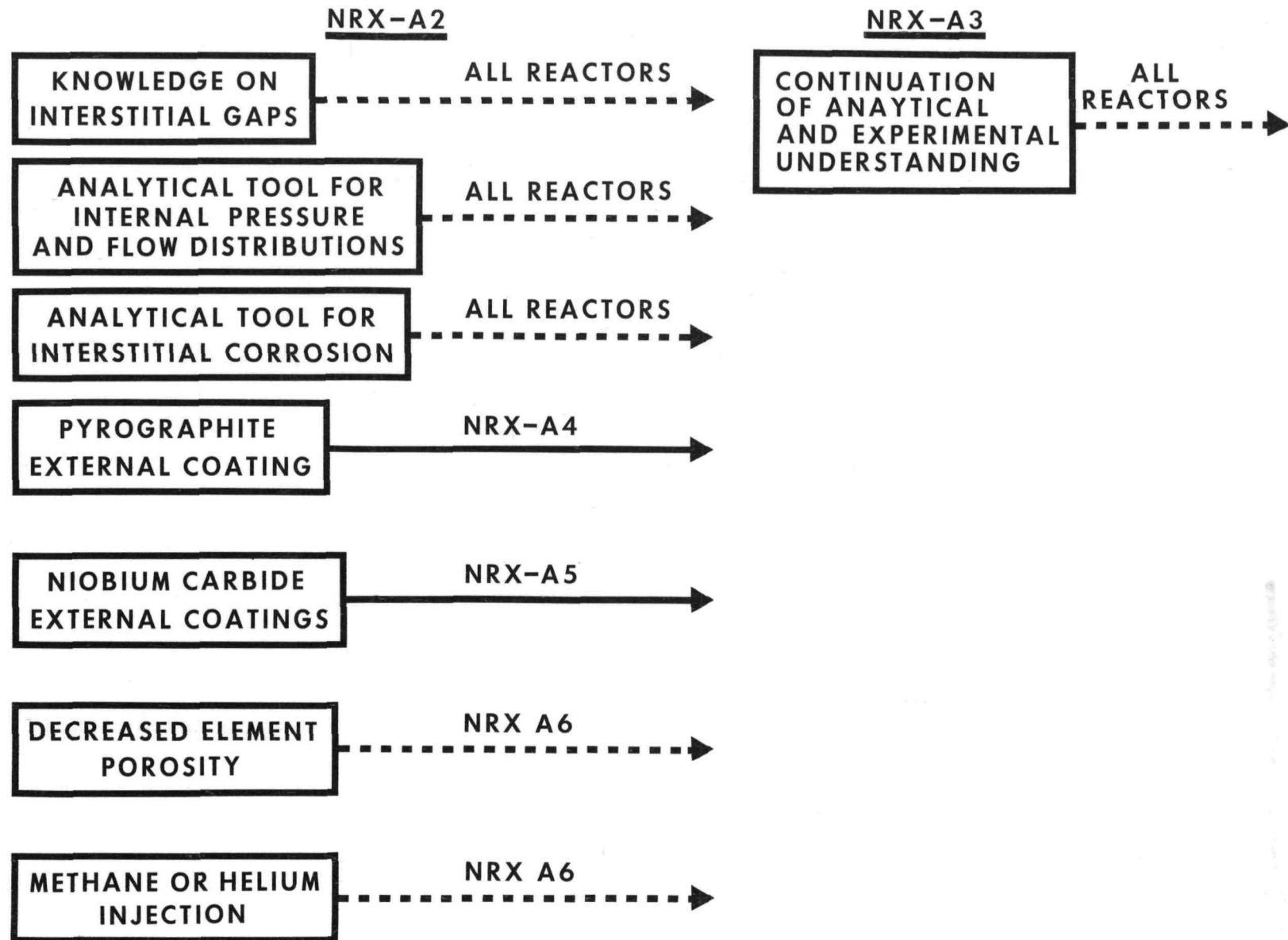
Tie Rod Liner Tubes After Exposure
 to Simulated NRX-A3 Environment

TEST NO.		H ₂ FLOW	CORROSION RATE
		lb/sec	mils/min
1670	4600		
SIDE #1	DEPTH .037 Inch (Max)	.0125	2
SIDE #2			
1671	4600		
SIDE #1	DEPTH .108 Inch (max)	.0060	6
SIDE #2			
1674	4700		
SIDE #1	DEPTH .130 inch (max)	.0155	7
SIDE #2			

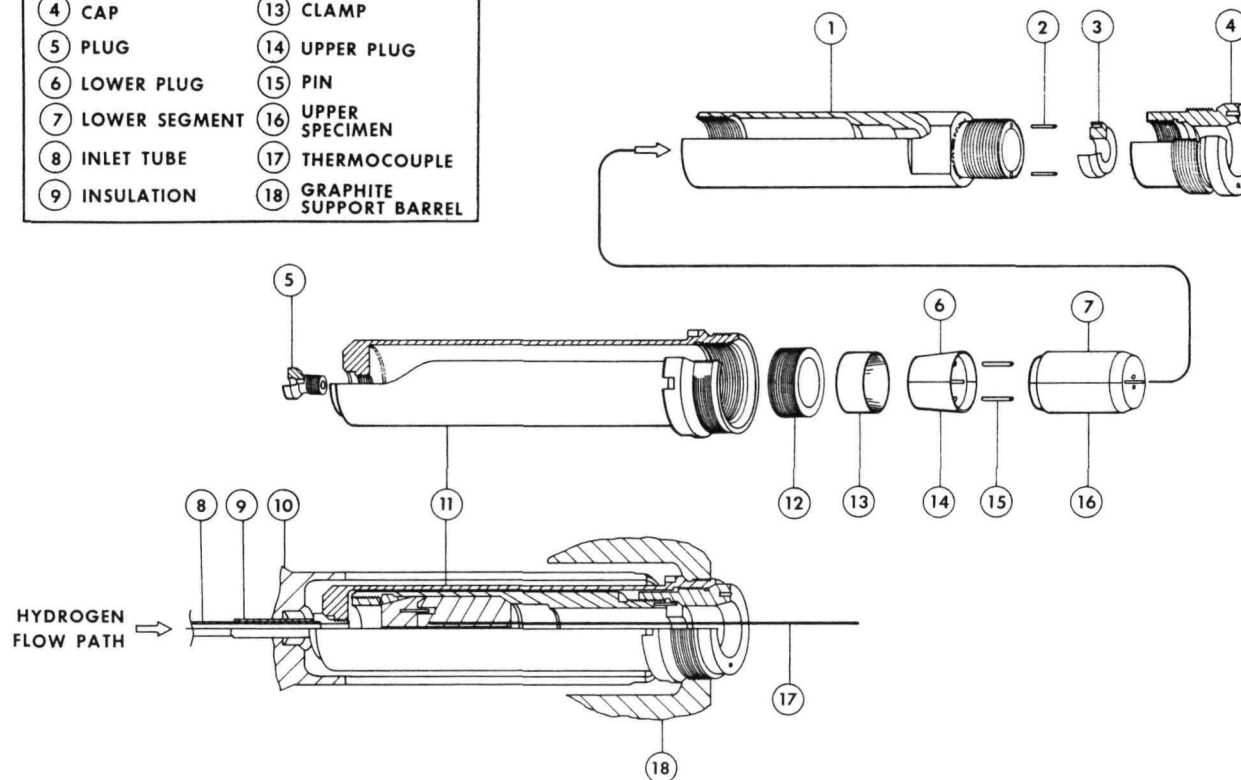
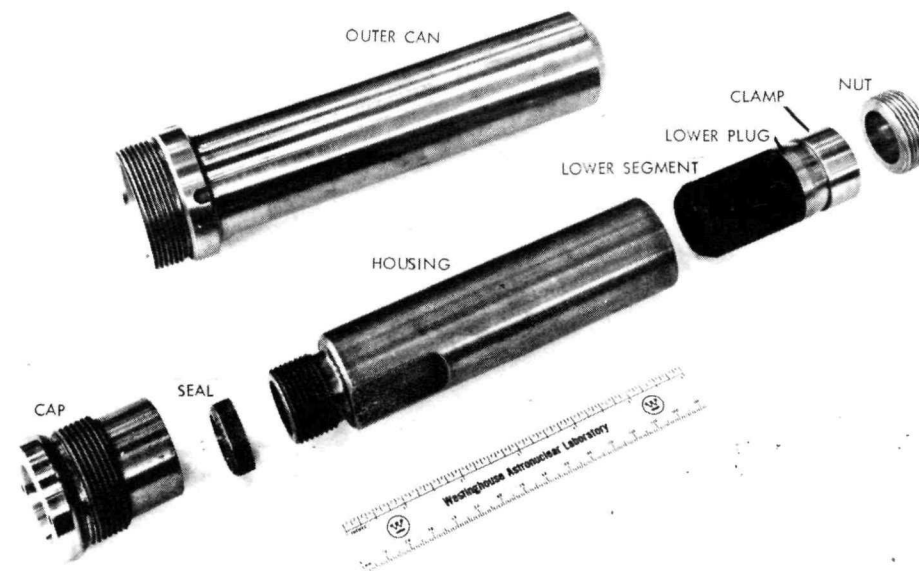
Center Elements After Exposure
 to Simulated NRX-A3 Environment



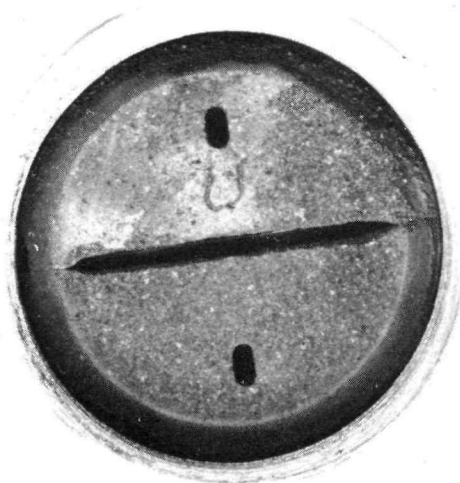
INTERSTITIAL CORROSION



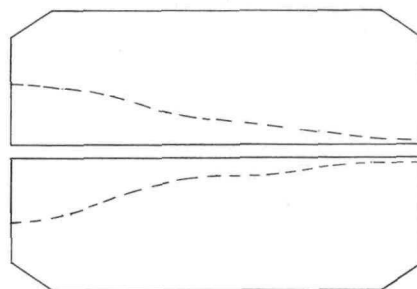
- | | |
|-----------------|----------------------------|
| 1 HOUSING | 10 HEATER |
| 2 PIN | 11 OUTER CAN |
| 3 SEAL | 12 NUT |
| 4 CAP | 13 CLAMP |
| 5 PLUG | 14 UPPER PLUG |
| 6 LOWER PLUG | 15 PIN |
| 7 LOWER SEGMENT | 16 UPPER SPECIMEN |
| 8 INLET TUBE | 17 THERMOCOUPLE |
| 9 INSULATION | 18 GRAPHITE SUPPORT BARREL |



INLET OF .006-INCH GAP
TEST FIXTURE AFTER TEST



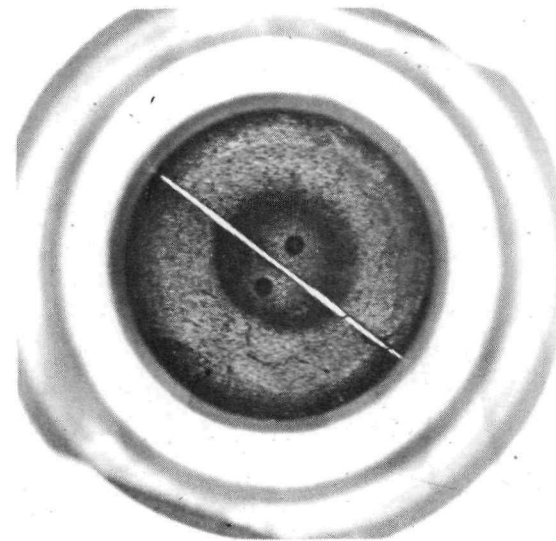
INLET



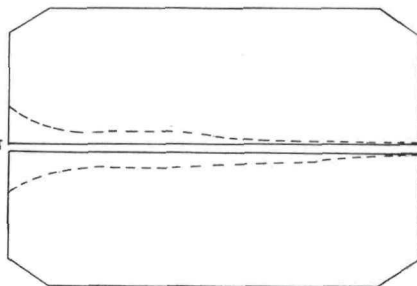
EXIT

INTERSTITIAL CORROSION TESTS .006-INCH ORIGINAL GAP THICKNESS
SURFACE REGRESSION AFTER 30 MIN. RUN AT 3521°R INLET AND 3539°R EXIT
INLET PRESSURE = 208 PSIG
FLOW RATE = .0000926 LB/SEC HYDROGEN
GAP LENGTH = 2-INCHES
GAP WIDTH = 1-INCH
WEIGHT LOSS = 2.363 GMS

EXIT OF .006-INCH GAP
TEST FIXTURE AFTER TEST

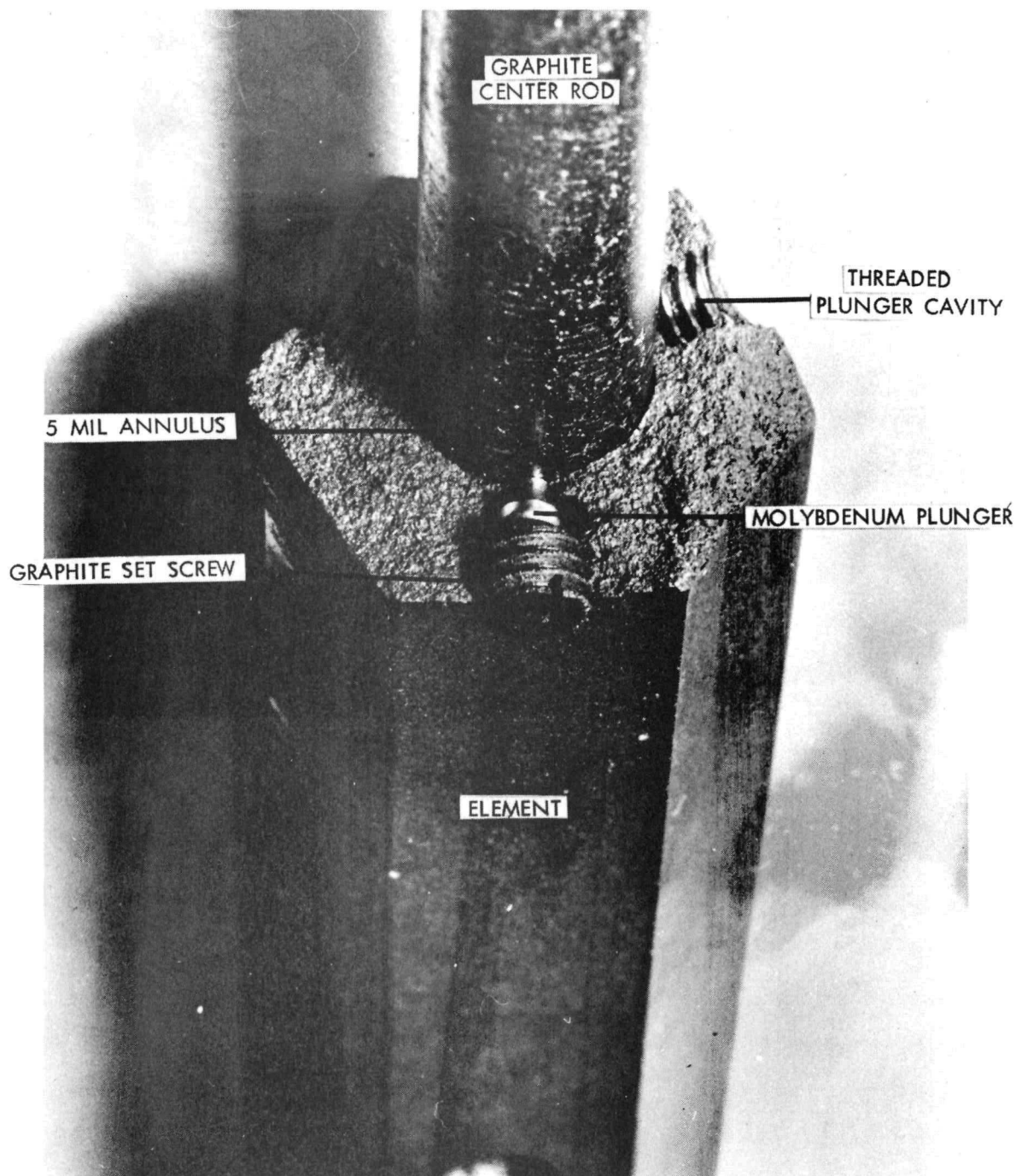


INLET



EXIT

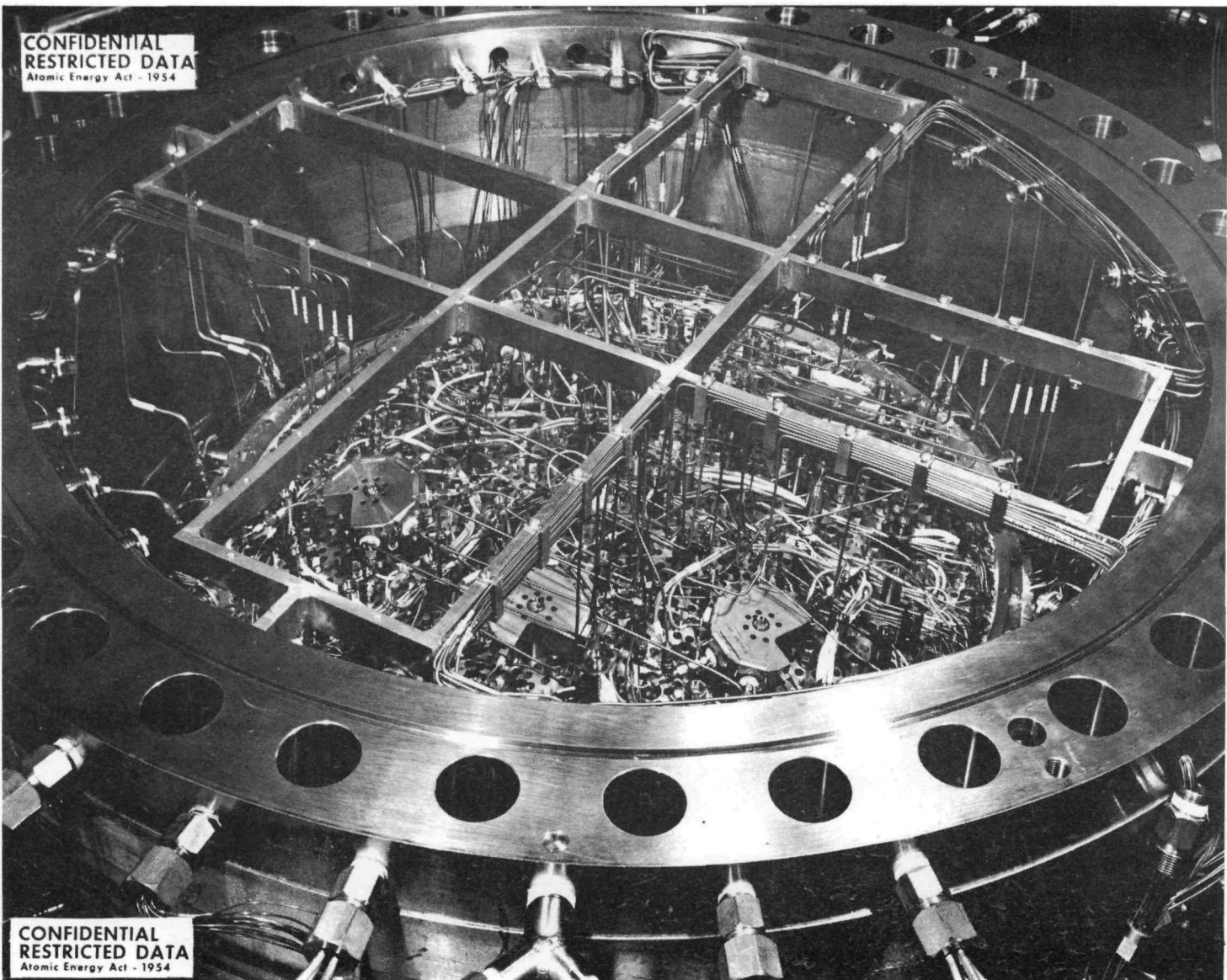
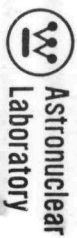
INTERSTITIAL CORROSION TEST .004-INCH ORIGINAL GAP THICKNESS
SURFACE REGRESSION AFTER 27 MIN. RUN AT 3443°R INLET AND 3485°R EXIT
INLET PRESSURE = 222 PSIG
FLOW RATE = .0000973 LB/SEC HYDROGEN
GAP LENGTH = 2-INCHES
GAP WIDTH = 1-INCH
WEIGHT LOSS = 1.966 GMS



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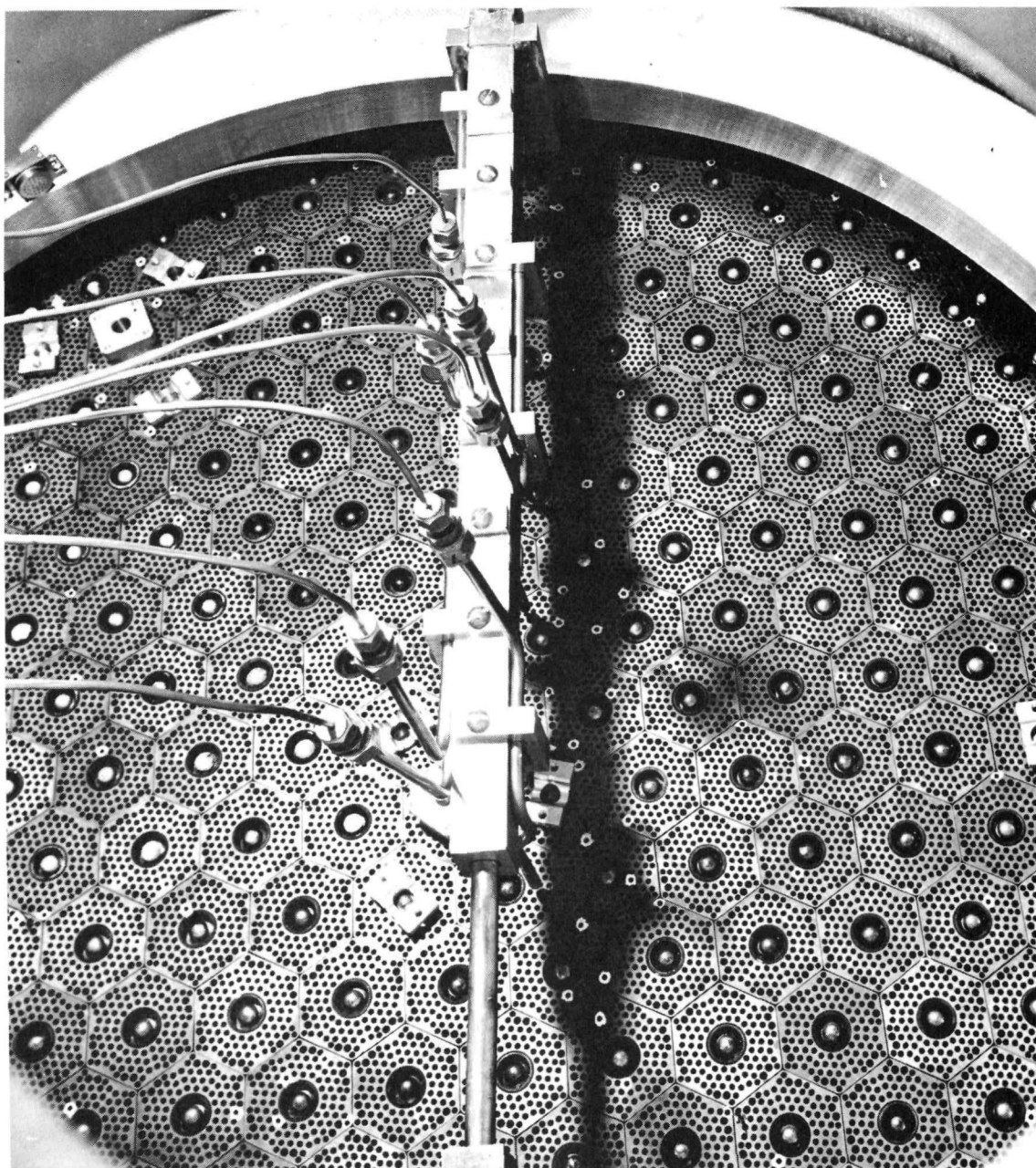


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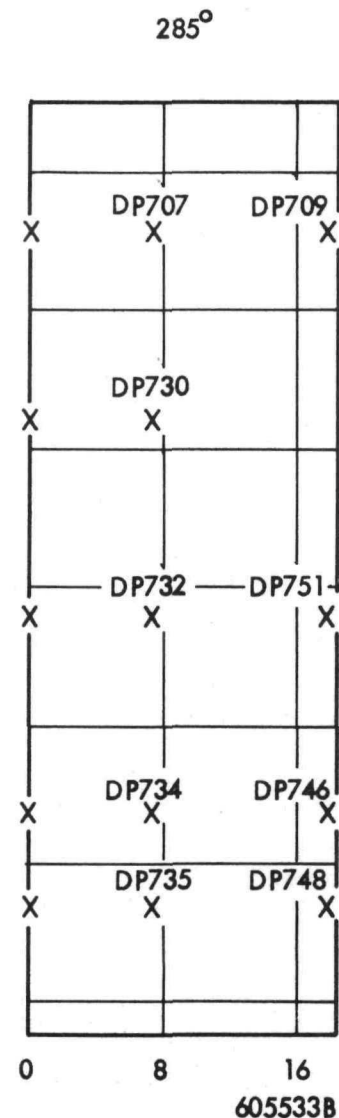
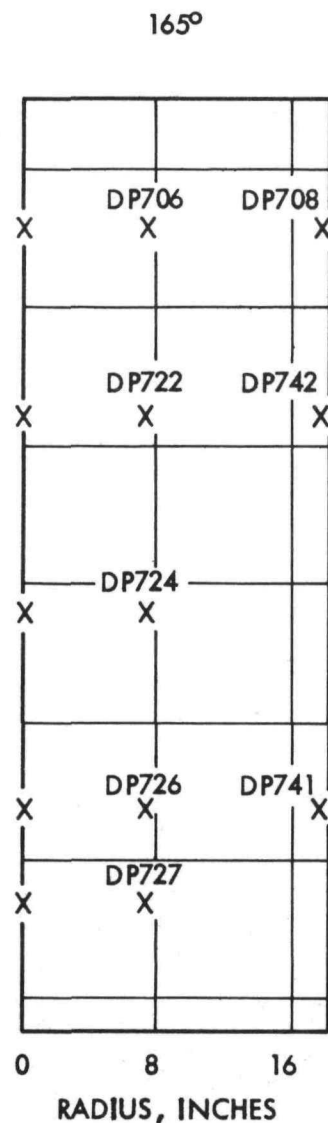
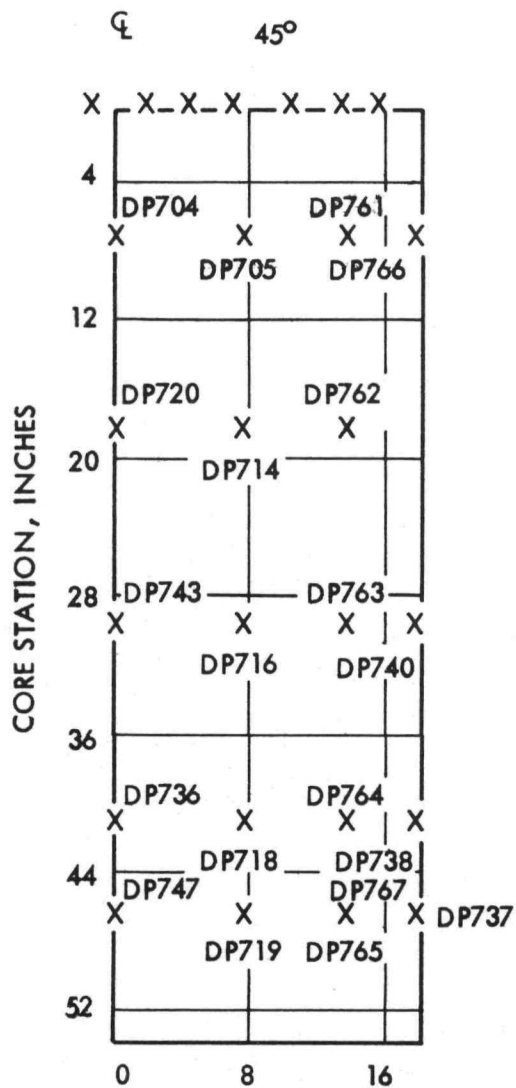
RESTRICTED DATA

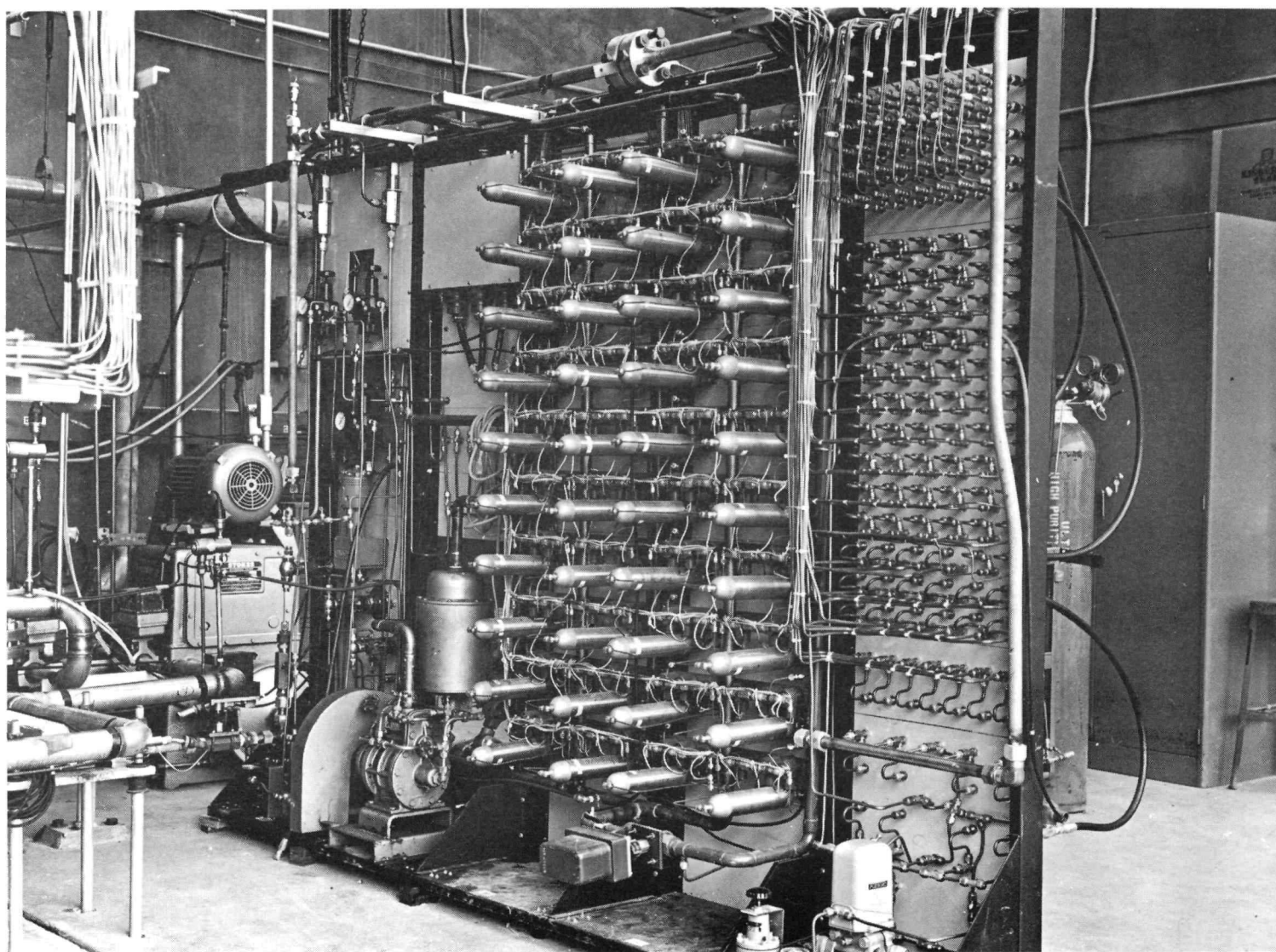
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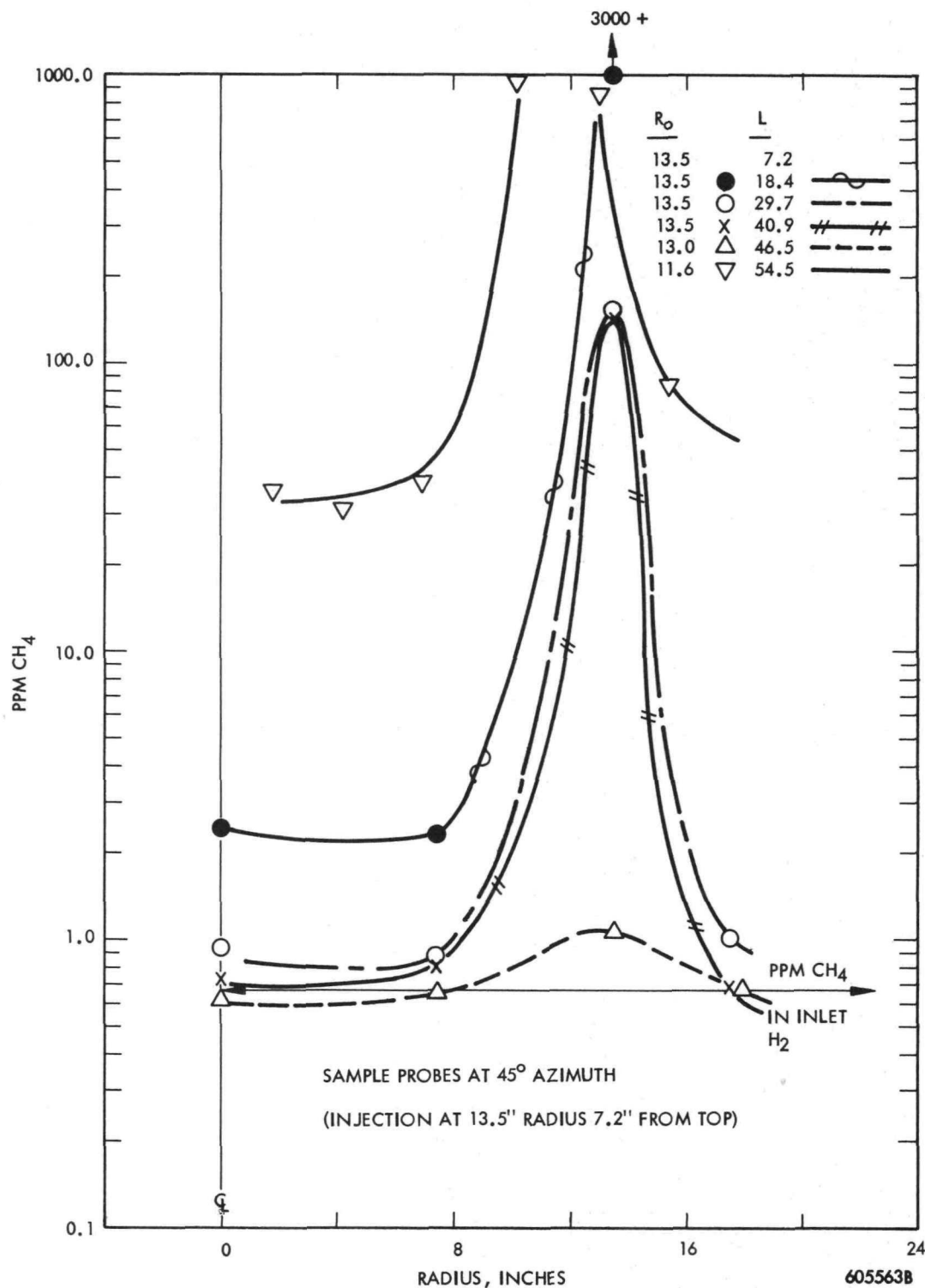
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RESTRICTED DATA
Atomic Energy Act - 1954





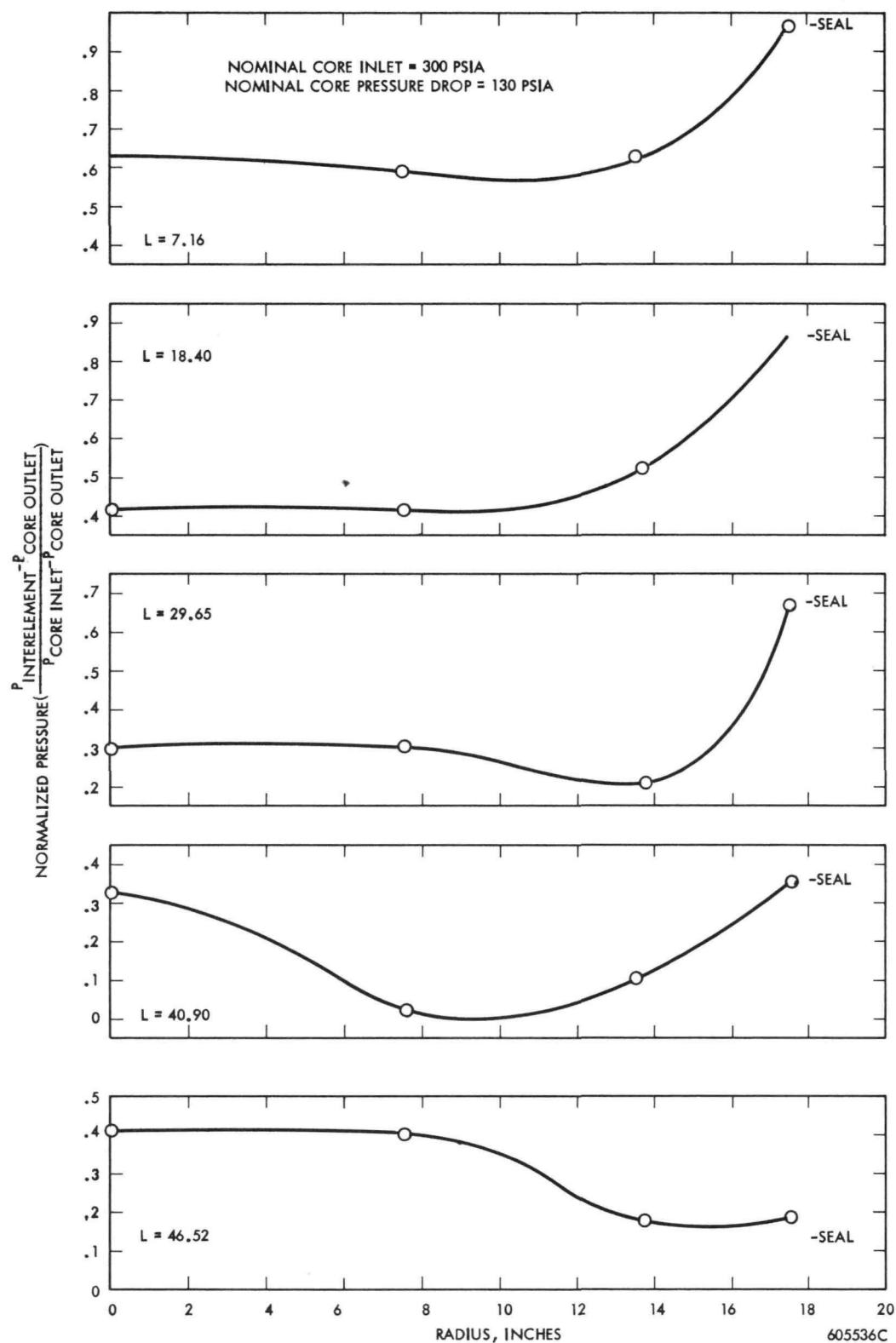


CONFIDENTIAL RESTRICTED DATA

Atomic Energy Act - 1954

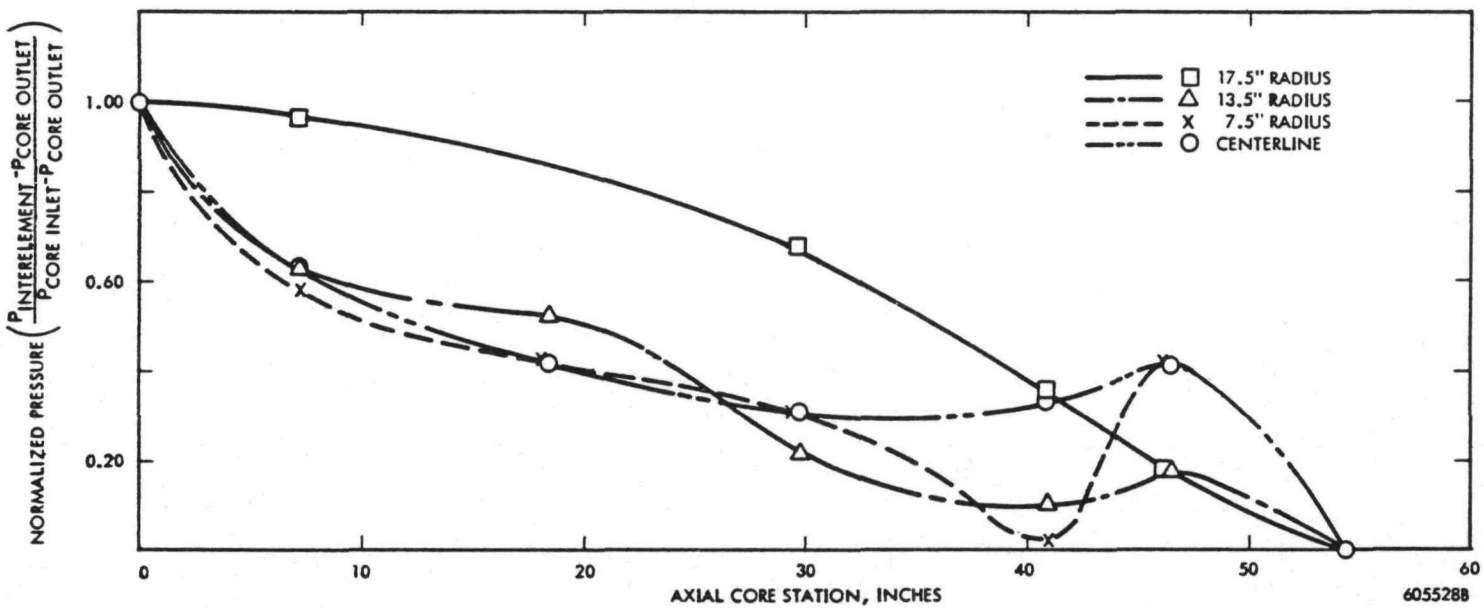


Astronuclear
Laboratory



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Atomic Energy Act - 1954



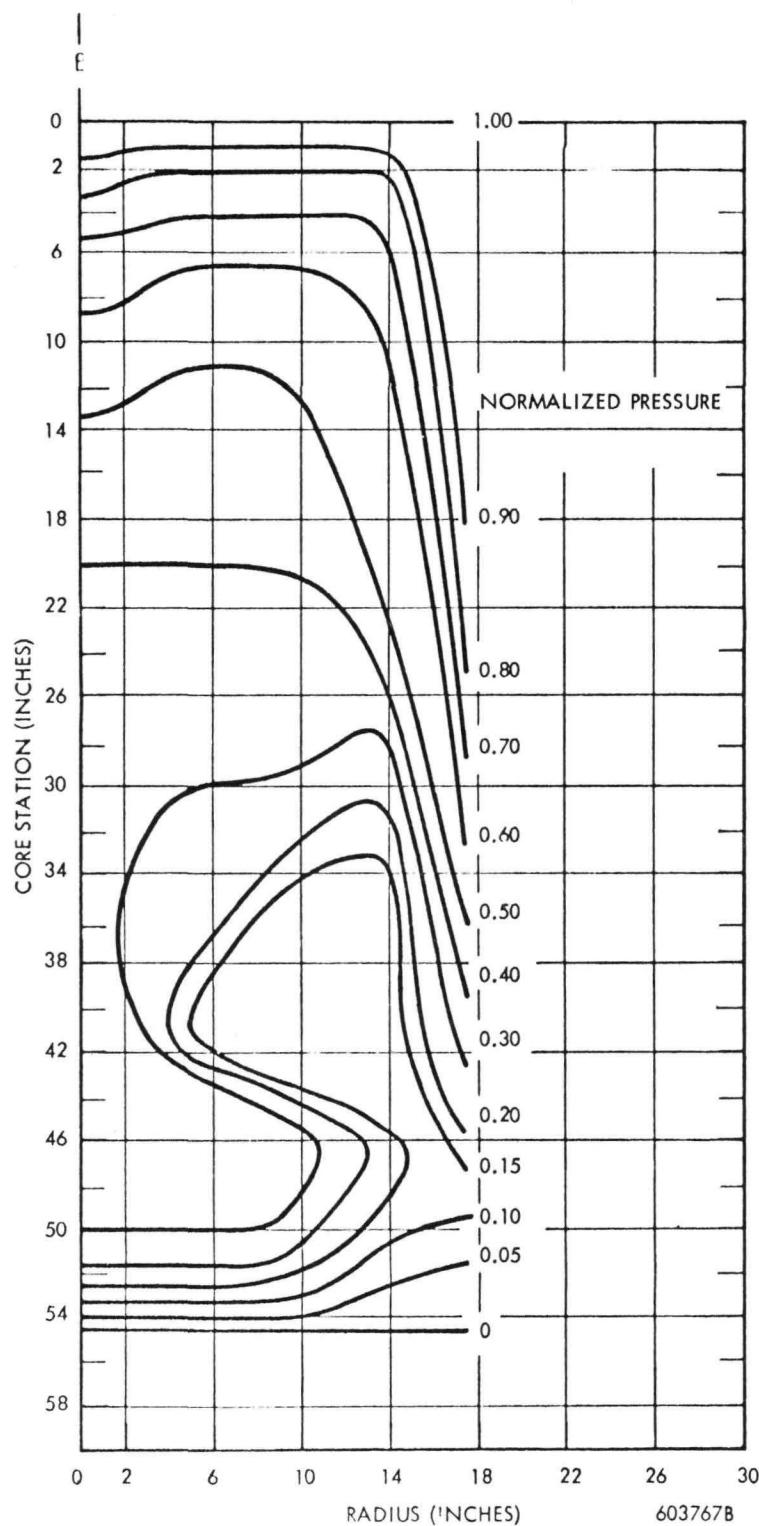
PLUGGED CORE FLOW TEST

**CONFIDENTIAL
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Atomic Energy Act - 1954



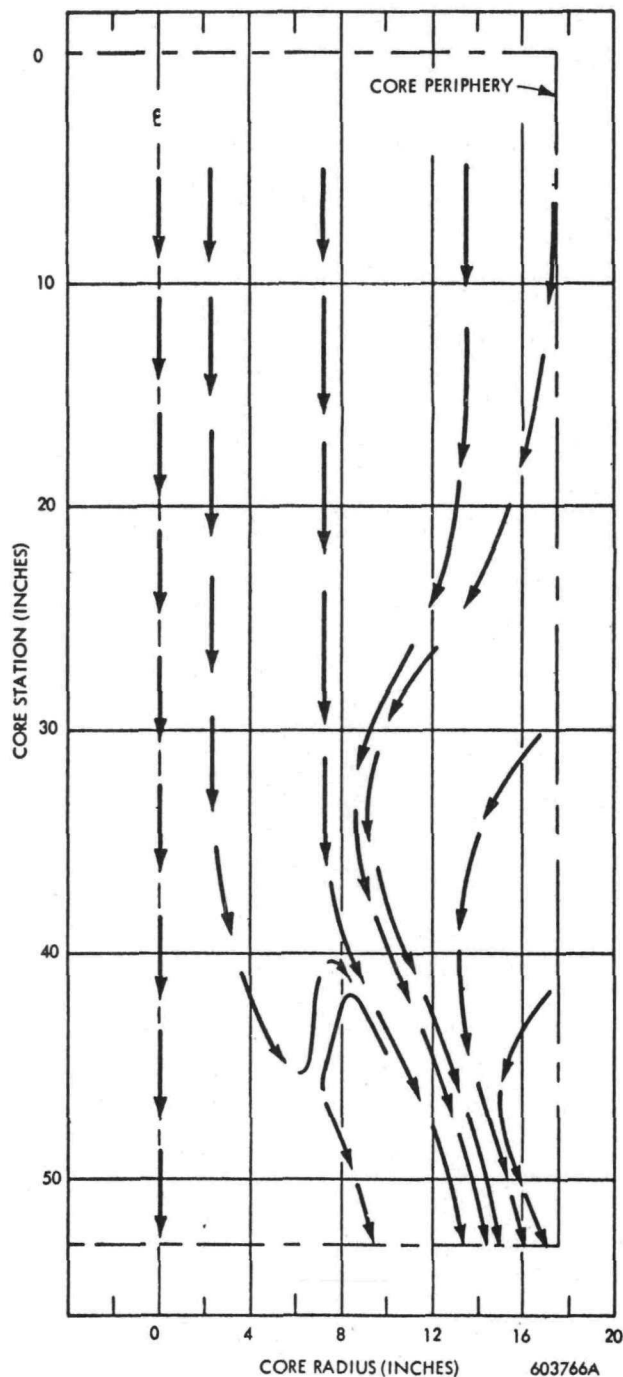
Astronuclear
Laboratory



ISOBAR MAP OF CORE
PLUGGED CORE FLOW TEST

**CONFIDENTIAL
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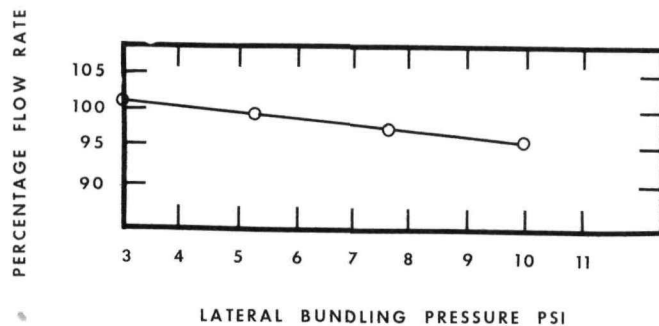
Atomic Energy Act - 1954



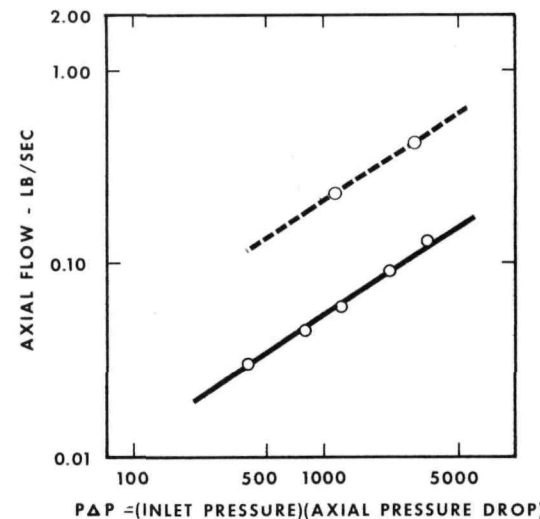
603766A
ESTIMATED STREAMLINE MAP
PLUGGED CORE TRACER GAS TEST SERIES

CORE EFFECTIVE GAP EXPERIMENTS

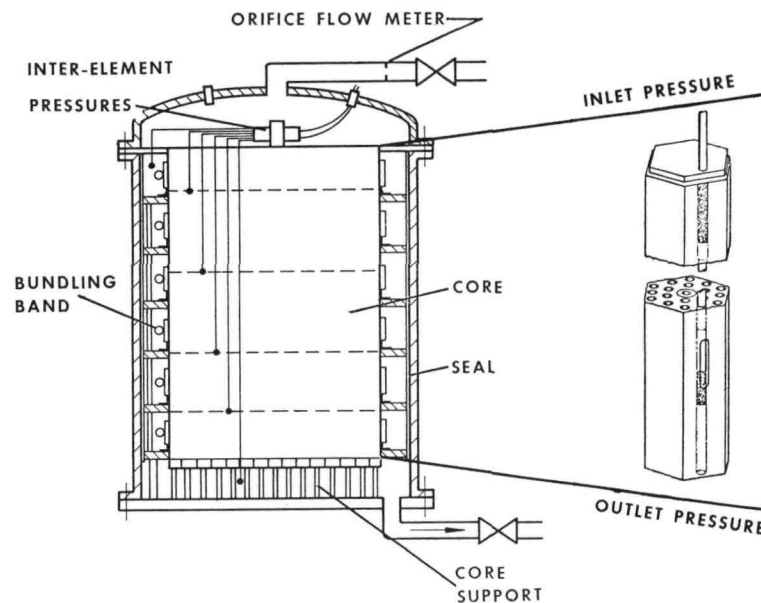
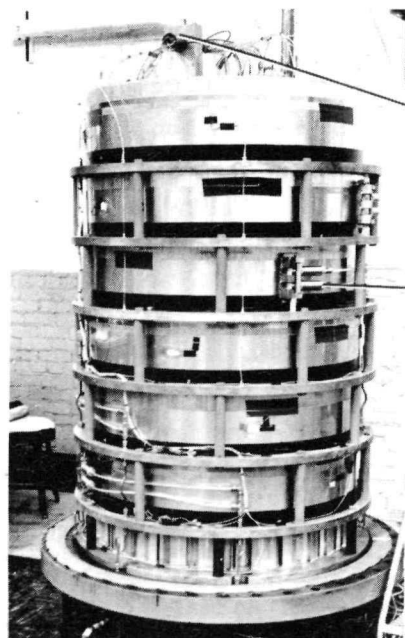
EFFECT OF BUNDLING ON FLOW



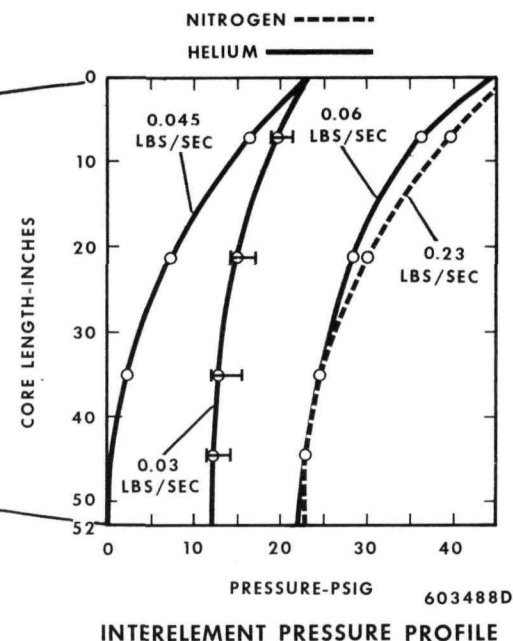
INTERELEMENT FLOW RATE



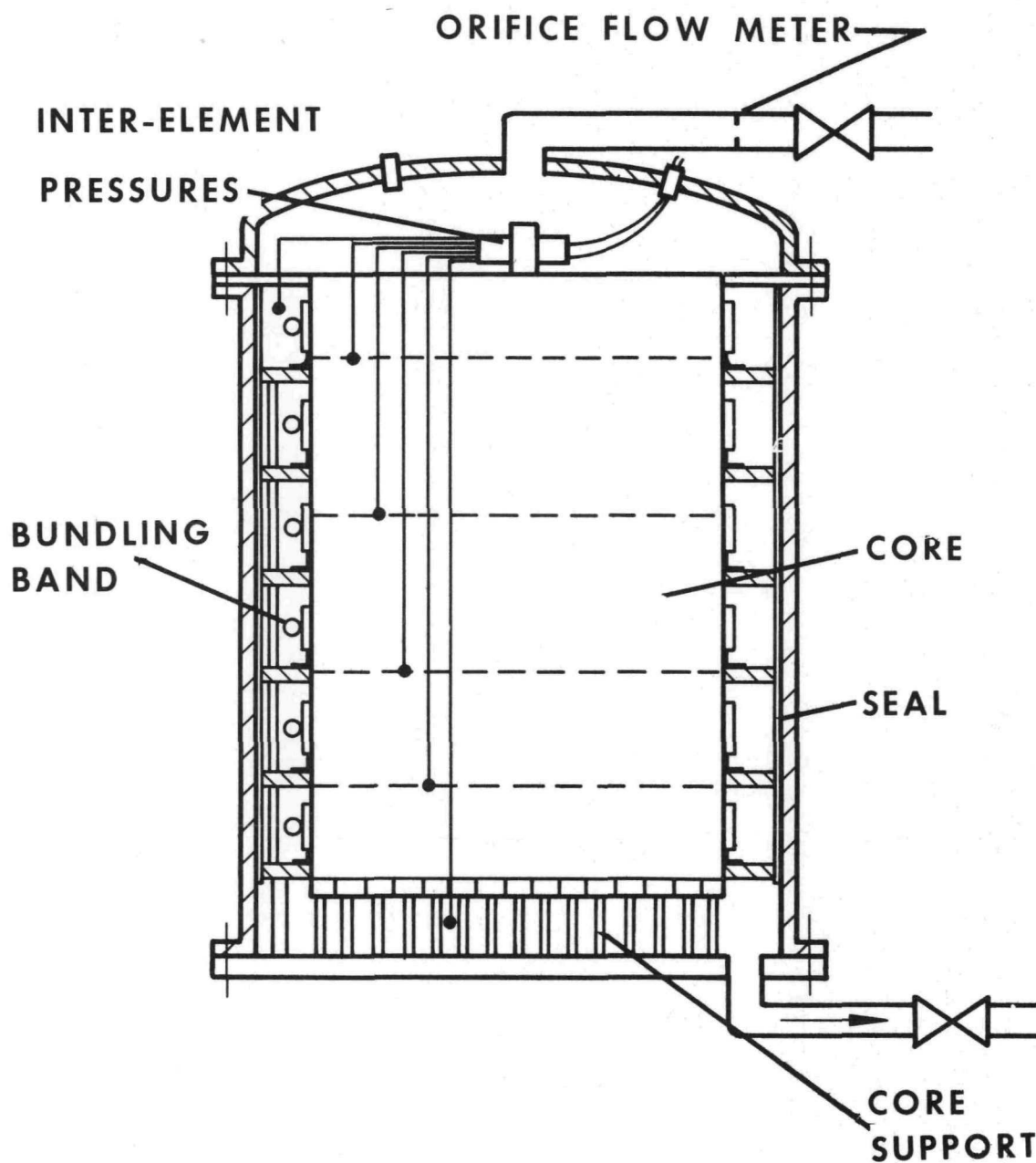
Page 89



CORE EFFECTIVE GAP SCHEMATIC

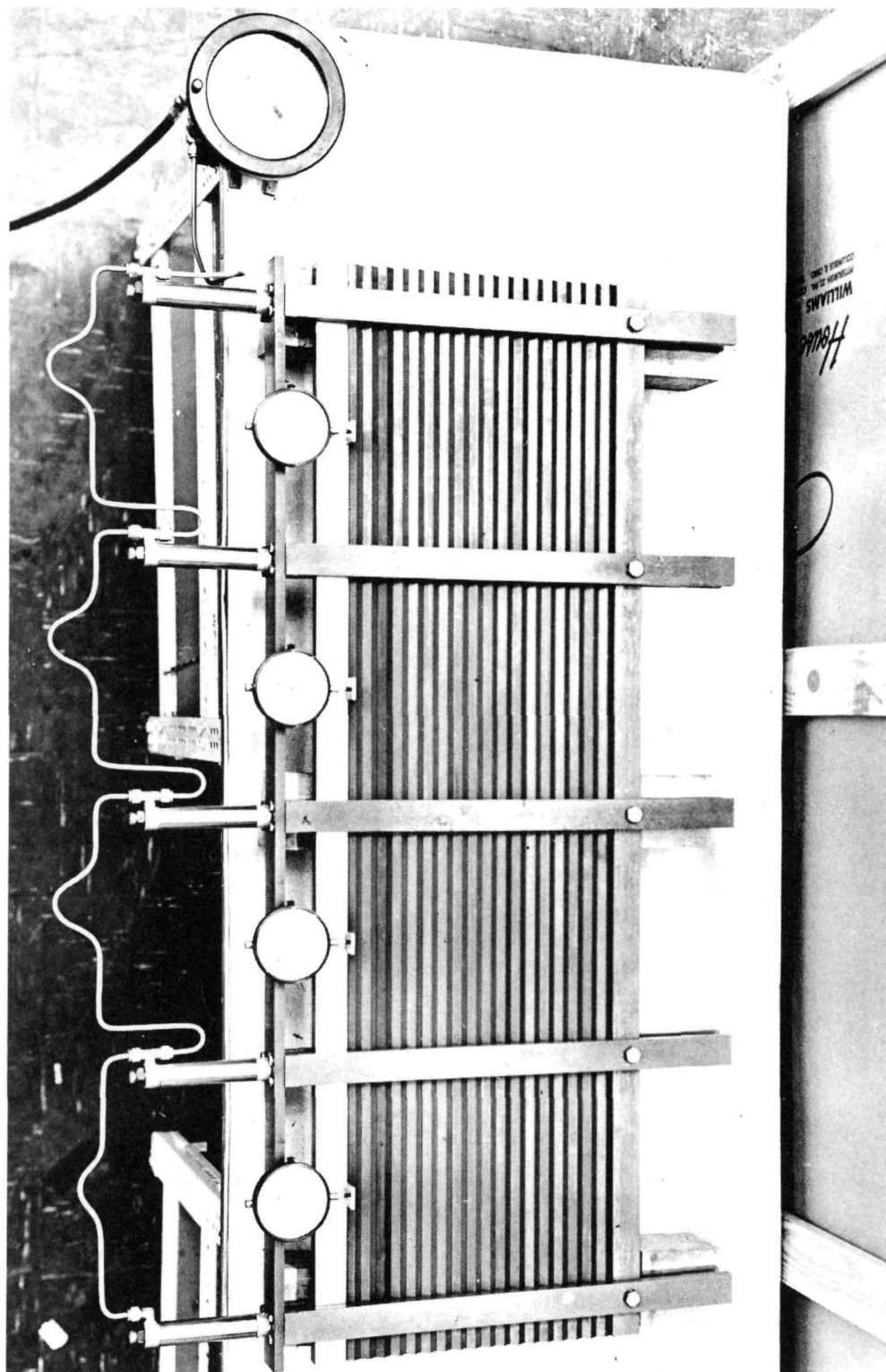


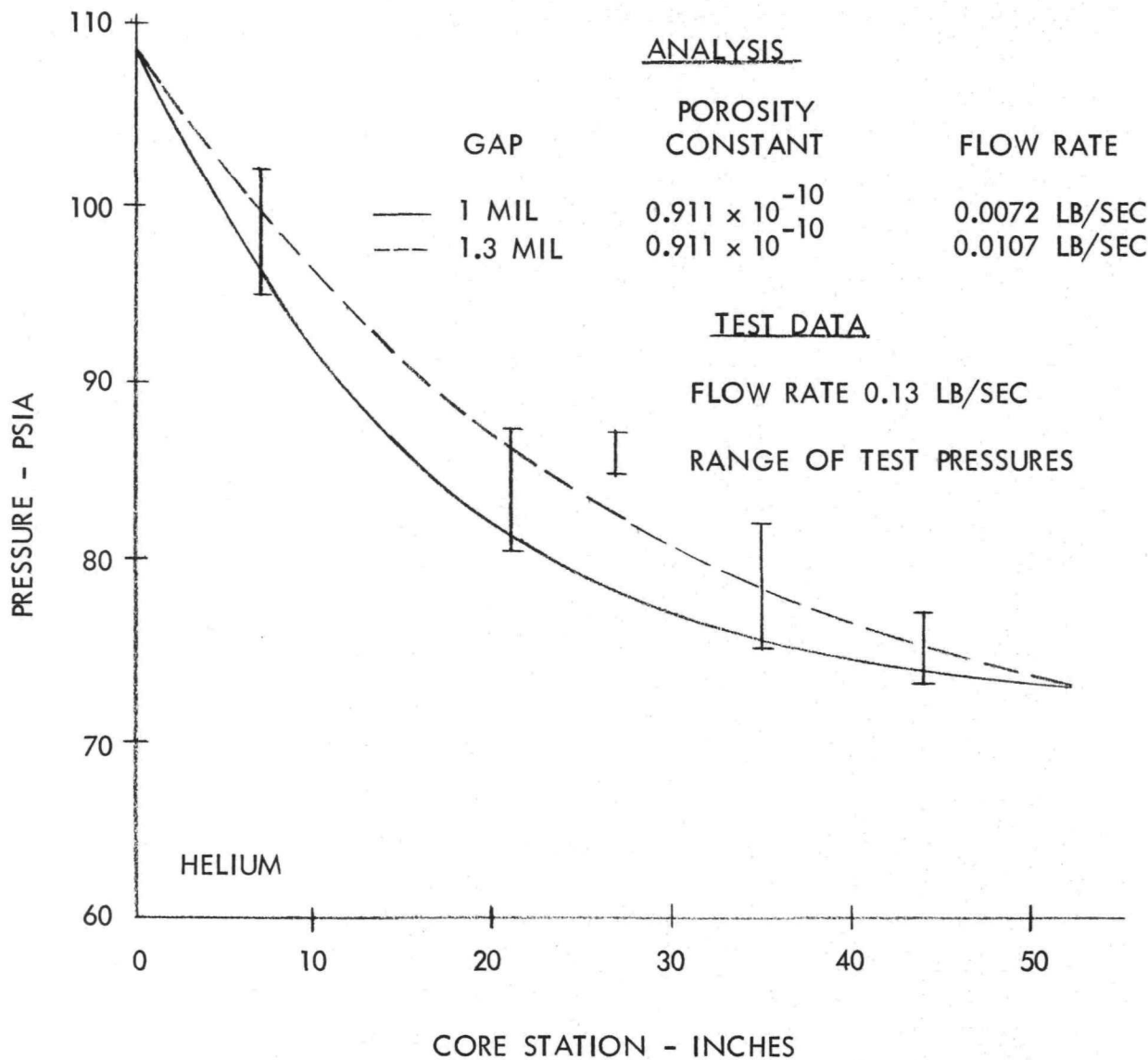
INTERELEMENT PRESSURE PROFILE



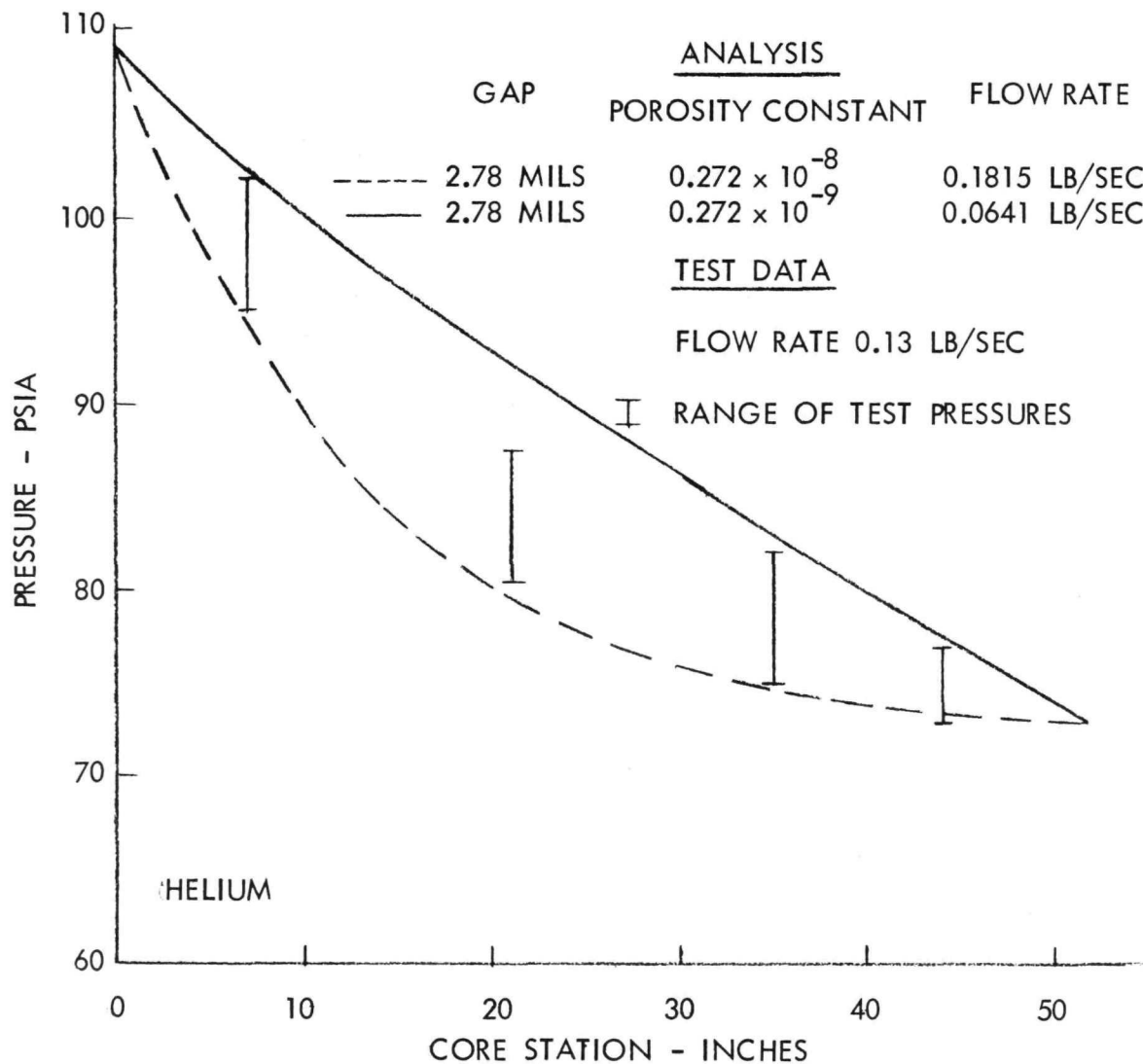
CORE EFFECTIVE GAP SCHEMATIC

603806





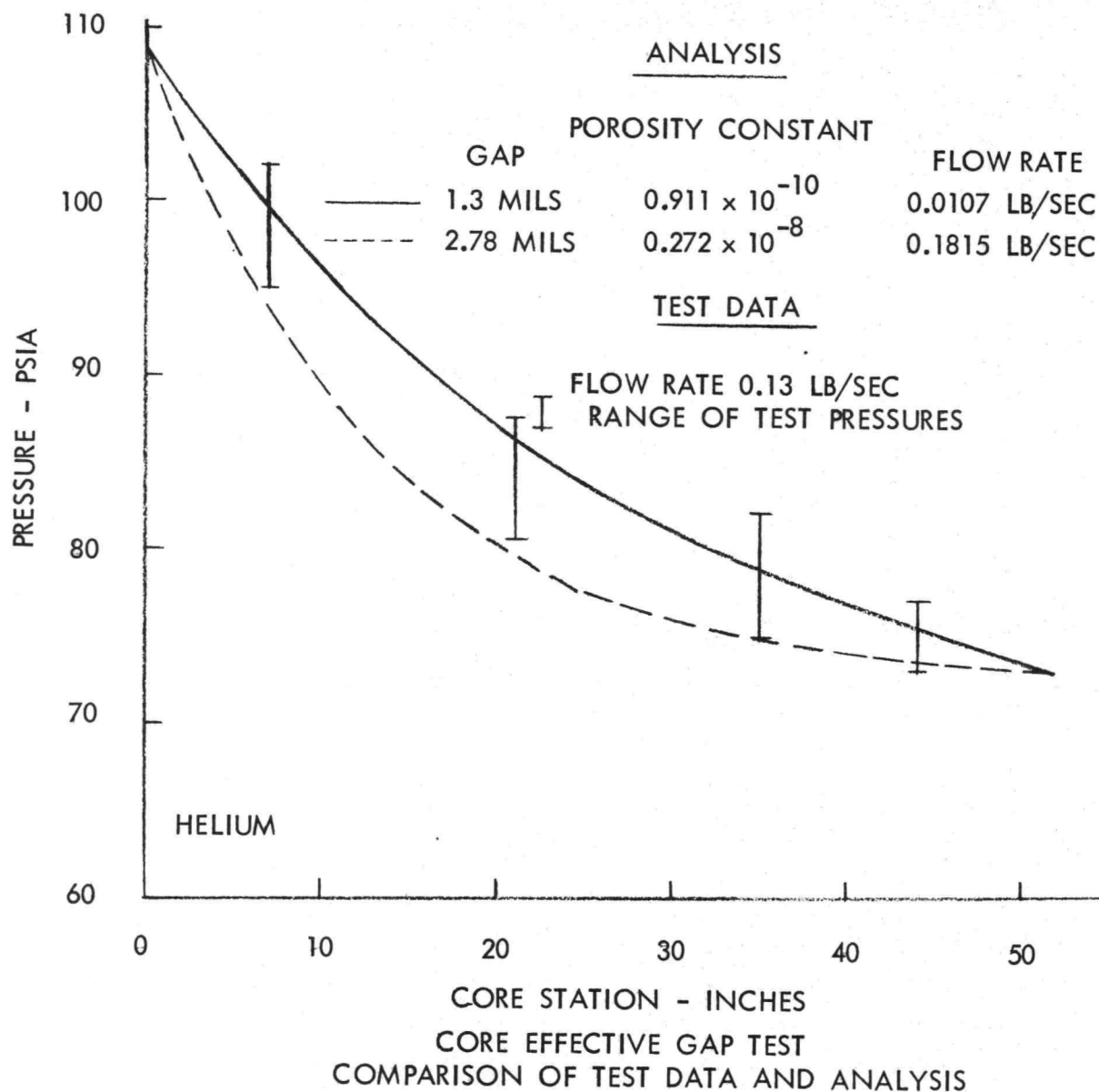
CORE EFFECTIVE GAP TEST COMPARISON OF TEST DATA
 AND ANALYSIS EFFECT OF INTERELEMENT GAP



CORE EFFECTIVE GAP TEST COMPARISON OF TEST DATA
 AND ANALYSIS EFFECT OF ELEMENT PERMEABILITY

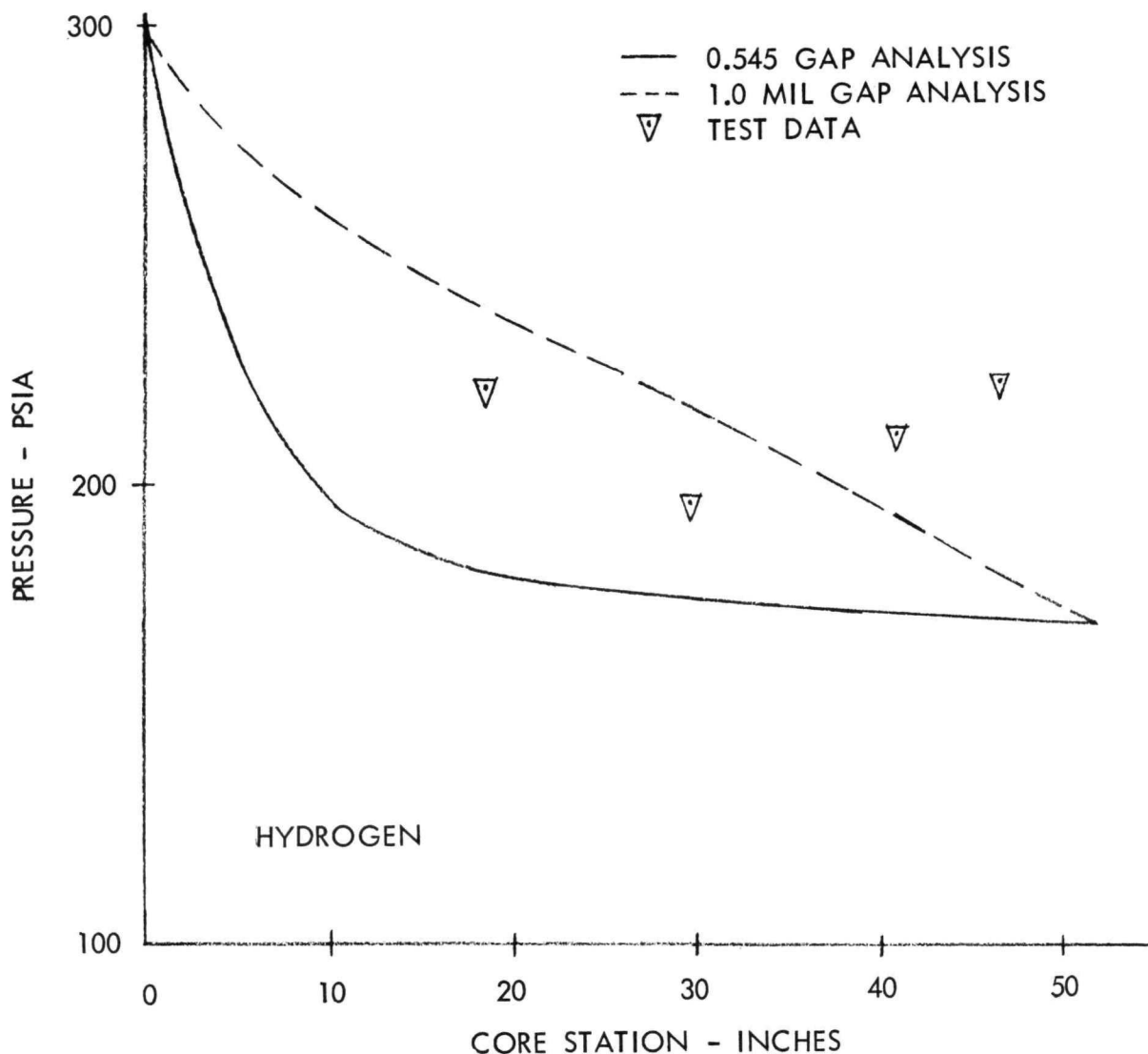
CONFIDENTIAL
RESTRICTED DATA

Atomic Energy Act - 1954



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RESTRICTED DATA

Atomic Energy Act - 1954



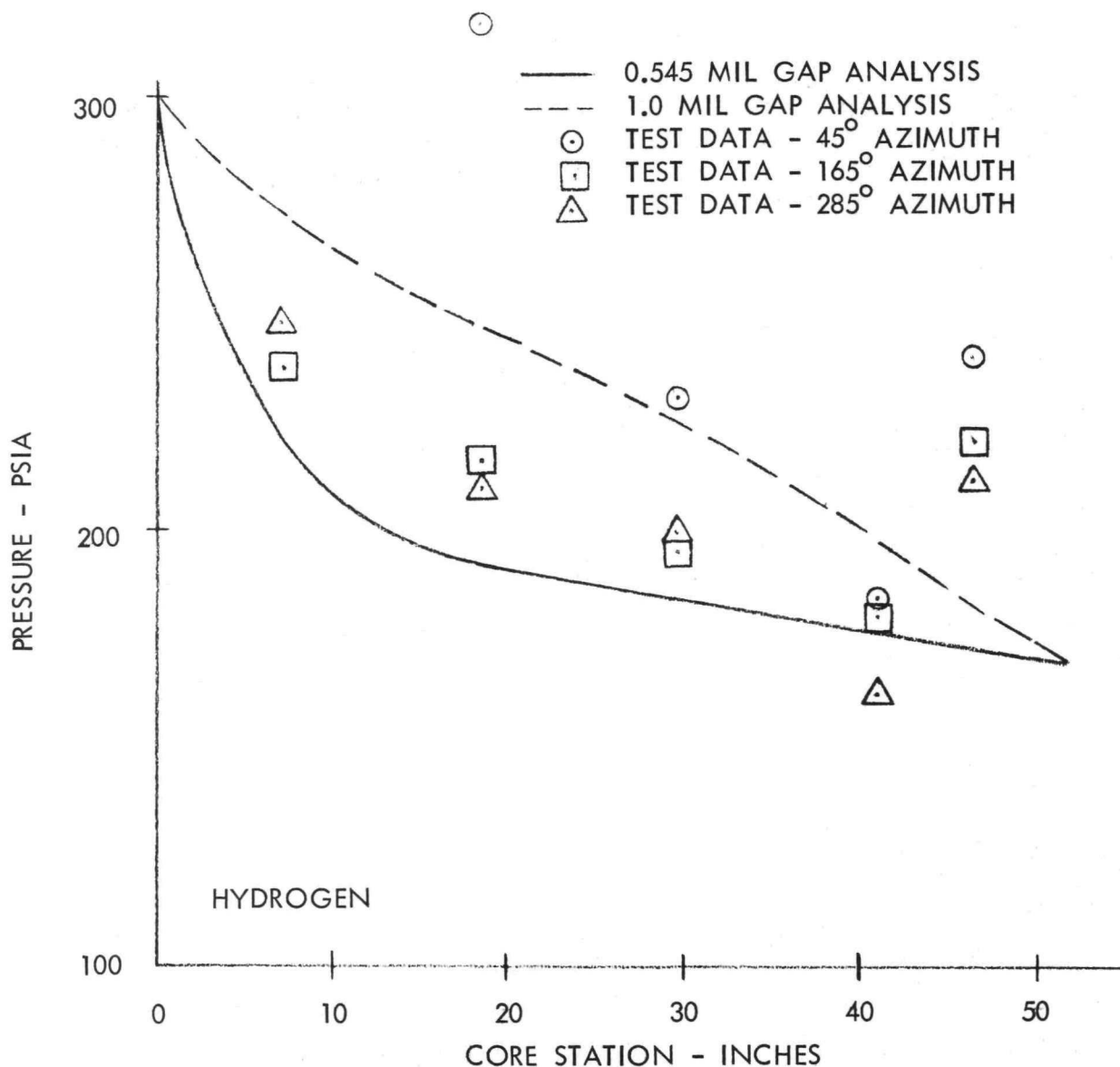
PLUGGED CORE TEST COMPARISON OF TEST DATA
AND ANALYSIS CENTERLINE PRESSURE DISTRIBUTION

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RESTRICTED DATA

Atomic Energy Act - 1954



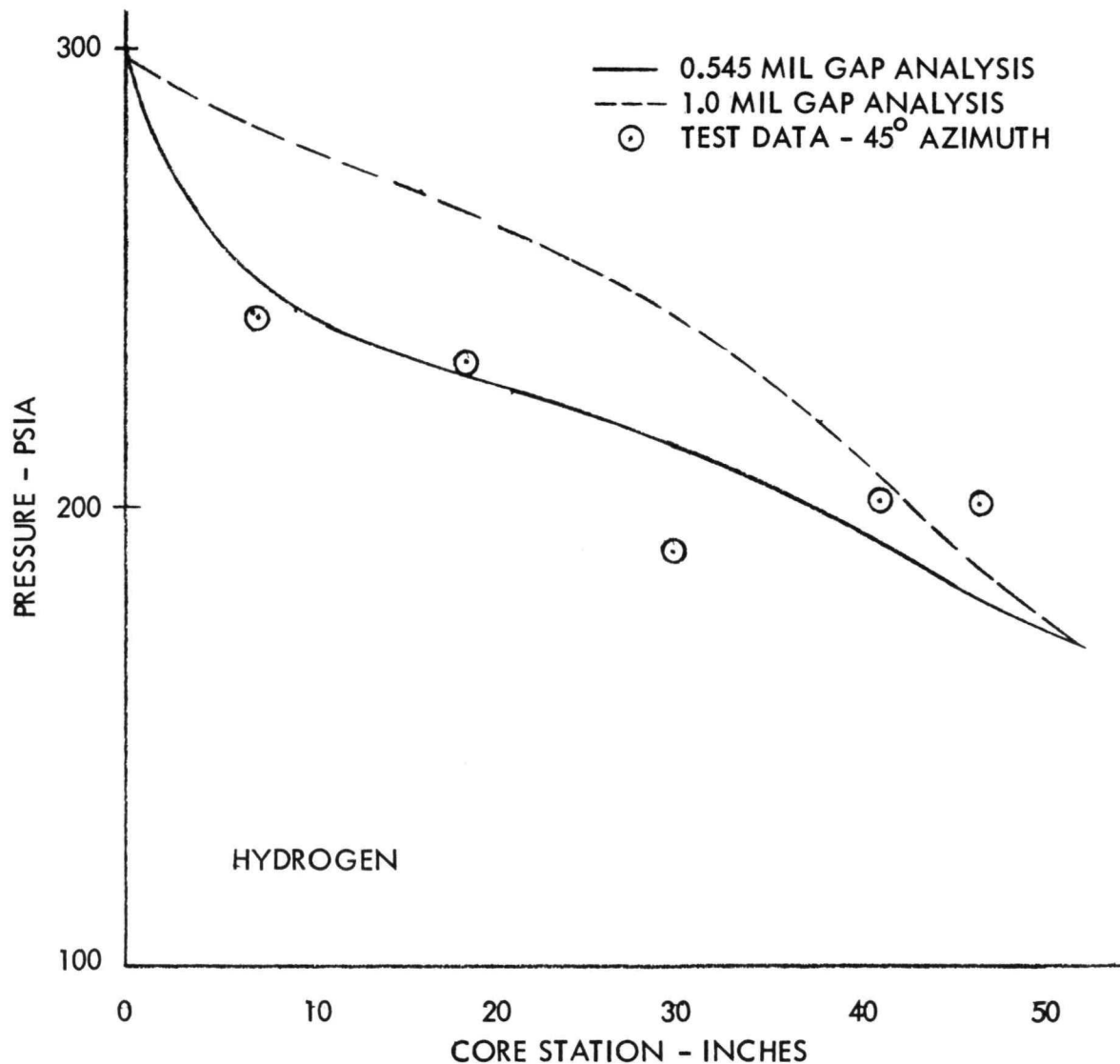
Astronuclear
Laboratory



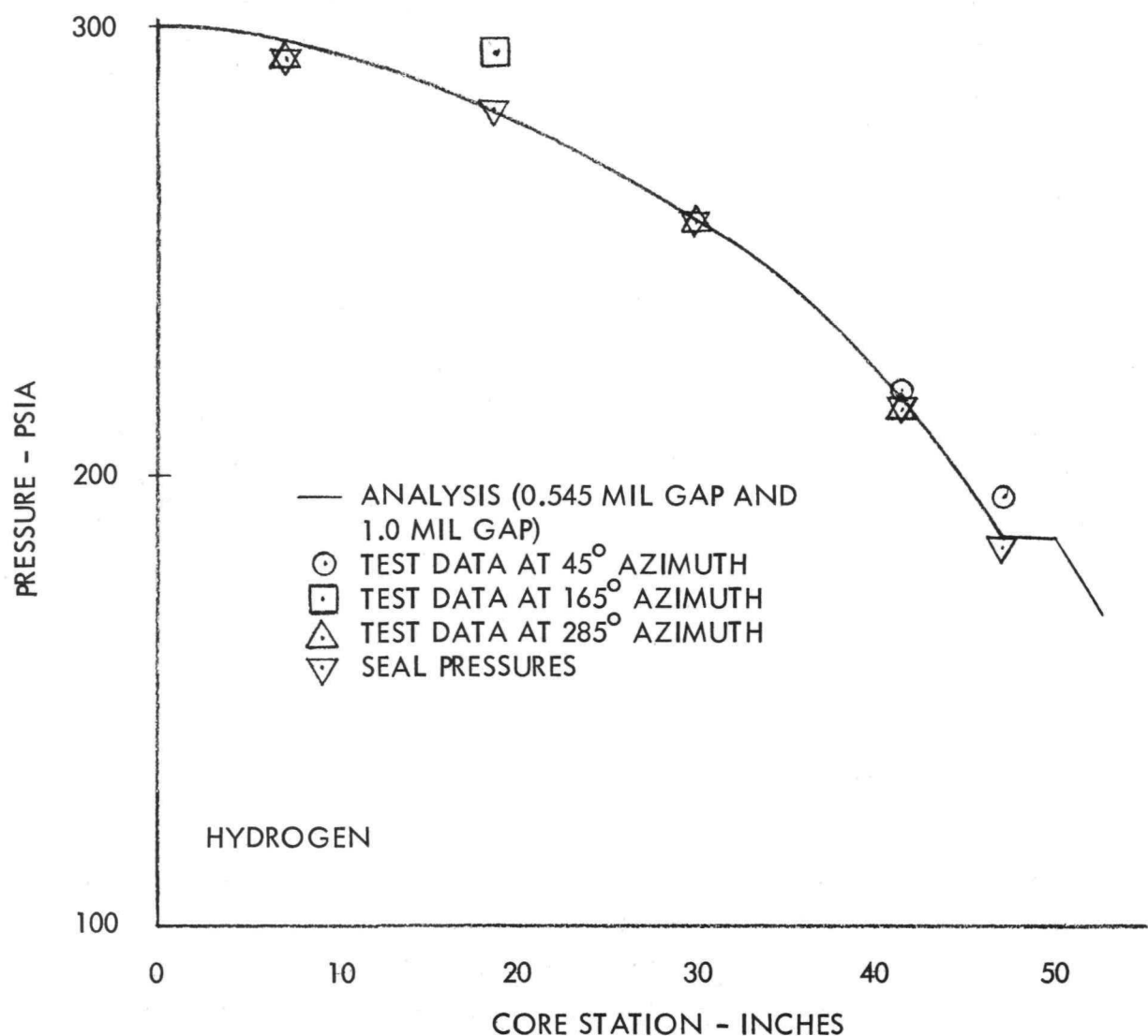
PLUGGED CORE TEST COMPARISON OF TEST DATA
AND ANALYSIS PRESSURE DISTRIBUTION AT 7.5 INCH RADIUS

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Atomic Energy Act - 1954



PLUGGED CORE TEST COMPARISON OF TEST DATA
AND ANALYSIS PRESSURE DISTRIBUTION AT 13.5 INCH RADIUS



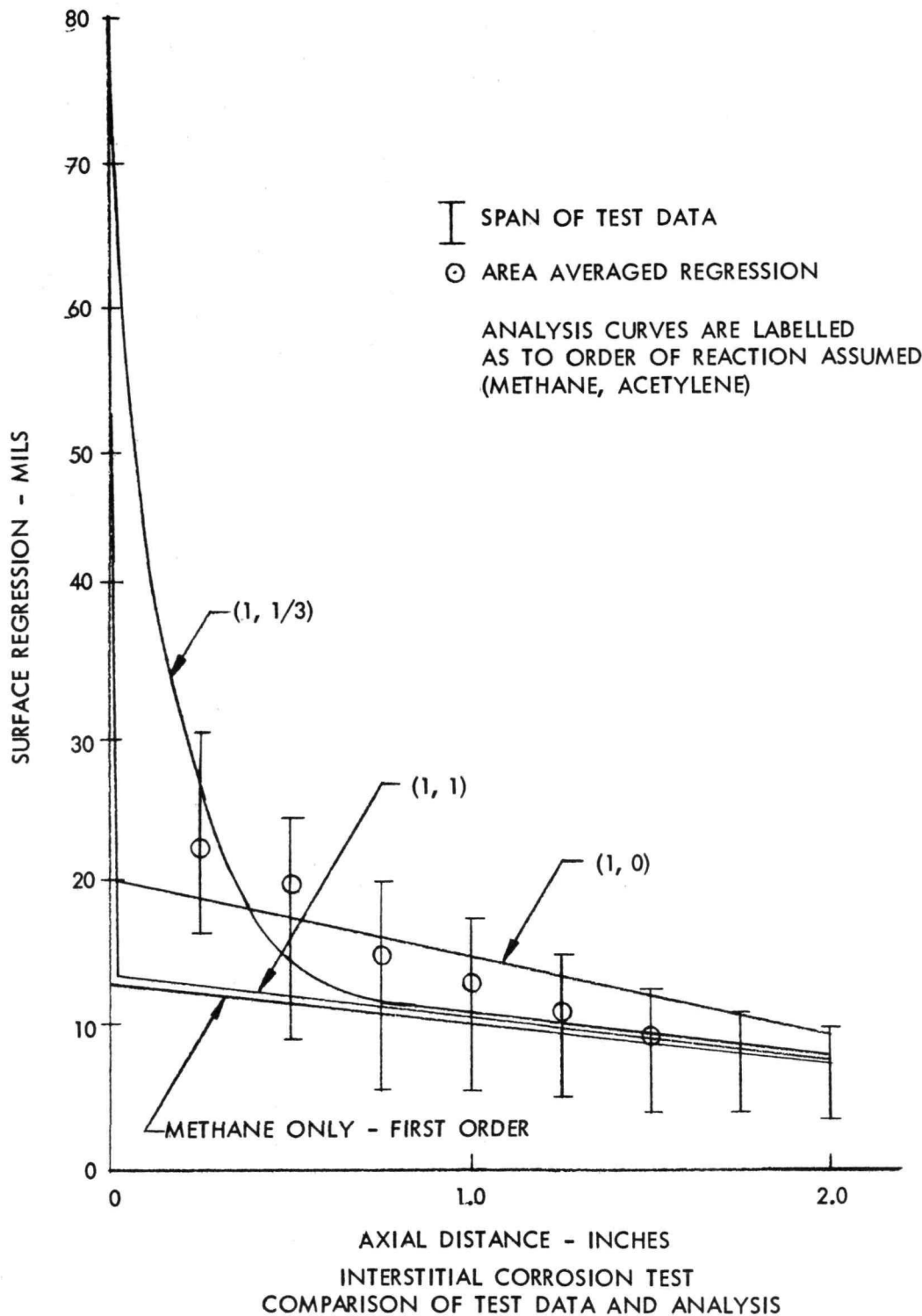
PLUGGED CORE TEST COMPARISON OF TEST DATA
AND ANALYSIS PRESSURE DISTRIBUTION AT 17.5 INCH RADIUS

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Atomic Energy Act - 1954



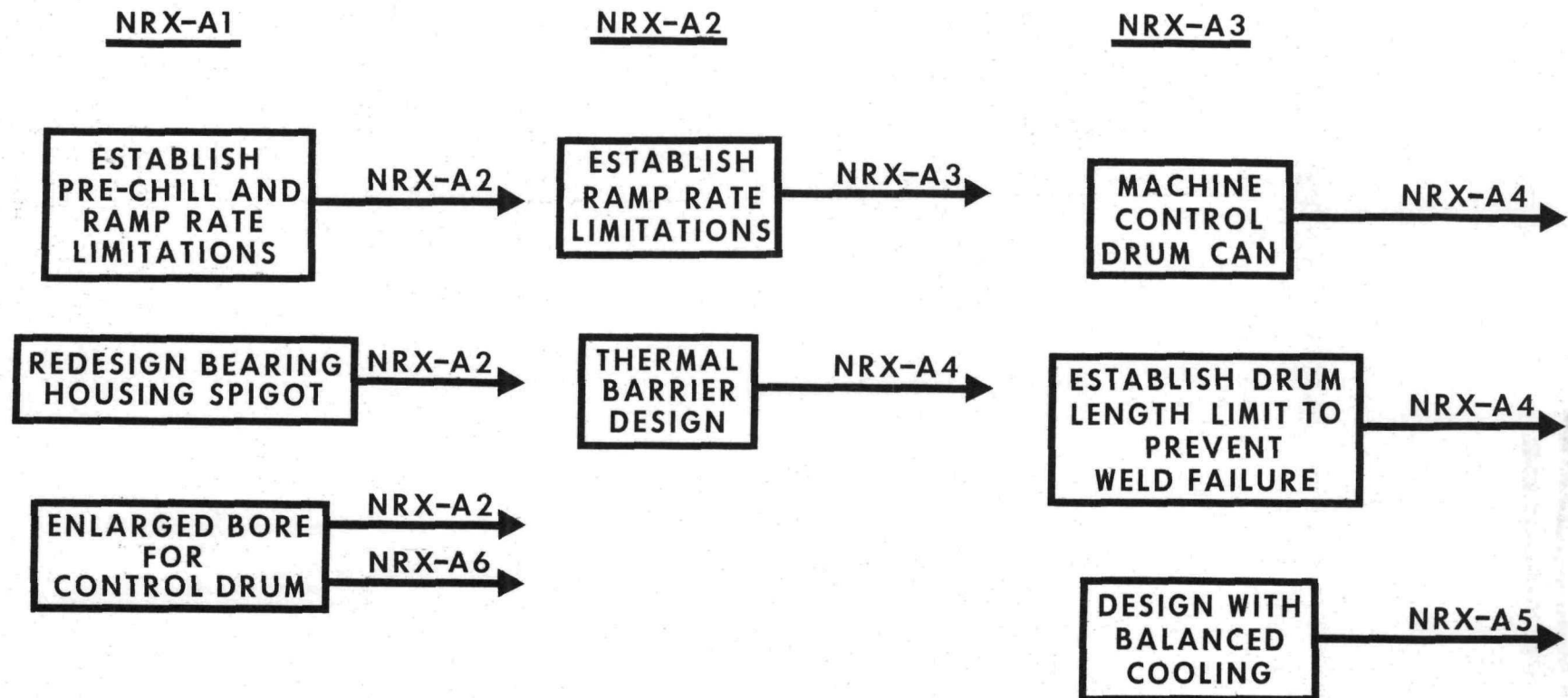
Astronuclear
Laboratory

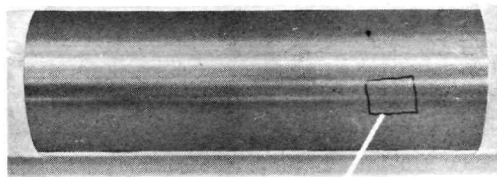


**CONFIDENTIAL
RESTRICTED DATA**

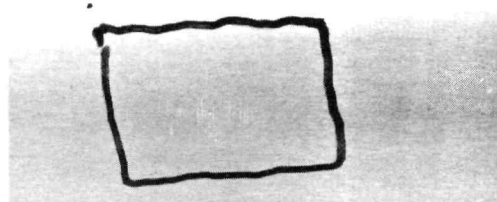
Atomic Energy Act - 1954

OUTER REFLECTOR TRANSIENT THERMAL GRADIENTS

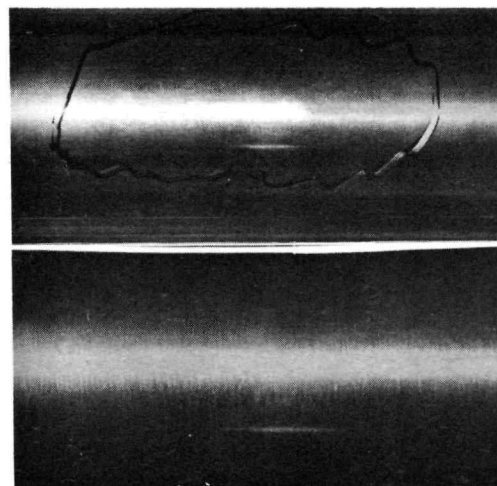




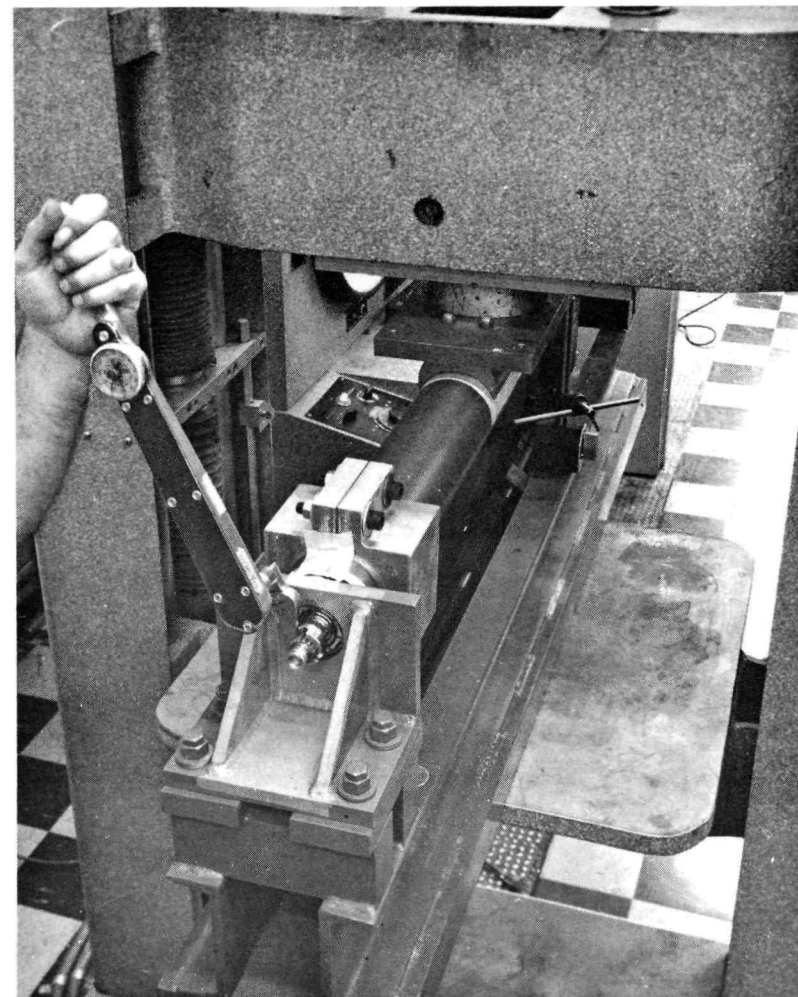
LABORATORY
TEST SETUP



DRUM SURFACE
AFTER
LABORATORY
TEST

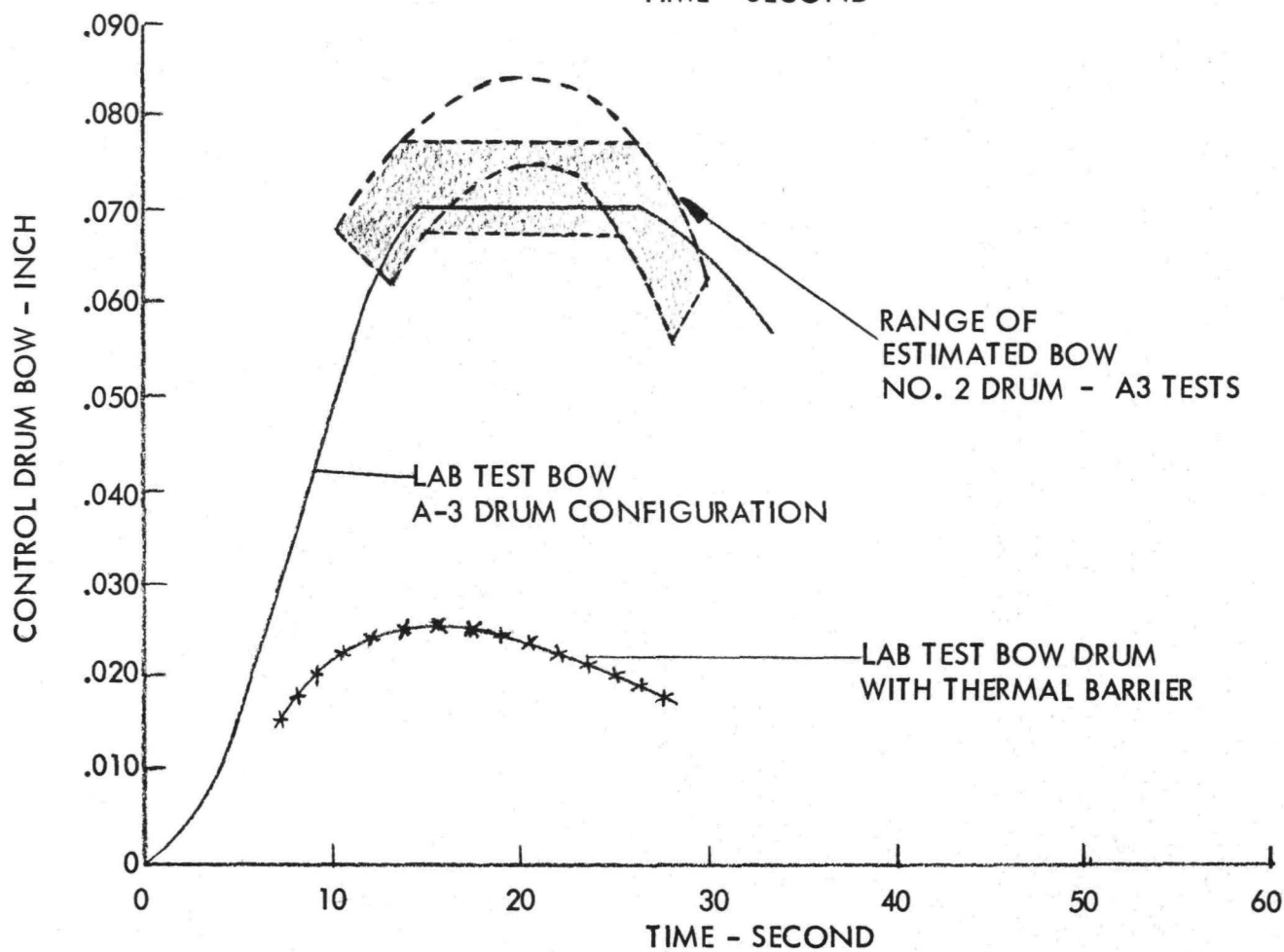
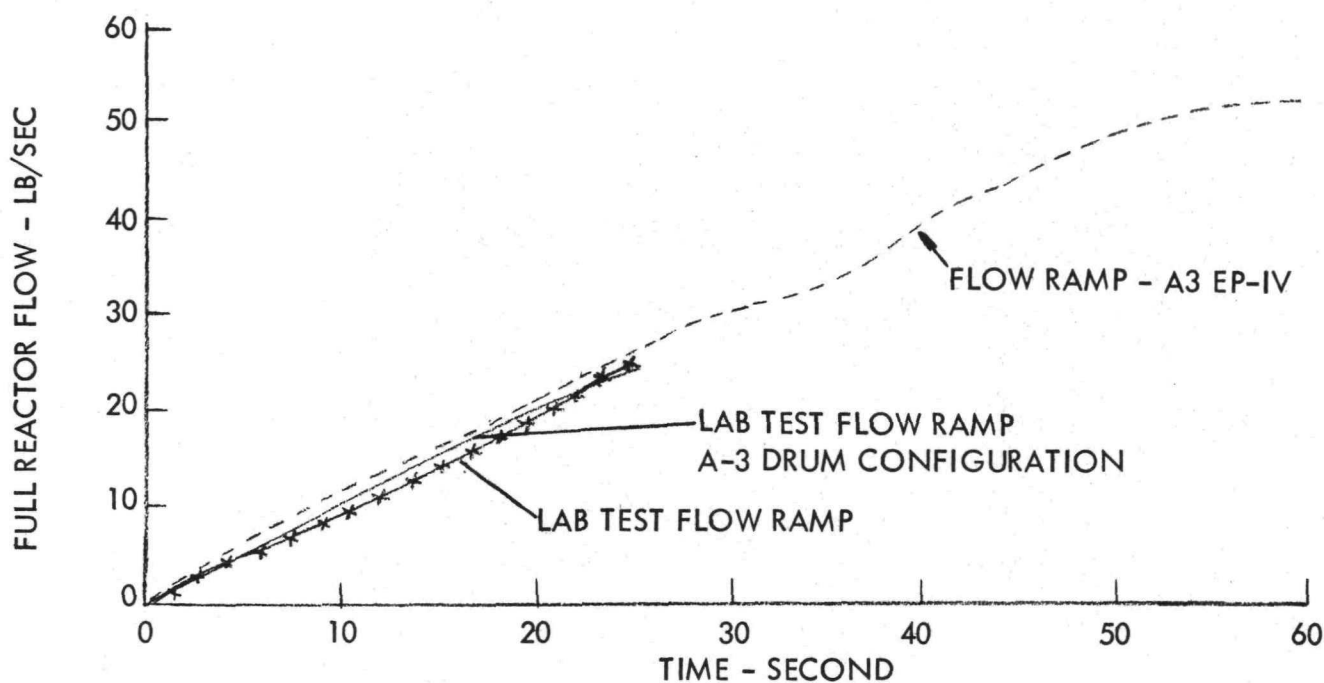


DRUM SURFACE
AFTER A-3 TEST

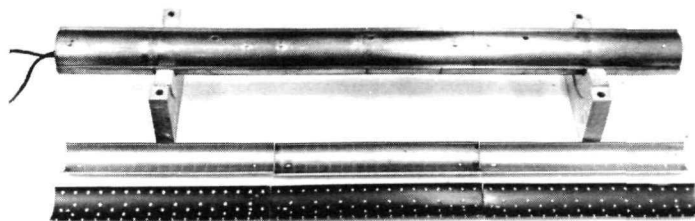


CONTROL DRUM REFLECTOR SECTOR RUB TEST

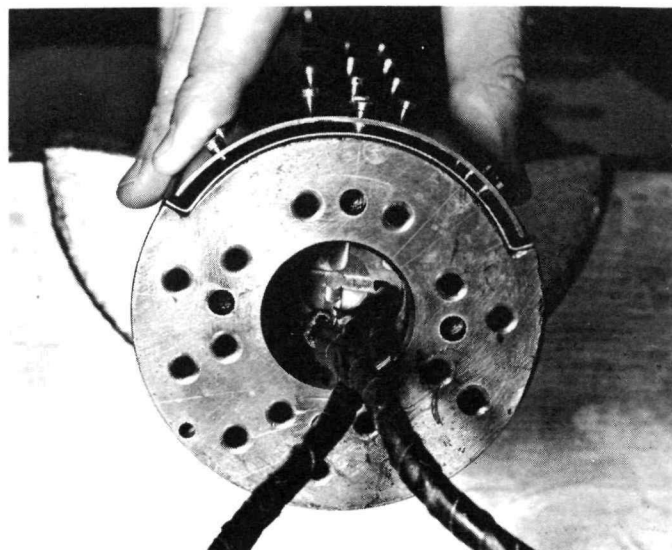
603483



NRX-A3 EP-IV TEST



THERMAL BARRIER ELEMENTS



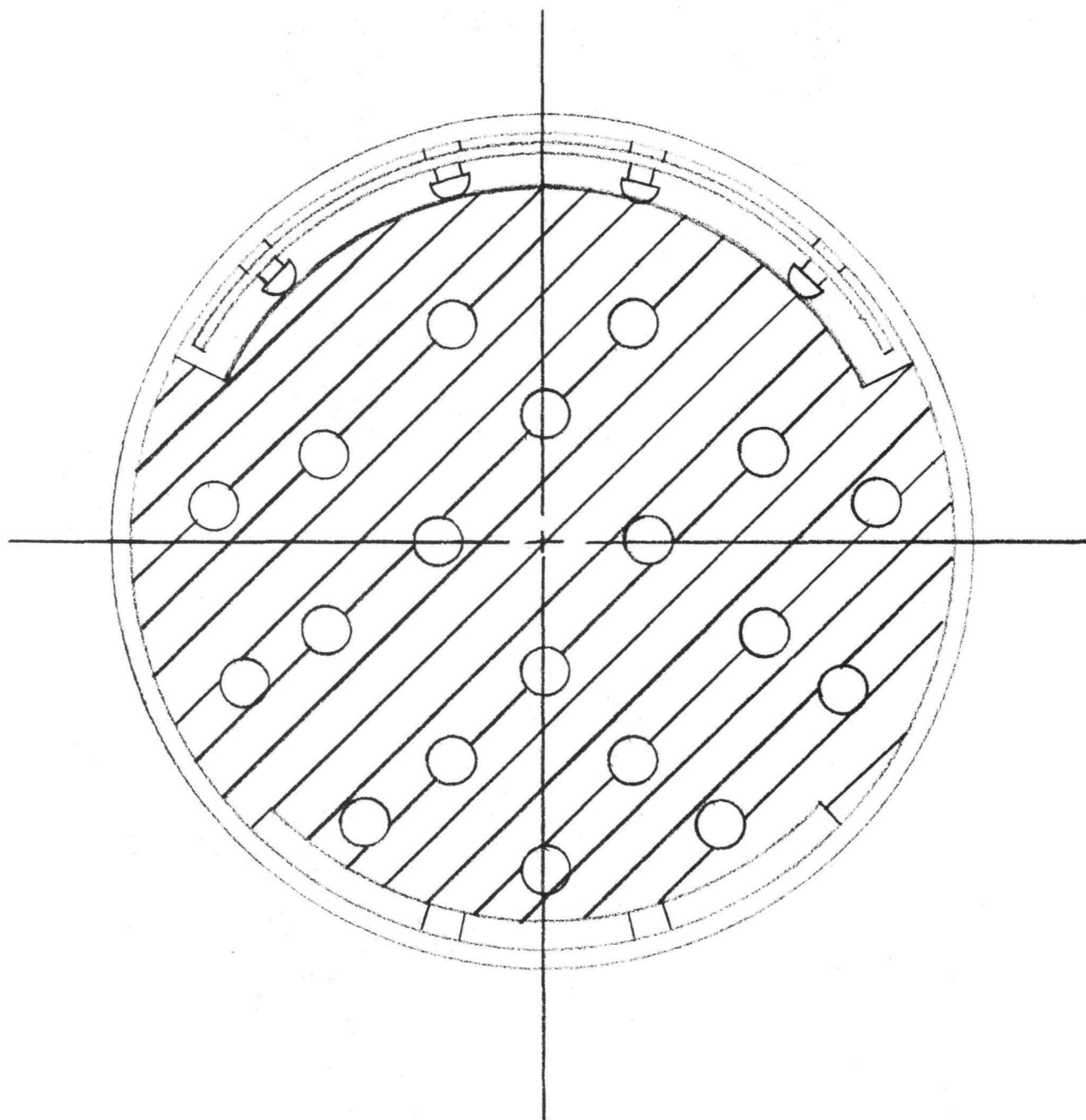
THERMAL BARRIER INSTALLED



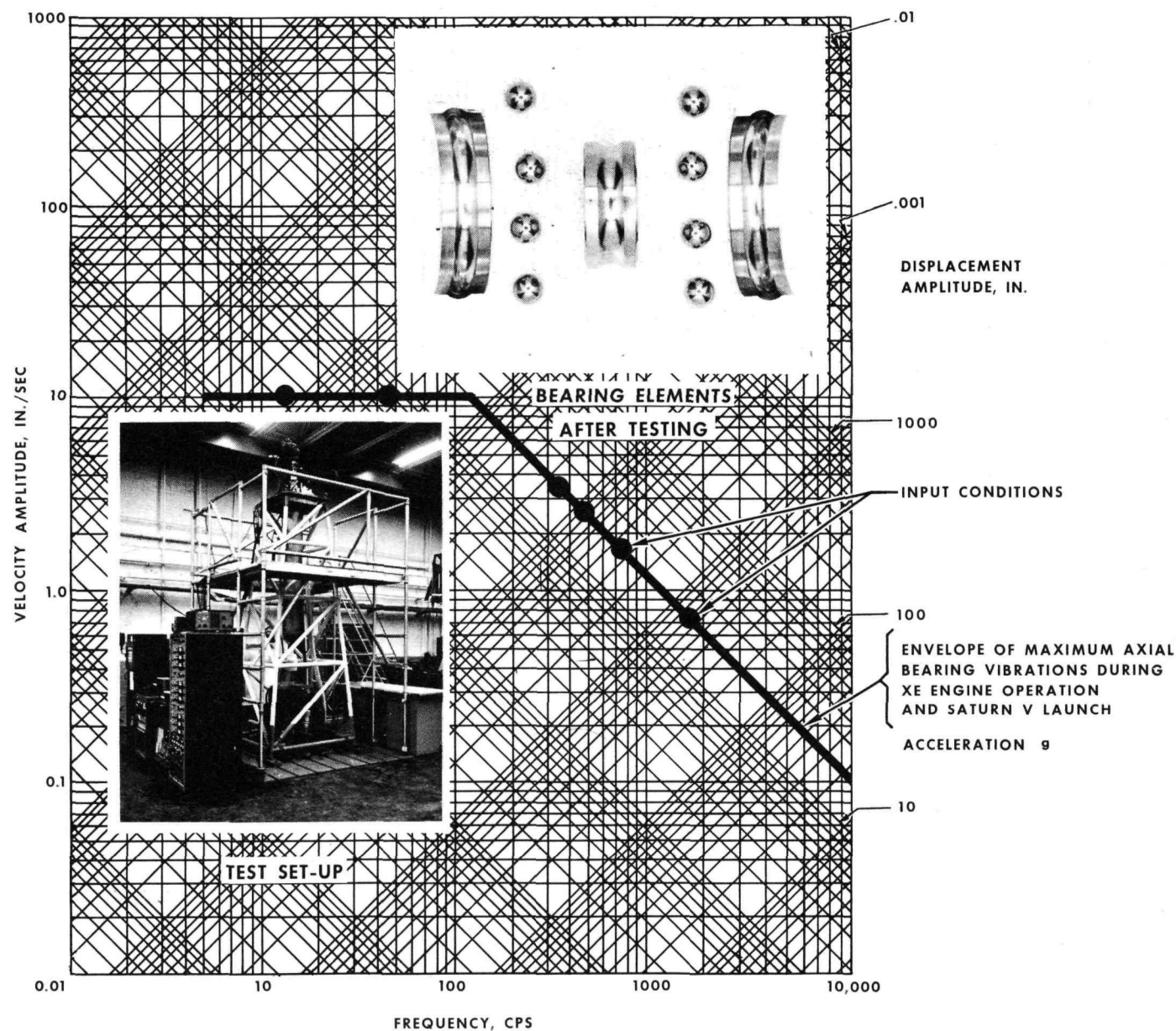
TEST RIG

CONTROL DRUM BOW TESTS
WITH THERMAL BARRIER

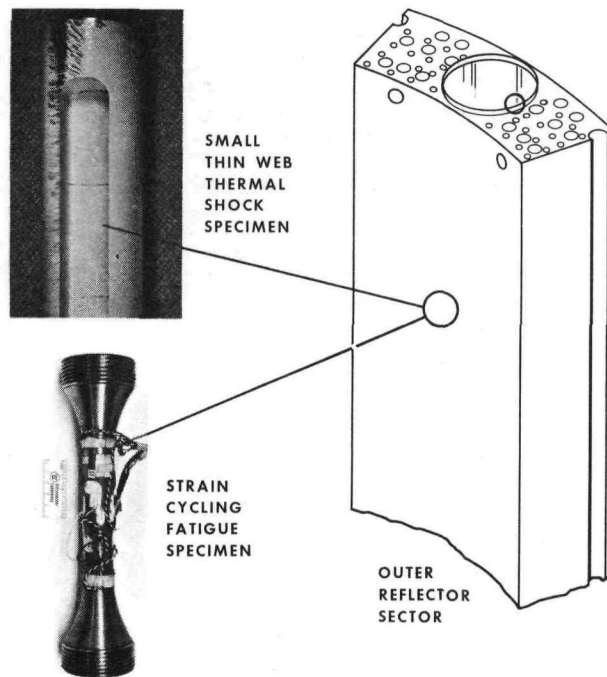
5643960



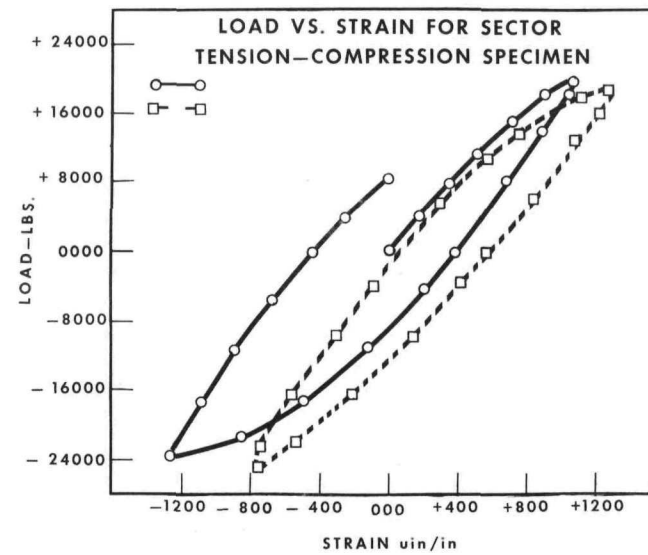
PROPOSED CONTROL DRUM MODIFICATION
TO REDUCE BOW



CONTROL DRUM BEARING AXIAL VIBRATION TESTS



SECTOR THERMAL STRESS TESTS



603482

DESIGN SUPPORT

FUEL ELEMENT
YIELD

MECHANICAL
DESIGN PROPERTIES

CONTROL DRUM
BEARING LIFE

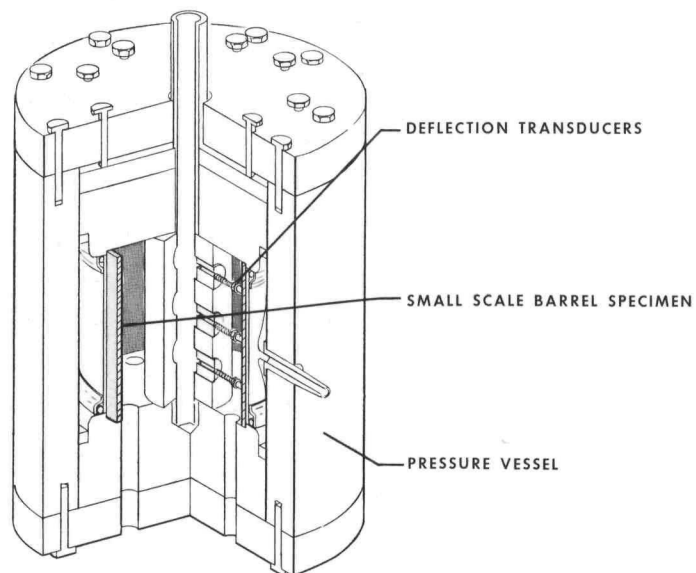
INSTALLED
INSTRUMENTATION
CALIBRATION

ORIFICE IMPEDENCE
CHARACTERISTICS

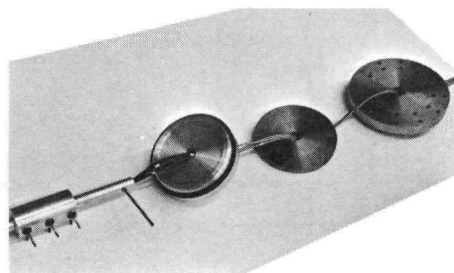
SMALL HARDWARE
CHARACTERISTICS

SEAL LEAKAGE
TESTING

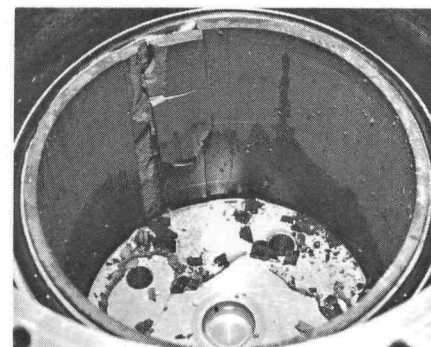
VIBRATION
TESTING



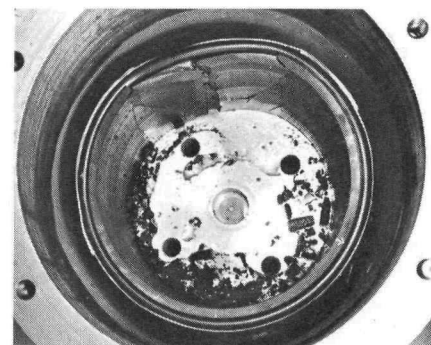
TEST FIXTURE



EXPLODED VIEW OF
DEFLECTION MEASURING SYSTEM



ONE PIECE BARREL
(.016 INCH EXTERNAL RESTRAINT)
FAILURE AT 575 PSI



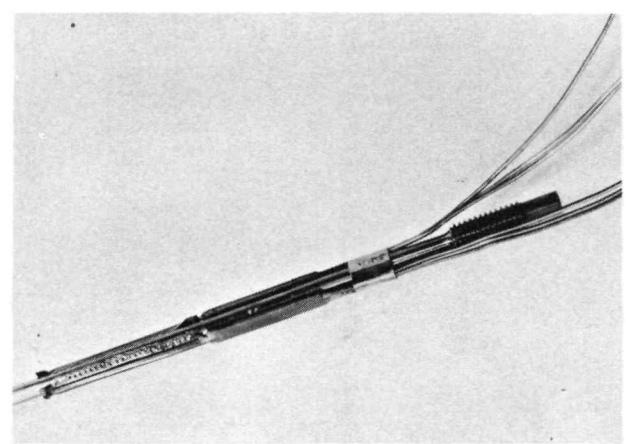
BONDED STAVE BARREL
(.016 INCH EXTERNAL RESTRAINT)
FAILURE AT 525 PSI

SMALL SCALE BARREL TESTS

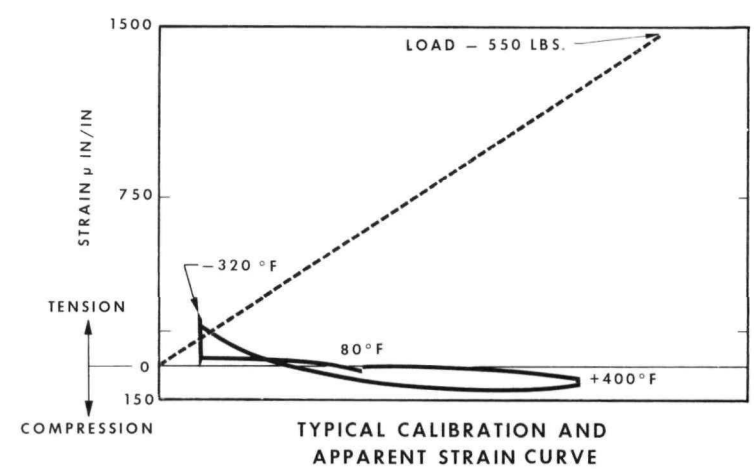
564394D

NRX-A3 AND NRX-A4 INSTRUMENTATION CALIBRATION AND PROOF TESTING

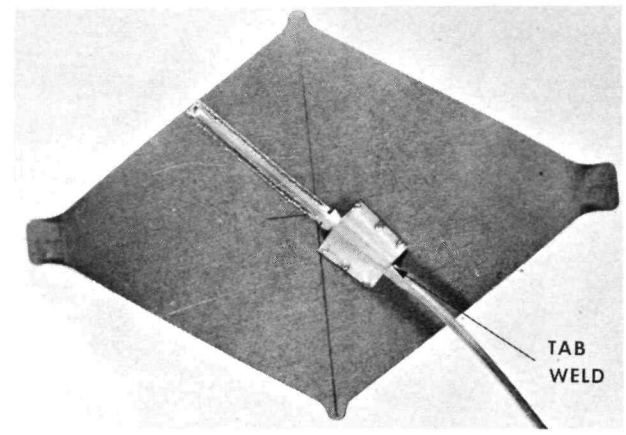
TIE RODS



INSTRUMENTED TIE ROD

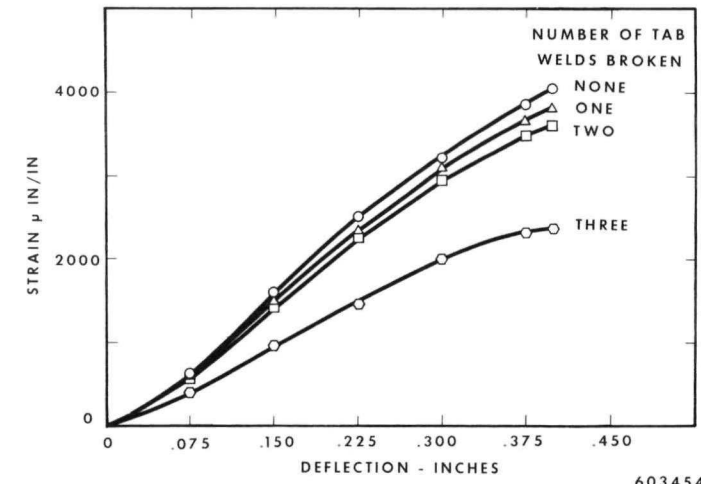


LATERAL SUPPORT SPRINGS



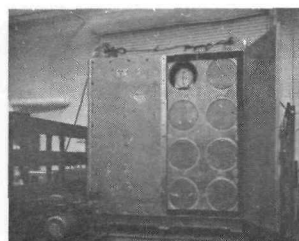
INSTRUMENTED LATERAL SUPPORT SPRING

EFFECT OF SUCCESSIVE REMOVAL OF TAB WELDS UPON CALIBRATION

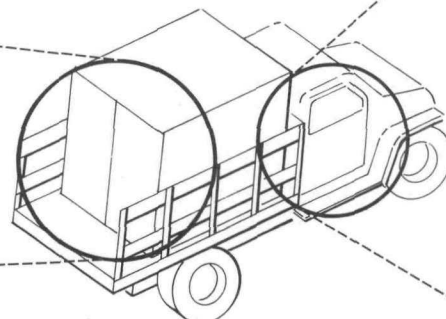


603454D

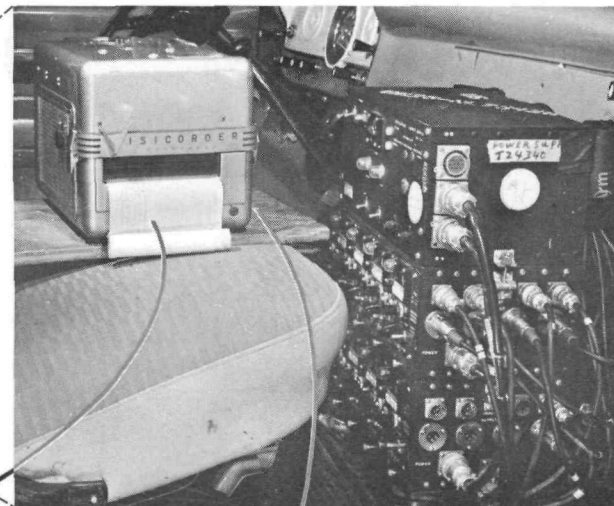
FUEL ELEMENT CONTAINER TRANSPORTATION TEST



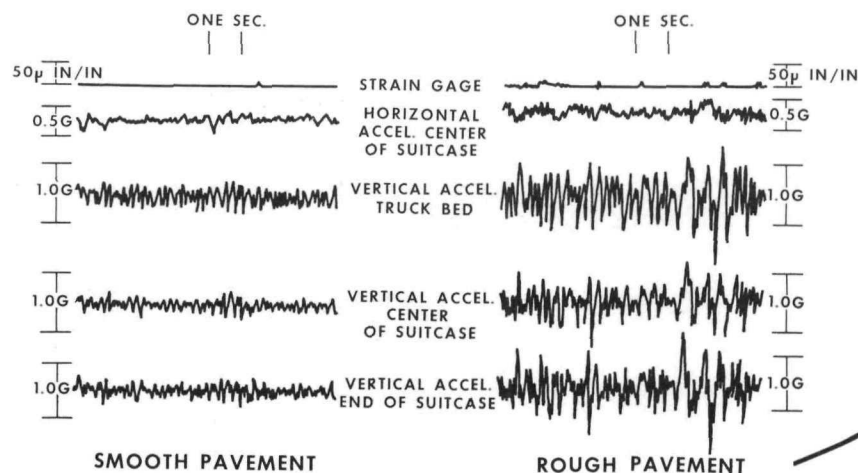
ELEMENT CONTAINER IN
 BIRDCAGE SHIPPING CONTAINER



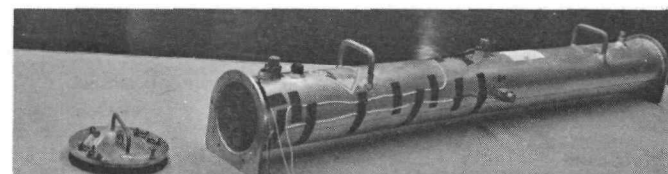
ROAD SHIPMENT FROM
 ASTRO FUEL LAB TO
 ASTRONUCLEAR LAB



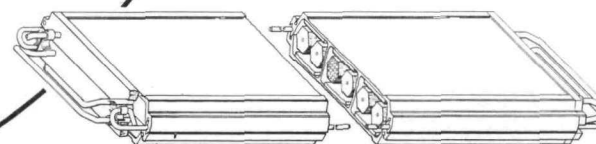
TRANSDUCER AMPLIFIER AND RECORDING SYSTEM



SAMPLE DATA RECORD



PRESENT ELEMENT CONTAINER
 ROAD TEST

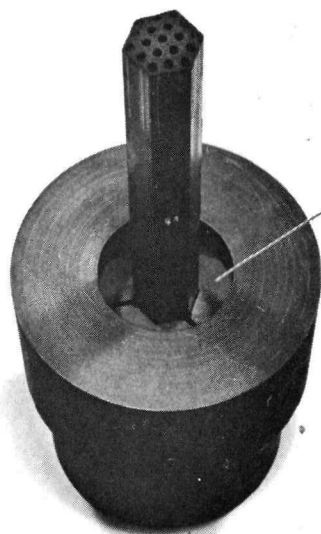


NEW DESIGN ELEMENT CONTAINER
 ROAD TEST AND VIBRATION TEST

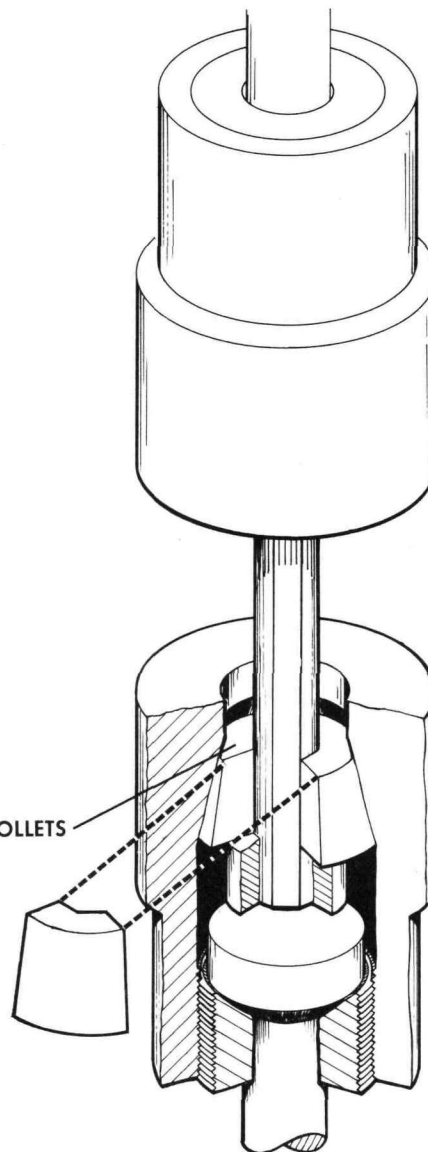
603456D



TENSION TEST FAILURE MODE

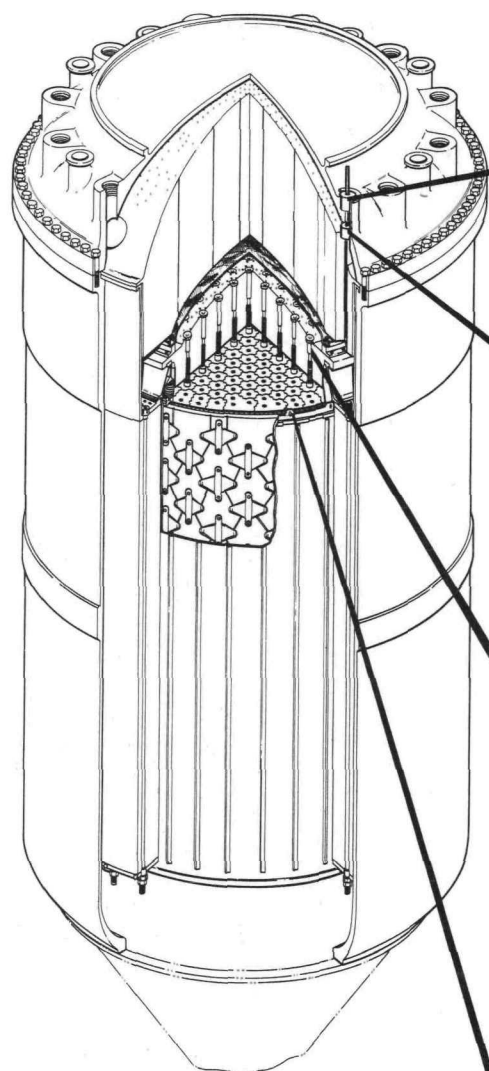


SPLIT COLLETS

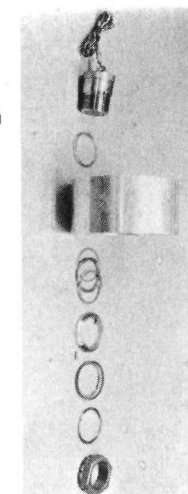


603443 D

GRIPPING ARRANGEMENT AND FAILURE MODE
FOR FUEL ELEMENT TENSION TESTS



INSTRUMENTATION
SEAL BLOCK ASSEMBLY

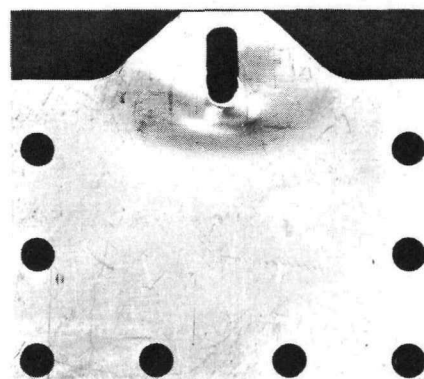


INSTRUMENTATION
SLEEVE ASSEMBLY

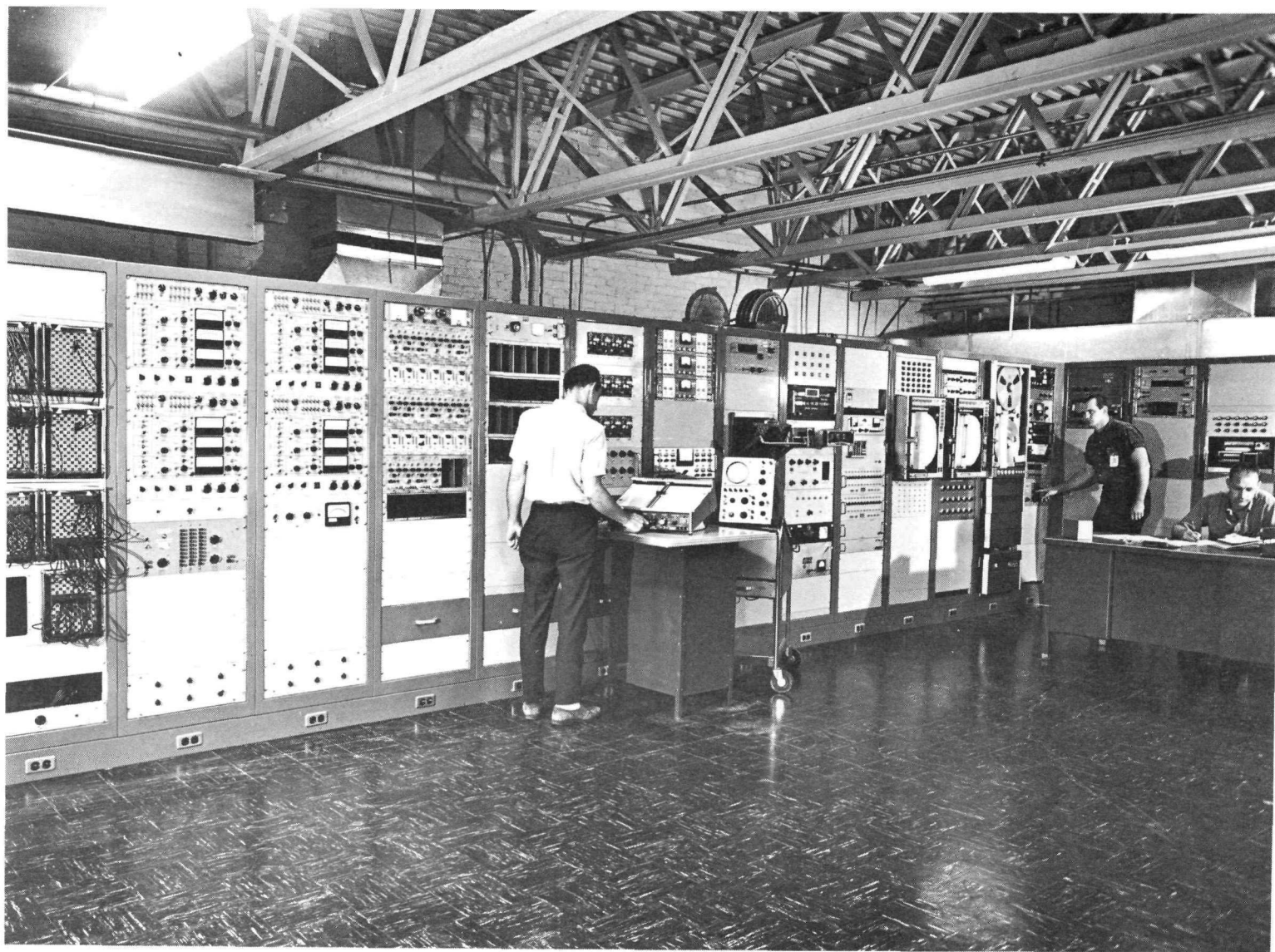


TIE ROD HOLDER-CLUSTER NUT
THREADED ASSEMBLY

REACTOR HARDWARE TESTS



CORE ASSEMBLY BARREL ATTACHMENT
TAB AFTER TESTING



FACILITIES IMPROVEMENT

FACILITY

HIGH TEMPERATURE
MECHANICAL LOAD

PROGRAM PLANS

LOADING FURNACE WITH
MORE VIEWPORTS AND
IMPROVED LOADING
CAPABILITY

ADDITIONAL NEEDS

LOADING FURNACE WITH
HIGHER HYDROGEN
CAPABILITY

FRICTION FURNACE

ADDITIONAL FACILITIES TO
ACCOMMODATE LARGER
SPECIMENS, PROVIDE BETTER
MEASUREMENTS OF
TEMPERATURE, NORMAL LOAD,
FRICTION FORCE AND
SPECIMEN MOTION

FACILITIES IMPROVEMENT

FACILITY

SINGLE ELEMENT
CORROSION FURNACES

PROGRAM PLANS

1. PARALLEL LOOPS FOR COOLANT AND ATMOSPHERE TO GIVE WIDE VARIATION IN FLOW AND GAS COMPOSITION. PARTS AVAILABLE — BUILD AND INSTALL WHEN MANPOWER IS AVAILABLE.
2. DEVELOPMENTAL FURNACE WITH MORE VIEWPORTS, SIMPLIFIED END CLOSURES AND TRANSITION AND BETTER ACCESSIBILITY. MEASURE TEMPERATURE AND MAKE VOLTAGE PROBE MEASUREMENTS AT 5 INCH INTERVALS.
3. WALTZ MILL VAPORIZER INSTALLED. OPERATIONAL IN OCTOBER.
4. SECOND FURNACE AND HEAT EXCHANGER IN PARALLEL.

ADDITIONAL NEEDS

CORROSION FACILITY
WITH PREHEATER SECTION
TO ELIMINATE ELECTRICAL
INTERACTIONS

CRYOGENIC HYDROGEN
SUPPLY TO PROVIDE
CORRECT TEMPERATURE
DISTRIBUTION FOR CENTER
HEX TESTS AND TIE TUBE
TESTS

FACILITIES IMPROVEMENT

FACILITY

PROGRAM PLANS

ADDITIONAL NEEDS

CLUSTER CORROSION
FURNACE

CLUSTER TESTING BUT WITH
LIMITED LOADING CAPABILITY.

REDESIGN OF ENDS TO
PROVIDE MORE LOADING
FLEXIBILITY.

NEED ADDITIONAL POWER
TO OPERATE AT FULL POWER
AND FULL FLOW. CAN RUN
ONLY 80% FLOW AT FULL
TEMPERATURE.

MODIFY END CLOSURES TO
PERMIT ADDITIONAL ROW
OF ELEMENTS IN PERIPHERAL
CORROSION RIG.

MULTIPLE CLUSTER
FURNACE

NEED FACILITY TO EVALUATE
PERIPHERAL DESIGN CANDIDATE
AND MODIFICATIONS.

FACILITIES IMPROVEMENT

FACILITY

HIGH TEMPERATURE
MECHANICAL LOAD

PROGRAM PLANS

ADDITIONAL LOADING
FURNACE WITH MORE
VIEWPORTS AND IMPROVED
LOADING CAPABILITY

ADDITIONAL NEEDS

LOADING FURNACE WITH
HIGHER HYDROGEN
CAPABILITY

FRICTION FURNACE

ADDITIONAL FACILITIES TO
ACCOMMODATE LARGER
SPECIMENS, PROVIDE BETTER
MEASUREMENTS OF
TEMPERATURE, NORMAL LOAD,
FRICTION FORCE AND
SPECIMEN MOTION

PERIPHERAL CORROSION

- A. SINGLE CLUSTER PERIPHERAL CORROSION SIMULATION
 - i. PYROSTRIP
 - ii. HOT BUFFER
 - iii. METHANE GENERATING PERIPHERY
- B. ENCAPSULATED SPRING COOLING
- C. HOT CAVITY MASS TRANSFER
- D. HOT BUFFER LOCKUP EVALUATION
 - i. THERMAL EXPANSION OF FILLER STRIPS
 - ii. FRICTION BETWEEN PYROTILE AND COATED ELEMENTS
- E. NRX-A6 SEAL SEGMENT FRICTION TESTS
- F. NRX-A6 LATERAL SUPPORT SYSTEM STRUCTURE INTEGRITY
- G. NRX-A6 LATERAL SUPPORT AND SEAL MODEL LEAKAGE TESTS
- H. NRX-A6 REFLECTOR SEAL LEAKAGE



FUEL ELEMENT AND SUPPORT BLOCK CORROSION

- A. SINGLE ELEMENT DEVELOPMENT CORROSION TESTS
 - i. BETTER REACTOR SIMULATION
 - ii. IMPROVED CHUCKS
 - iii. PINHOLE SIMULATION
 - iv. END AND UNDERCUT CORROSION EVALUATION
- B. CLUSTER CORROSION TESTING
- C. NON-SUPPORT BLOCK CLUSTER CORROSION TESTING AND MECHANICAL LOAD TESTING
 - i. EVALUATE CONCEPTS
 - ii. QUALIFICATION TESTING
- D. SUPPORT BLOCK CORROSION TESTING
- E. SUPPORT BLOCK MECHANICAL TESTING
 - i. TEMPERATURE AND LOAD
 - ii. BONDING AND FLAKING OF NbC INTERFACES
- F. MECHANICAL PROPERTIES OF CORRODED ELEMENTS AND SUPPORT BLOCKS
- G. ACOUSTICAL TESTS OF COATING INTEGRITY ON CORRODED AND UNCORRODED COMPONENTS
- H. ORIFICE AND ORIFICE SEALANT TESTS



INTERSTITIAL CORROSION

- A. CORROSION IN LONG NARROW GAPS WITH REACTOR TEMPERATURE DISTRIBUTION
- B. CORROSION IN SHORT GAPS - ISOTHERMAL
- C. SIMULATE WELDING OF ELEMENTS OBSERVED IN NRX-A3
- D. CORE GAP AND PRESSURE DISTRIBUTION VS. BUNDLING. LOOK AT EFFECTS OF PINHOLING, LARGE GAPS, OVER-SIZE ELEMENTS AND RADIAL INFLOW

CENTRAL ELEMENT AND TIE ROD INTEGRITY

- A. CORROSION TEST OF NRX-A4 CONFIGURATION
- B. CORROSION TEST WITH COUNTERFLOW TIE TUBE

OUTER REFLECTOR TRANSIENT THERMAL GRADIENTS

- A. CONTROL DRUM BOW TESTING - NRX-A5 AND NRX-A6
- B. NRX-A6 OUTER REFLECTOR MODEL TESTS
THERMAL SHOCK AND THERMAL GRADIENT

DESIGN SUPPORT

- A. MECHANICAL TESTS TO EVALUATE COMPONENTS SUCH AS CLUSTER NUTS, TIE RODS, INSTRUMENTATION SEAL BLOCKS, ETC.
- B. RESONANCE OF REACTOR ON ETS ASSEMBLY
- C. MINOR SEAL TESTING

- - - - -

PROBLEMS ARISING FROM REACTOR TEST RESULTS

- A. MUST HAVE FACILITIES AND CAPABILITY TO RESPOND RAPIDLY TO MAJOR PROBLEMS
 - i. NRX-A3 FLOW SCRAM
 - ii. SUPPORT BLOCKS
- B. NEED MARGIN TO EVALUATE DESIGN CHANGES
- C. DAY-TO-DAY PROBLEMS NEEDING TEST RESULTS